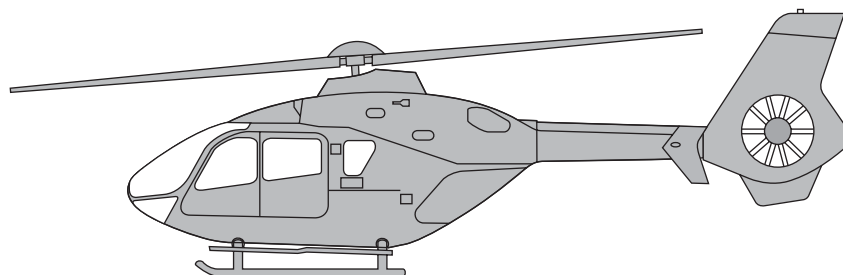




EUROCOPTER
EC135

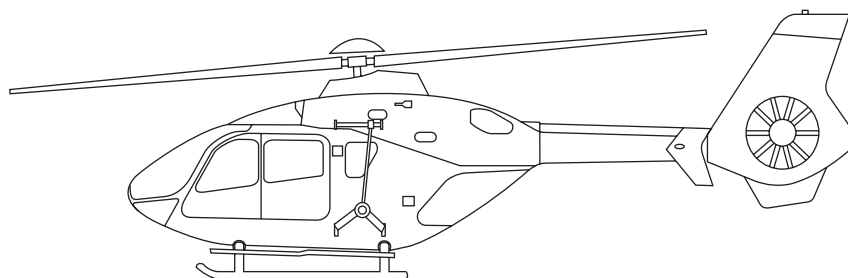
Technical
Data

EC135
(Civil Version)



EC135

EC635
(Military version)



Utility/armed version
EC635

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Eurocopter's policy is one of on-going product enhancement which means that alterations in definition, weights, dimensions or performance may be made at any time without notice being included in those documents that have already been issued.

This document cannot thus be taken as an offer or serve as an appendix to a contract without a prior check as to its validity and prior written agreement of Eurocopter.

The operational or certification regulations, as defined by the local authorities, can make compulsory the installation of some of the equipment or recommended solutions, listed in this document. This list does not claim to cover the whole of the worldwide operational requirements nor the equipment not specifically related to the helicopter (for example: life jacket) or necessary for particular missions (for example: supplemental oxygen). The operator is responsible for ascertaining with his local authorities that the planned configuration of the helicopter complies with regulatory requirements for the area(s) of operations and the type(s) of mission(s) considered.

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Some items are controlled under the provisions of the US International Traffic in Arms Regulations and therefore require a US re-export licence. EUROCOPTER is committed to strict compliance with all applicable US Regulations and requires a corresponding commitment from the customer, too. EUROCOPTER cannot guarantee that a US ITAR licence will be granted. Nevertheless EUROCOPTER will perform all necessary procedures for obtaining such licence. Customer shall support the licence application procedure by providing the necessary documentation and information, particularly end-user certificates, undertakings, documents on the utilization of the commodities, etc. Customer shall adhere to the parameters and requirements of the re-export licence and US regulations. EUROCOPTER reserves the right to deliver the helicopter without the ITAR-controlled items should the US authorities deny or delay the (re-)export licence. The same is valid if the US authorities do not react at all or in cases of a return without action or if otherwise the absence of a required licence or provisions of US law inhibit a delivery in time.

1 Foreword



The EC135 is a light twin-engine, multi-purpose helicopter of the 2-3 ton class with up to 8 seats for pilot/s and passengers. Underlining its multi-role capabilities, it can even be operated single pilot IFR as an option. The helicopter combines Eurocopter's latest Technologies, like advanced cockpit design, modern avionics, Fenestron® anti-torque-device and all-composite bearingless main rotor system, giving the helicopter an outstanding maneuverability. Optimized main rotor blades with advanced tip geometry in combination with a Fenestron® with unequal blade spacing make the EC135 the quietest helicopter in its class, bringing it 6.5 dBA below the ultra-stringent ICAO limit. The built-in anti resonance isolation system (ARIS) filters rotor-induced vibrations and thus enhances flying comfort to a maximum. As a result, the vertical vibration level is far below 0.1g at hover with no increase with speed.

Due to its extreme simplicity, the rotor system contributes to highest safety standards and, at the same time, reduces maintenance to a minimum. The first scheduled maintenance is the intermediate inspection after 400 Flh. In addition, the rotor system together with high TBO gearbox and airframe components grant for high in-service-time of the helicopter.

Depending on the operator's preferences, the EC135 can be equipped with either Arrius 2B2 or Pratt & Whitney PW206B2 power plants - both are FADEC controlled. These powerful and reliable engines in combination with the lifting system provide outstanding performance and vital power reserves even in OEI scenarios.

For training purpose an OEI training mode is implemented to perform a realistic OEI training. This training mode is based on a twin engine training concept featuring a so called TRAINING and a TRAINING IDLE engine.

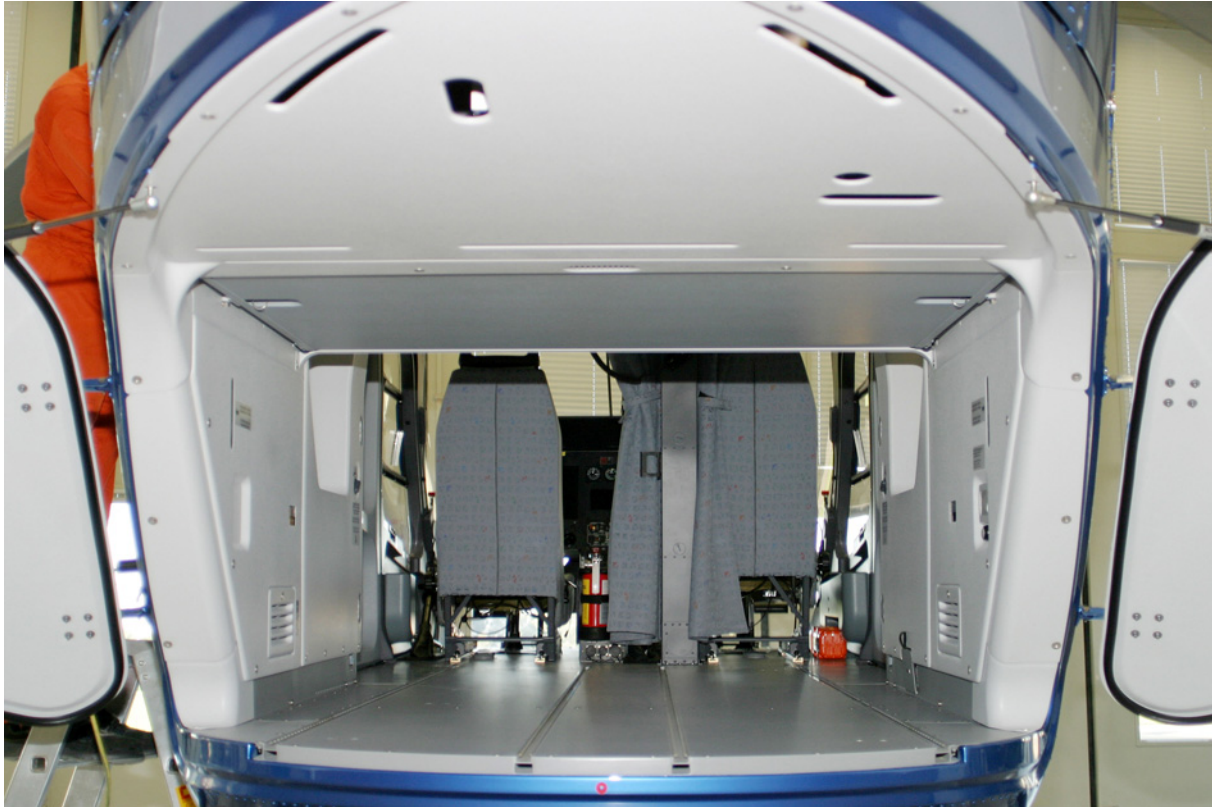
Twin-engine reliability is complemented by a tandem hydraulic and dual electrical system as well as a redundant lubrication and cooling system for the main transmission.

Further safety aspects of the EC135 are design elements like energy absorbing fuselage and seats, as well as the crash resistant fuel cells.

A wide range of quick interchangeable optional equipment is available for the EC135, e.g. emergency floats, hoist, SX16 search light, single or dual cargo hook and many more. Together with its most versatile cabin layout the EC135 is ready to operate in different missions, like police / surveillance, passenger / VIP transport, EMS, public service, to highlight on a few.

Compared to other helicopters in its class, the EC135 offers a large cabin, featuring:

- Excellent outside visibility for pilots and passengers
- Roomy cabin which accommodates long or bulky freight
- Unrivalled side loading (no door posts) and rear loading capability
- Unobstructed and flat floor all over the cabin area with integrated airline style rails



Alternatively to a conventional cockpit, the EC135 is available with "glass cockpit", which comprises primary flight displays (PFD) and NAV displays (ND). All LCD screens are well arranged on the instrument panel, easy to read even if viewed from an angle and feature perfect readability in any light conditions. The unique color coding, warning and information concept helps the pilot/s to collect all relevant parameters while suppressing presentation of non-relevant information.

Common to the conventional and the glass cockpit is the Central Panel Display System (CPDS). Included in this CPDS there is Eurocopter's unique first limit indicator (FLI) which dramatically simplifies engine and torque monitoring. Being relieved from the instrument scan without missing vital information, the pilot/s can dedicate more of his/their attention to the mission.

Conventional Instrumentation (analog instruments)



| | |
|--|--|
| VFR- Single Pilot or Dual Pilot | Packages based on Avionics Solution 1 |
| IFR- Dual Pilot | Packages based on Avionics Solution 3 or 4 |
| IFR- Single Pilot | not available |

Glass Cockpit Instrumentation (MEGHAS / FCDS)

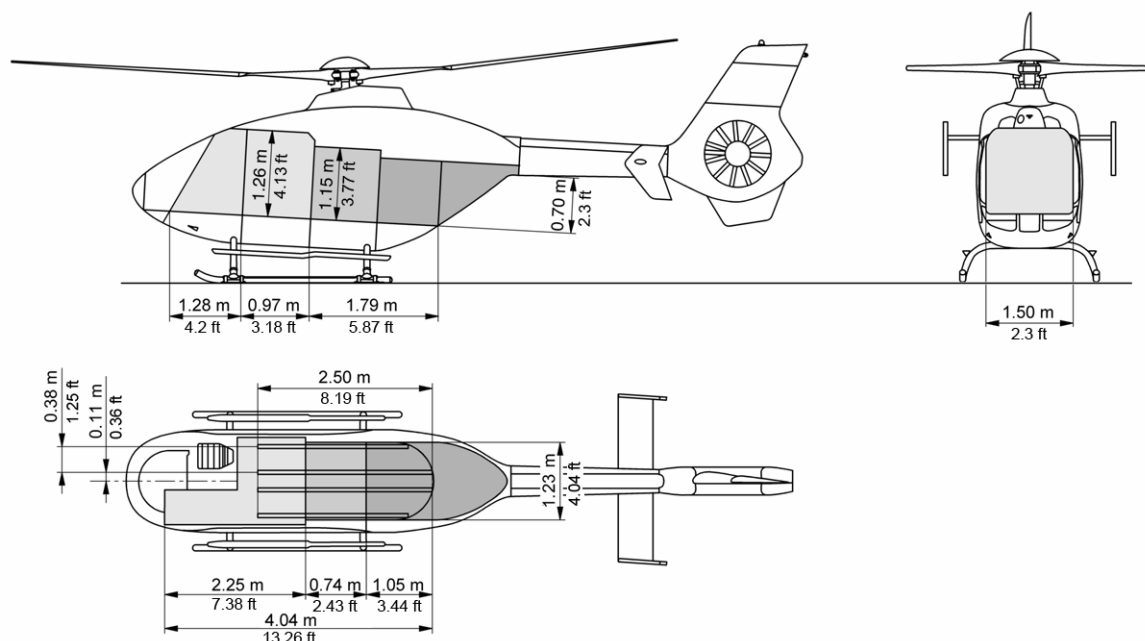


| | |
|---|--|
| VFR- Single Pilot or Dual Pilot | covered by SP or DP-IFR solutions |
| IFR- Dual Pilot or Single/Dual Pilot | Packages based on Avionics Solution, 8 or 11 |
| IFR- Single Pilot | Packages based on Avionics Solution 10 or 12 |

Latest news / highlights:

- Light twin HUMS
- External hoist capability LH and RH
- EUROCOPTER owned simulator for training

2.2 Interior dimensions



| | Floor area | | Volume | |
|-----------------------------|---------------------|-----------------------|---------------------|------------------------|
| Cabin & baggage compartment | 4.35 m ² | 46.83 ft ² | 4.90 m ³ | 173.04 ft ³ |
| Cockpit (pilot side) | 1.15 m ² | 12.38 ft ² | 1.00 m ³ | 35.31 ft ³ |
| Total (undivided) | 5.50 m ² | 59.21 ft ² | 5.90 m ³ | 208.35 ft ³ |

2.3 Possible cabin arrangement (seats & equipment as option)

| | |
|----------------------------|--|
| Passenger transport | <ul style="list-style-type: none"> ■ 1 or 2 pilots + 7 or 6 passengers ("6 Passenger Transport" version) ■ 1 or 2 pilots + 6 or 5 passengers ("5 Passenger Transport" version) ■ 1 or 2 pilots + 6 or 5 passengers ("5 Corporate Passenger Transport" version) ■ 1 or 2 pilots + 5 or 4 passengers ("4 VIP Passenger Transport" version) |
| Casualty evacuation | <ul style="list-style-type: none"> ■ 1 pilot + 1 litter + up to 4 seats for doctor and attendants ■ 1 pilot + 2 litters + up to 3 seats for doctor and attendant ■ 2 pilots + 1 litter + up to 3 seats for doctor and attendants ■ 2 pilots + 2 litters + up to 2 seats for doctor and attendant |
| Freight transport | <ul style="list-style-type: none"> ■ 1 pilot + 4.9 m³ (173.04 ft³) in cabin and cargo compartment |

2.4 Weight

Note : margin $\pm 1.5\%$

| | kg | lb |
|--|-------|-------|
| ■ Empty weight, wet (in standard aircraft configuration) | 1,455 | 3,208 |
| ■ Useful load (for standard aircraft configuration) | 1,455 | 3,208 |
| ■ Pilot | 80 | 176 |
| ■ Payload and / or fuel | 1,375 | 3,031 |
| ■ Maximum take-off weight | 2,910 | 6,415 |

2.5 Fuel Capacities

Note: Tolerance of fuel figures: $\pm 2\%$
Fuel density used is 0.8 kg/liter.

| | Usable Fuel | | | Unusable Fuel | |
|---------------|-------------|-------|-------|---------------|-----|
| | lb | kg | l | lb | kg |
| ■ Main Tank | 1038.6 | 471.1 | 588.9 | 7.5 | 3.4 |
| ■ Supply Tank | 196.8 | 89.3 | 111.6 | 9.3 | 4.2 |
| ■ Total | 1235.4 | 560.4 | 700.5 | 16.8 | 7.6 |

2.6 Engines

2 Pratt & Whitney turbine engines – PW206B2

or

2 Turbomeca turbine engines - ARRIUS 2B2

Engine ratings

Thermodynamic limits per engine at SL, ISA

| | kW | ch | shp |
|---|-----|-----|-----|
| PW206B2 | | | |
| ■ One Engine Inoperative (OEI), 30 sec power | 609 | 828 | 816 |
| ■ One Engine Inoperative (OEI), 2.0 min power | 580 | 789 | 777 |
| ■ One Engine Inoperative (OEI), MCP | 528 | 718 | 708 |
| ■ Take-Off Power (TOP) | 498 | 677 | 667 |
| ■ Maximum Continuous Power (MCP) | 457 | 621 | 612 |
| ARRIUS 2B2 | | | |
| ■ One Engine Inoperative (OEI), 30 sec power | 609 | 828 | 816 |
| ■ One Engine Inoperative (OEI), 2.0 min power | 580 | 789 | 777 |
| ■ One Engine Inoperative (OEI), MCP | 528 | 718 | 708 |
| ■ Take-Off Power (TOP) | 473 | 643 | 634 |
| ■ Maximum Continuous Power (MCP) | 442 | 601 | 592 |

2.7 Main transmission ratings

| Single engine operation | | kW | ch | shp |
|-------------------------|--------------------------------|---------|---------|---------|
| ■ | 30 sec OEI-power | 1 x 526 | 1 x 715 | 1 x 705 |
| ■ | 2.0 min OEI-power | 1 x 513 | 1 x 698 | 1 x 687 |
| ■ | Maximum continuous OEI-power | 1 x 368 | 1 x 501 | 1 x 493 |
| Twin engine operation | | | | |
| ■ | Take-Off Power (TOP) | 2 x 320 | 2 x 435 | 2 x 429 |
| ■ | Maximum Continuous Power (MCP) | 2 x 283 | 2 x 385 | 2 x 380 |

3 Baseline Aircraft Definition

GENERAL

- Energy absorbing fuselage
- Tail boom with fixed horizontal stabilizer and two end-plates
- Vertical fin with Fenestron®
- Upper deck with fittings for main gearbox, engines, hydraulic and cooling system
- Cowlings for main transmission and engines
- Skid-type landing gear with skid protectors, capable of taking ground-handling wheels
- Long boarding steps, LH and RH
- Maintenance built-in steps and grips
- Exterior painting (single color)

COCKPIT, CABIN AND CARGO COMPARTMENT

- One-level cockpit, cabin and cargo compartment floor with integrated rails
- Glazed canopy
- Two hinged cockpit doors with sliding window
- Map case in pilot's door
- Two wide passenger sliding doors
- Two rear hinged clam-shell doors
- Longitudinally adjustable energy absorbing pilot and copilot seats with head rest and 4-point safety belts with an inertia-reel locking system
- Cabin boarding grips (LH and RH)
- Interior paneling with integrated basic sound insulation
- Flight controls (pilot side)
- Instrument panel with extension on pilot's side and glare shield
- Ram-air and electrical ventilating system for cockpit and cabin
- Headset holder in the cockpit
- Headset holder in the cabin
- Portable fire extinguisher
- Stowage net for first aid kit at the LH rear clam-shell door
- Flash light (torch)
- 4 Mobile tie-down rings
- 5 headset plugs in the cabin

BASIC INSTRUMENTATION

- Central Panel Display System (CPDS), consisting of:
 - Caution Advisory Display (CAD) with indication of:
 - Caution and advisory information
 - Fuel quantity indication
 - Vehicle and Engine Management Display (VEMD) with indication of:
 - Torque
 - Engine parameters (N1-RPM (for P&W) or Δ N1-RPM (for TM), oil pressure, oil temperature, Turbine Outlet Temperature (TOT), engine/FADEC rep EEC failure and parameter code messages, self diagnoses)
 - FLI (First Limit Indicator) for TQ, TOT, N1 (for P&W) or Δ N1 (for TM) as analogue display
 - Main transmission parameters (oil pressure, oil temp.)
 - Dual ammeter (generator)
 - Ammeter (battery)
 - Dual voltmeter
 - Outside Air Temperature (OAT)
 - Automatic in flight power check
 - Parameters of optional equipment (e.g. internal long range fuel tank)
- Engine cycle counter (on flight report page)
- Clock (2")
- Magnetic compass
- Triple (rotor and engines) RPM-indicator (2")
- Standard instruments: (single pilot) ¹⁾
 - Encoding altimeter (3")
 - Airspeed indicator (3")
 - Vertical speed indicator (3")
- Warning unit:
 - Engine fire warning with fuel emergency shut-off
 - Warning lights
 - Aural warning
- Main switch panel:
 - DC power control
 - Digital engine control (FADEC)
- Pitot / static system with electrical heated pitot tube, pilot side
- Static pressure crossover system
- Air Data Computer

1) If glass cockpit instrumentation is chosen as optional equipment, these standard instruments are deleted and an altimeter (2") and an airspeed indicator (2") as back-up instruments are added.

POWER PLANT

- Two Pratt & Whitney PW206B2 turbine engines or Two Turbomeca ARRIUS 2B2 turbine engines
- These 2 engines are equipped with:
 - fire detectors
 - Full Authority Digital Engine Control (FADEC)
 - chip detectors with quick-disconnect plugs
 - overspeed protection system
- Twin-engine OEI-training mode
- Oil cooling and lubricating system with thermostatic valve
- Crash resistant fuel system with a flexible bladder-type fuel main tank and supply tank (split into two sections)
- Automatically controlled variable rotor speed system
- Lockable Fuel tank filler door,

TRANSMISSION SYSTEM

- Two stage main gearbox
- Chip detector with quick-disconnect plug (main gearbox)
- Redundant oil cooling and lubrication system
- Main gearbox attachment with Anti-Resonance Isolation System (ARIS)
- Free wheel assemblies in the engine input drives
- Tail rotor drive shaft
- Tail rotor gearbox with splash lubrication
- Chip detector with quick-disconnect plug (tail rotor gearbox)

ROTOR AND FLIGHT CONTROLS

- Bearingless Main Rotor system (BMR), consisting of:
 - One piece Rotor head/mast
 - Four fiber-reinforced composite main rotor blades with anti-erosion strips, control cuff, elastomeric lead-lag dampers and special blade tip painting
- Main rotor control with dual hydraulic boost system
- Electrical trim system (cyclic)
- Basic provisions for an easy integration of a track and balance system
- Fenestron-type tail rotor with ten metal blades (asymmetric blade spacing) and stator
- Tail rotor gearbox cover
- Tail rotor control system with flexball cable and single hydraulic booster
- Yaw-SAS (Stability Augmentation System)
- Mast moment indication system

ELECTRICAL INSTALLATION

- Two starter/generators (2 x 160 A, 28 VDC)
- Nickel-Cadmium battery, (24 V, 17 Ah)
 - External power connector (STANAG 3302)
 - Power distribution system:
 - Two primary busbars
 - Two shedding busbars
 - Two essential busbars
 - Two high load busbars (80 A) - for optional equipment only
 - Two high power busbars (200 A)
 - Battery bus
 - One utility receptacle in LH side of cargo compartment (28VDC, 10A)
 - Lighting:
 - Anti-collision warning light (red flashing)
 - Fixed, nose-mounted landing light (250 W)
 - Three position lights (red, green, white)
 - Adjustable instrument lighting
 - One utility light in the cockpit
 - 5 spot-lights in the cabin
 - One light in cargo compartment RH side

GROUND HANDLING KIT¹

- Two ground-handling wheels
- Basic aircraft covers (short time)
- Main rotor blade tie-down lash bags
- Oil drain hoses
- Fuel tank drain device
- Keys for cockpit doors, cabin doors, baggage compartment doors and tank flap (one-key system)
- Battery key
- Lifting points

DOCUMENTATION (in English)

- One Flight Manual²
- One Pilots-Checklist, revision service for five years¹
- One Logbook¹ (only paper, CD ROM on demand)
- One Historical Record² (only paper, CD ROM on demand)
- One CD-ROM^{1,2} including AMM³, SDS³, WDM³, IPC, MSM
- One additional Master Servicing Manual (MSM)^{1,2} on paper
- One Service Bulletin Catalogue (SB)^{1,2} per contract, on paper
- One List of Applicable Publications (LOAP)^{1,2}, on paper
- One Avionics Manual (for avionics installed by Eurocopter)^{1,3} (on paper)
- Engine Documentation¹, furnished by supplier, including:
 - Maintenance Manual
 - Illustrated Parts Catalogue (IPC)
- Service Bulletins

¹ Weight not included in the standard helicopter empty weight

² Revision service included as long as the aircraft is operational

³ Customized documentation

4 Basic configuration choice

Selection of a PINAO package

Please select your PINAO code according to your operational needs by using the following table:

| Pilot | P | Single | Dual | Single/Dual |
|-----------------------|---|--------|-------|-------------|
| | | 1 | 2 | 3 |
| VFR/IFR | I | VFR | IFR | |
| | | 0 | 1 | |
| Day/night | N | day | night | |
| | | 0 | 1 | |
| Cat. A | A | no | yes | |
| | | 0 | 1 | |
| JAR-OPS 3 equipment * | O | no | yes | |
| | | 0 | 1 | |

* This offered equipment package is derived from JAR-OPS 3 Amendment 3. It does not cover oxygen equipment and equipment required for over water operations. As the national operating rule may differ from the JAR-OPS 3 Amendment 3, the operator has to contact its national authority to assure that the planned equipment configuration is acceptable for the intended kind of operation.




| P | I | N | A | O |
|---|---|---|---|---|
| | | | | |

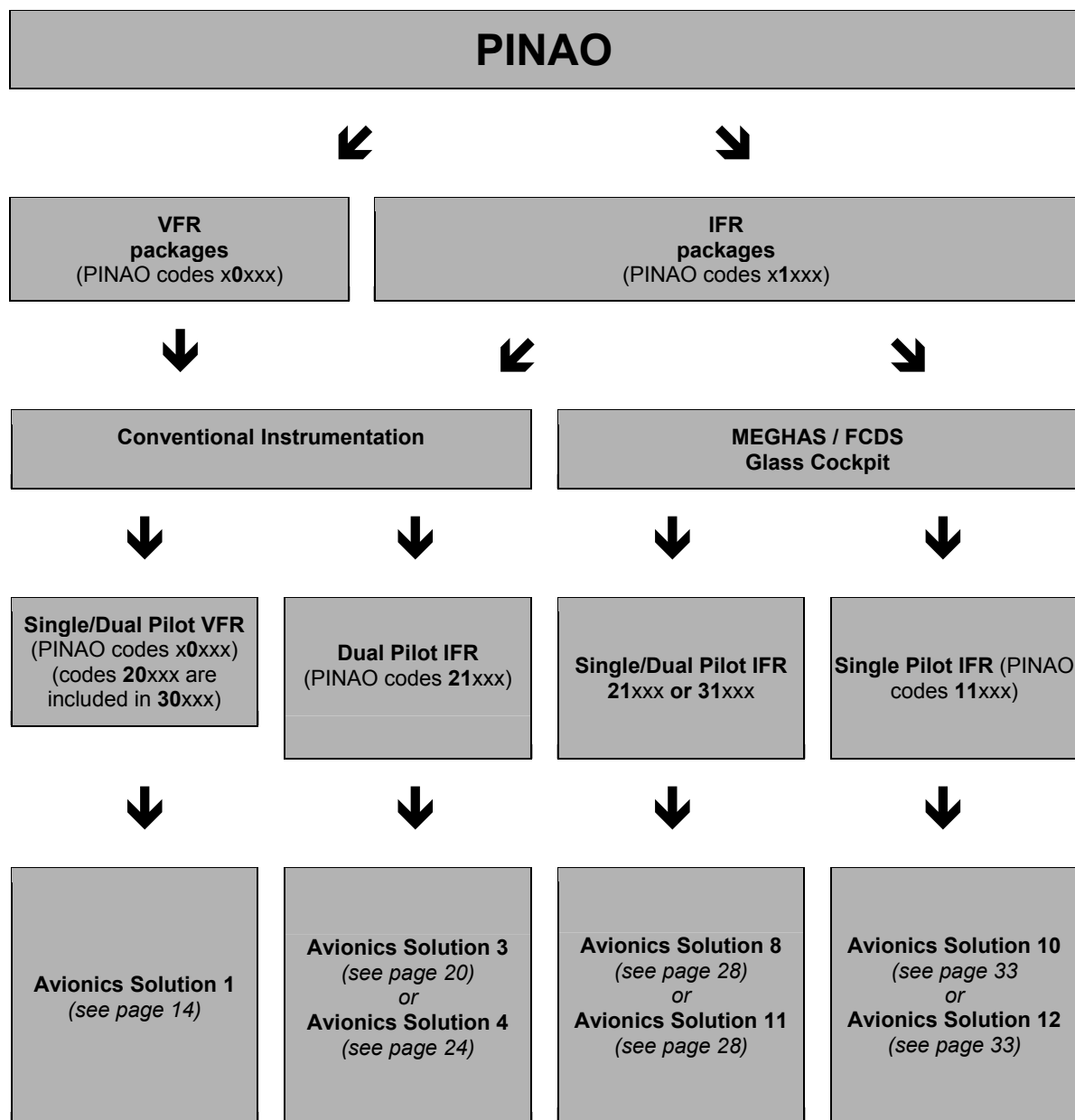
Use this code to find your required "PINAO" packages on the following pages.

- As a general guidance, use the diagram on the next page
- One PINAO code may lead to different PINAO packages

IMPORTANT NOTES:

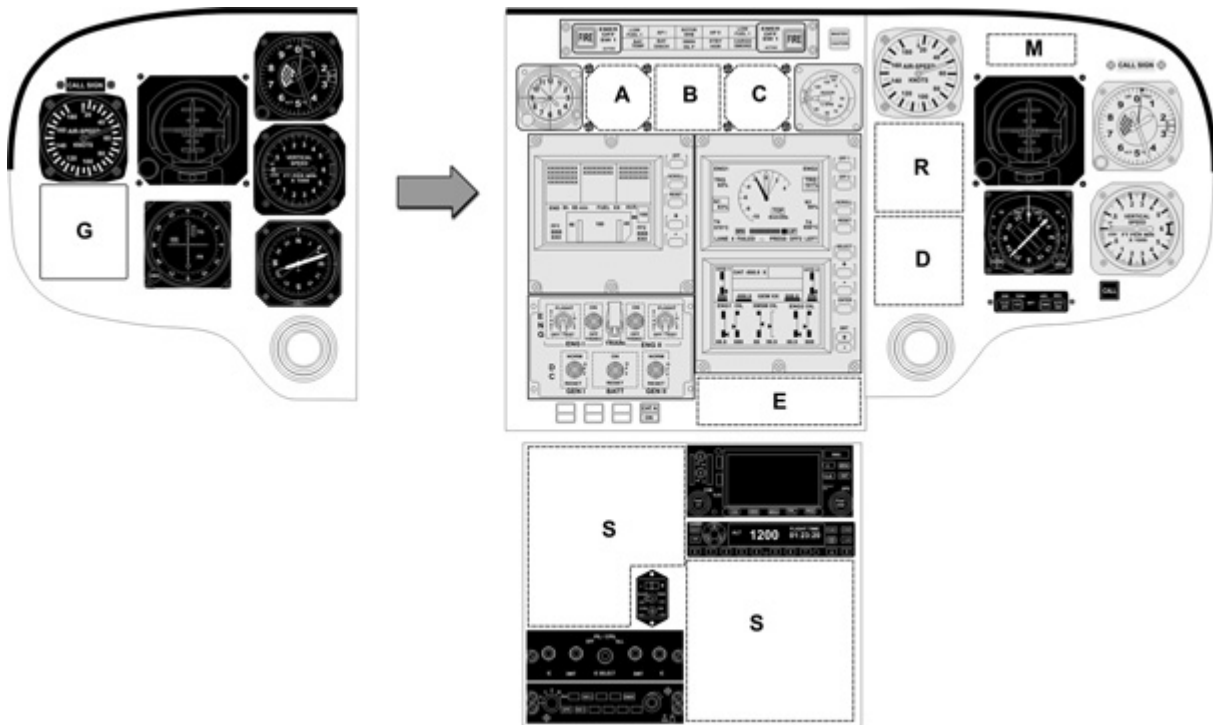
- For IFR, there is no difference between "day" and "night". Therefore only IFR "night" PINAO packages are listed.
- All possible PINAO codes are listed in the following pages.
- Weight margin in this chapter: $\pm 3\%$
- For all intercom systems, the following impedances are standard: LOW IMPEDANCE → Microphone: $5\ \Omega$ (dynamic) / Headset: $8\ \Omega$ (military - Eurocopter typical)
- Symbol  shown beside an item denotes some constraints (see table on page 55)

Use this diagram to find the appropriate Avionics Solution based on your individual PINAO selection.



4.1 VFR packages (based on Avionics Solution 1)

4.1.1 Instrument panel overview



Additional space:

- A - for 2" back-up airspeed indicator (used in MEGHAS/FCDS "Glass cockpit" solutions)
- B - for 2" standby horizon AI 804 DC
- C - for 2" back-up altimeter (used in MEGHAS/FCDS "Glass cockpit" solutions)
- D - e.g. for 3" RMI
- E - e.g. for DME or ELT remote control
- G - e.g. for 2nd gyro 205 1BL (Goodrich)
- M - e.g. for marker lights
- R - e.g. for 3" radar altimeter indicator (KNI 416)
- S - e.g. for 2nd GPS/COM/NAV GNS430 or other equipment

4.1.2 Content of Avionics Solution 1 (basic for all VFR PINAO packages)

| <i>Document reference</i> | <i>Commercial reference</i> | <i>Title</i> |
|---------------------------------|-----------------------------|---|
| 08-00001-A | L2300-001-04 | Avionics Solution 1, consisting of: |
| <i>Intercom System</i> | | |
| 08-16053-B | L2341-192-01 | Audio/Comm. control system AS 3100-12 (Becker), pilot |
| 08-16053-B | L2341-293-01 | Intercom amplifier IC 3100-4 (Becker) |
| <i>Transponder</i> | | |
| 08-22026-B | L2325-092-12 | Transponder (Mode S) GTX 330 (Garmin) |
| <i>Radio Switch</i> | | |
| 08-29003-B | L2480-090-01 | Avionics/Radio master switches (Special Eurocopter) |
| <i>GPS/NAV/COM</i> | | |
| 08-43018-B | L3442-092-00 | GPS / NAV / COM GNS 430, pilot (Garmin) with I-panel annunciation/switch unit MD 41 (MidContinent) |
| <i>Conventional instruments</i> | | |
| 08-51012-A | L3425-092-02 | 4" Artificial horizon GH14-391, pilot (Honeywell) |
| 08-52013-A | L3421-092-02 | Gyro Magnetic Heading System KCS 55 A (Honeywell) incl. KG-102A, KMT-112, KA-51B with HSI KI-525A, pilot |
| <i>Miscellaneous</i> | | |
| 08-99000-A | L0000-150-01 | Avionics Solution 1 interconnection / wiring |

4.1.3 Minimum required equipment

| Minimum required equipment for Avionics Solution 1 – Single pilot | | | | | PINA0 | | | | | | |
|---|----------------------|---|--|------|--------------------|--------------------|--------------------|--------------------|-------|--------------------|--------------------|
| Document reference | Commercial reference | Title | Weight (margin $\pm 3\%$) kg lb | | 10000 ⁴ | 10001 ⁴ | 10010 ⁴ | 10011 ⁴ | 10100 | 10110 ⁴ | 10111 ⁴ |
| 05-03007-C | L2562-001-00 | First aid kit ⁵ | 1.3 | 2.9 | X | | X | | | | X |
| 05-22008-C | L2621-001-00 | Engine fire extinguishing system | 3.6 | 7.9 | | | X | X | | X | X |
| 05-33001-B | L3113-001-00 | Slant panel | 0.8 | 1.8 | X | X | X | X | X | X | X |
| 05-33002-B | L3113-004-00 | Center console | 2.2 | 4.9 | X | X | X | X | X | X | X |
| 05-41004-C | L2104-100-00 | Bleed air heating system ⁶ | 6.6 | 14.6 | X | X | X | X | X | X | X |
| 05-44002-B | L2122-001-00 | Ventilation extruder without copilot I-panel extension | 0.3 | 0.7 | X | X | X | X | X | X | X |
| 05-61010-B | L2433-003-00 | Battery, type (Saft) ULM, 27 Ah, 24 V instead of standard battery | 8.2 | 18.1 | | X | X | X | | X | X |
| 05-62010-C | L2420-002-00 | AC System (350VA) | 3.2 | 7.1 | X | X | X | X | X | X | X |
| 05-63003-B | L2432-001-00 | Starter/generators (2 x 200 A, 28 V DC), instead of standard generators | 3.6 | 7.9 | X | X | X | X | X | X | X |
| 06-45023-B | L3343-003-00 | Landing & search light, 450 W | 3.4 | 7.5 | | | | | X | X | X |
| 06-67044-B | L2563-801-06 | ELT C406-N HM (Artex) incl. NAV. opt. | 3.3 | 7.3 | | X | | X | | | X |
| 08-00001-A | L2300-001-04 | Avionics Solution 1 | 39.7 | 87.5 | X | X | X | X | X | X | X |
| 08-21014-C | L3441-090-04 | Radar altimeter KRA 405B (Honeywell) | 3.5 | 7.7 | | | X | X | | X | X |
| 08-21014-C | L3441-092-03 | Radar altimeter indicator KNI 416 (Honeywell) | 1.2 | 2.6 | | | X | X | | X | X |
| 08-51013-B | L3425-806-51 | 2" std-by horizon AI 804 DC (Goodrich) with emergency battery | 6.6 | 14.6 | | | X | X | | X | X |

⁴ For VFR flights on routes not navigated by reference to visual landmarks, a 2nd GPS/NAV/COM GNS430 (see possible add-ons) is required.

⁵ First aid kit complies with german regulation rules 1. DV LuftBO paragraph 5(2). Type of operation, procedures or regulations may require a different/specific first aid kit.

⁶ For helicopters dedicated for EMS select "Bleed air heating system: EMS version L2104-003-00" (05-41004-C) (7.0 kg / 15.4 lb.)

| Minimum required equipment for Avionics Solution 1- Single/Dual pilot | | | | PINAO | | | | | | | |
|---|----------------------|--|-------------------------------|-------|------|------|------|------|-------|-------|-------|
| Document reference | Commercial reference | Title | Weight (margin $\pm 3\%$) | | 3000 | 3001 | 3010 | 3011 | 30100 | 30110 | 30111 |
| | | | kg | lb | | | | | | | |
| 05-03007-C | L2562-001-00 | First aid kit ⁸ | 1.3 | 2.9 | X | | X | | | | X |
| 05-22008-C | L2621-001-00 | Engine fire extinguishing system | 3.6 | 7.9 | | | X | X | | X | X |
| 05-33001-B | L3113-001-00 | Slant panel | 0.8 | 1.8 | X | X | X | X | X | X | X |
| 05-33002-B | L3113-004-00 | Center console | 2.2 | 4.9 | X | X | X | X | X | X | X |
| 05-37016-C | L6701-001-00 | Copilot flight controls | 6.0 | 13.2 | X | X | X | X | X | X | X |
| 05-38010-B | L3111-001-00 | 10" copilot instrument panel with glare shield | 2.9 | 6.4 | X | X | X | X | X | X | X |
| 05-41004-C | L2104-100-00 | Bleed air heating system ⁹ | 6.6 | 14.6 | X | X | X | X | X | X | X |
| 05-61010-B | L2433-003-00 | Battery, type (Saft) ULM, 27 Ah, 24 V instead of standard battery | 8.2 | 18.1 | X | X | X | X | X | X | X |
| 05-62010-C | L2420-003-00 | Dual AC System (2 x 350VA) | 6.6 | 14.6 | X | X | X | X | X | X | X |
| 05-63003-B | L2432-001-00 | Starter/generators (2 x 200 A, 28 V DC), instead of standard generators | 3.6 | 7.9 | X | X | X | X | X | X | X |
| 06-45023-B | L3343-003-00 | Landing & search light, 450 W | 3.4 | 7.5 | | | | | X | X | X |
| 06-67044-B | L2563-801-06 | ELT C406-N HM (Artex) incl. NAV. opt. | 3.3 | 7.3 | X | | X | | | | X |
| 08-00001-A | L2300-001-04 | Avionics Solution 1 | 39.7 | 87.5 | X | X | X | X | X | X | X |
| 08-16053-B | L2341-191-01 | Audio/Comm. control system AS 3100-12 (Becker), copilot | 2.0 | 4.4 | X | X | X | X | X | X | X |
| 08-21014-C | L3441-090-04 | Radar Altimeter KRA 405B (Honeywell) | 3.5 | 7.7 | | | X | X | | X | X |
| 08-21014-C | L3441-092-03 | Radar altimeter indicator KNI 416 (Honeywell) | 1.2 | 2.6 | | | X | X | | X | X |
| 08-51012-A | L3425-091-02 | 4" artificial horizon GH14-391 (Honeywell), copilot | 2.5 | 5.5 | X | X | X | X | X | X | X |
| 08-51013-B | L3425-806-51 | 2" std-by horizon AI 804 DC (Goodrich) with emergency battery | 6.6 | 14.6 | | | X | X | | X | X |
| 08-52010-A | L3421-091-02 | 2nd directional Gyro (3" unslaved indicator) 205 1BL on copilot side (Goodrich) | 1.4 | 3.1 | | | X | | | | X |
| 08-54001-C | L3411-001-00 | Copilot pitot static system | 1.5 | 3.3 | X | X | X | X | X | X | X |
| 08-60003-A | L3412-002-00 | Copilot 3" instruments (airspeed indicator, altimeter, vertical speed indicator; United Instruments) | 1.7 | 3.8 | X | X | X | X | X | X | X |
| 08-61010-B | L3166-091-04 | RMI KI 229 (Honeywell), copilot | 2.2 | 4.9 | X | X | X | X | X | X | X |
| 08-61011-A | L3167-091-02 | CDI KI 204 (Honeywell), copilot | 1.0 | 2.2 | X | X | X | X | X | X | X |



⁷ For VFR flights on routes not navigated by reference to visual landmarks, a 2nd GPS/NAV/COM GNS430 (see possible add-ons) is required.

⁸ First aid kit complies with german regulation rules 1. DV LuftBO paragraph 5(2). Type of operation, procedures or regulations may require a different/specific first aid kit.

⁹ For helicopters dedicated for EMS select "Bleed air heating system: EMS version L2104-003-00" (05-41004-C) (7.0 kg / 15.4 lb.)

4.1.4 Possible add-ons

| Possible add-ons for Avionics Solution 1 – Single Pilot | | | | | | PINA0 | | | | | |
|---|----------------------|--|--------------------------|------|------|-------|-------|-------|-------|-------|-------|
| Document reference | Commercial reference | Title | Weight (margin ± 3 %) | | 1000 | 1001 | 10010 | 10011 | 10100 | 10110 | 10111 |
| | | | kg | lb | | | | | | | |
| 06-67044-B | L2563-801-06 | ELT C406-N HM (Artex) incl. NAV. opt. | 3.3 | 7.3 | X | | X | | X | X | |
| 08-16053-B | L2341-191-01 | Audio/Comm. control system AS 3100-12 (Becker), copilot | 2.0 | 4.4 | X | X | X | X | X | X | X |
| 08-21014-C | L3441-090-04 | Radar altimeter KRA 405B (Honeywell) | 3.5 | 7.7 | X | X | | | X | | |
| | L3441-092-03 | Radar altimeter indicator KNI 416 (Honeywell) | 1.2 | 2.6 | | | | | | | |
| 08-25014-A | L3455-092-03 | Distance Measuring Equipment DME KN 63 | 2.0 | 4.4 | | | | | | | |
| | L3169-092-02 | Distance Measuring Equipment DME indicator KDI 572 (Honeywell) | 0.8 | 1.8 | X | X | X | X | X | X | X |
| 08-26010-B | L3431-092-02 | Marker Beacon receiver / lights KR 21 (Honeywell) | 0.9 | 2.0 | X | X | X | X | X | X | X |
| 08-35007-B | L2327-001-01 | Traffic Advisory System TAS 9900BX (Ryan) | 9.5 | 20.9 | X | X | X | X | X | X | X |
| 08-35007-B | L3442-881-00 | Traffic Advisory System TAS 9900BX interface with GNS430 | 0.2 | 0.4 | X | X | X | X | X | X | X |
| 08-51013-B | L3425-806-51 | 2" std-by horizon AI 804 DC (Goodrich) with emergency battery | 6.6 | 14.6 | X | X | | | X | | |
| 08-71002-C | L2217-001-10 | VFR SAS (VFR pitch/roll Stability Augmentation System) | 5.6 | 12.3 | X | X | X | X | X | X | X |

| Possible add-ons for Avionics Solutions 1 – Dual pilot Single - Dual pilot | | | | | | PINA0 | | | | | | | |
|--|-------------------------|--|-------------------------------|------|------|-------|------|------|-------|-------|-------|---|---|
| Document reference | Commercial reference | Title | Weight (margin $\pm 3\%$) | | 3000 | 3001 | 3010 | 3011 | 30100 | 30110 | 30111 | | |
| | | | kg | lb | | | | | | | | | |
| 06-67044-B | L2563-801-06 | ELT C406-N HM (Artex) incl. NAV. opt. | 3.3 | 7.3 | X | | X | | X | X | | | |
| 08-21014-C | L3441-090-04 | Radar altimeter KRA 405B (Honeywell) | 3.5 | 7.7 | X | X | | | X | | | | |
| | L3441-092-03 | Radar altimeter indicator KNI 416 (Honeywell) | 1.2 | 2.6 | | | | | | | | | |
| 08-25014-A | L3455-092-03 | Distance Measuring Equipment DME KN 63 | 2.0 | 4.4 | | | | | | | | | |
| | L3169-092-02 | Distance Measuring Equipment DME indicator KDI 572 (Honeywell) | 0.8 | 1.8 | X | X | X | X | X | X | X | X | X |
| 08-26010-B | L3431-092-02 | Marker Beacon receiver / lights KR 21 (Honeywell) | 0.9 | 2.0 | X | X | X | X | X | X | X | X | X |
|  08-35007-B | L2327-001-01 | Traffic Advisory System TAS 9900BX (Ryan) | 9.5 | 20.9 | X | X | X | X | X | X | X | X | X |
|  08-35007-B | L3442-881-00- | Traffic Advisory System TAS 9900BX interface with GNS430 | 0.2 | 0.4 | X | X | X | X | X | X | X | X | X |
| 08-43018-B | L3442-091-00 | GPS/NAV/COM GNS 430 (Garmin), copilot, incl. annunciator/switch MD 41 (MidContinent) | 8.3 | 18.3 | | X | X | X | X | X | X | X | X |
| 08-61011-A | L3167-091-01 | CDI KI206 instead of CDI KI 204 (Honeywell) | 1.3 | 2.9 | | | | | | | | | |
| 08-51013-B | L3425-806-51 | 2" std-by horizon AI 804 DC (Goodrich) with emergency battery | 6.6 | 14.6 | X | X | | | X | | | | |
| 08-71002-C | L2217-001-10 | VFR SAS (VFR pitch/roll Stability Augmentation System) | 5.6 | 12.3 | X | X | X | X | X | X | X | X | X |

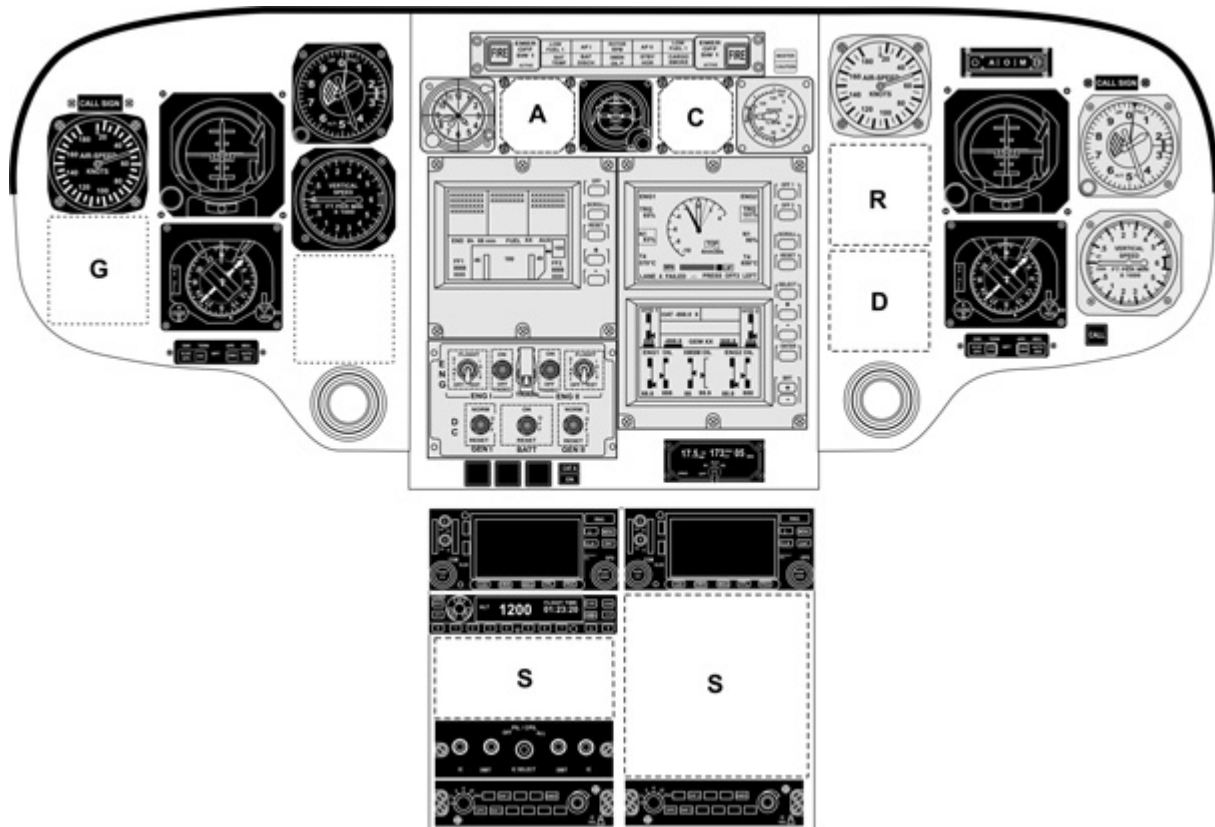
4.1.5 On request items

- Multifunction display KMD 850 (Honeywell) for weather radar or digital map
- Color weather radar RDR2000 (Honeywell) on KMD850 display
- Moving Map EURONAV IV - RN6 (Euro Avionics)

4.1.6 Further avionics add-ons see chapter 6 page 50

4.2 Dual pilot IFR packages, conventional instrumentation, basic (based on Avionics Solution 3)

4.2.1 Instrument panel overview



Additional space:

- A - for 2" back-up airspeed indicator (used in MEGHAS/FCDS "Glass cockpit" solutions)
- C - for 2" back-up altimeter (used in MEGHAS/FCDS "Glass cockpit" solutions)
- D - e.g. for 3" RMI or 3" CDI
- G - e.g. for 2nd gyro 205 1BL (Goodrich)
- R - e.g. for 3" radar altimeter indicator (KNI 416)
- S - e.g. for tactical or other optional equipment

4.2.2 Contents of Avionics Solution 3

| <i>Document reference</i> | <i>Commercial reference</i> | <i>Title</i> |
|---------------------------------|-----------------------------|--|
| 08-00002-A | L2300-003-04 | Avionics Solution 3, consisting of: |
| <i>Intercom System</i> | | |
| 08-16053-B | L2341-191-01 | Audio/Comm. control system AS 3100-12 (Becker), copilot |
| 08-16053-B | L2341-192-01 | Audio/Comm. control system AS 3100-12 (Becker), pilot |
| 08-16053-B | L2341-293-01 | Intercom amplifier IC 3100-4 (Becker) |
| <i>Transponder</i> | | |
| 08-22026-B | L2325-092-12 | Transponder (Mode S) GTX 330 (Garmin) |
| <i>DME</i> | | |
| 08-25014-A | L3169-092-02 | Distance Measuring Equipment DME indicator KDI 572 (Honeywell) |
| 08-25014-A | L3455-092-03 | Distance measuring equipment DME KN 63 (Honeywell) |
| <i>VOR / ILS /MKR receivers</i> | | |
| 08-26010-B | L3431-092-02 | Marker beacon receiver / lights KR 21(Honeywell) |
| <i>Radio Switch</i> | | |
| 08-29003-B | L2480-090-01 | Avionics/Radio master switches (Special Eurocopter) |
| <i>GPS/NAV/COM</i> | | |
| 08-43018-B | L3442-091-00 | GPS/NAV/COM GNS 430 (Garmin), copilot, incl. annunciator/switch MD 41 (MidContinent) |
| 08-43018-B | L3442-092-00 | GPS / NAV / COM GNS 430 (Garmin), pilot incl. annunciator/switch MD 41 (MidContinent) |
| <i>Conventional instruments</i> | | |
| 08-51012-A | L3425-091-02 | 4" Artificial horizon GH14-391, copilot (Honeywell) |
| 08-51012-A | L3425-092-02 | 4" Artificial horizon GH14-391, pilot (Honeywell) |
| 08-51013-B | L3425-806-51 | 2" std-by horizon AI 804 DC (Goodrich) with emergency battery |
| 08-52014-A | L3421-092-01 | Gyro Magnetic Heading System C14D (Honeywell) |
| 08-60003-A | L3412-002-00 | Copilot 3" instruments (airspeed indicator, altimeter, vertical speed indicator (United Instruments) |
| 08-61012-A | L3165-091-01 | Horizontal Situation Indicator - KPI 552 (Honeywell), copilot |
| 08-61012-A | L3165-092-01 | Horizontal Situation Indicator - KPI 552 (Honeywell), pilot |
| <i>Miscellaneous</i> | | |
| 08-99000-A | L0000-150-03 | Avionics Solution 3 interconnection / wiring |



4.2.3 Minimum required equipment

| Minimum required equipment for Avionics Solution 3 | | | | | PINAO | | |
|--|----------------------|---|--------------------------|-------|-------|-------|-------|
| Document reference | Commercial reference | Title | Weight (margin ± 3 %) | | 21100 | 21110 | 21111 |
| | | | kg | lb | | | |
| 05-03007-C | L2562-001-00 | First aid kit ¹⁰ | 1.3 | 2.9 | | | X |
| 05-22008-C | L2621-001-00 | Engine fire extinguishing system | 3.6 | 7.9 | | X | X |
| 05-33001-B | L3113-001-00 | Slant panel | 0.8 | 1.8 | X | X | X |
| 05-33002-B | L3113-004-00 | Center console | 2.2 | 4.9 | X | X | X |
| 05-37016-C | L6701-001-00 | Copilot flight controls | 6.0 | 13.2 | X | X | X |
| 05-38010-B | L3111-001-00 | 10" copilot instrument panel with glare shield | 2.9 | 6.4 | X | X | X |
| 05-39006-B | L2514-003-01 | Map case in copilot door | 0.5 | 1.1 | X | X | X |
| 05-39007-B | L3111-001-10 | Map case on instrument panel glare shield | 0.6 | 1.3 | X | X | X |
| 05-39008-B | L3113-004-10 | Illuminated chart holder for pilot side | 0.9 | 2.0 | | | X |
| 05-41004-C | L2104-100-00 | Bleed air heating system ¹¹ | 6.6 | 14.6 | X | X | X |
| 05-61010-B | L2433-006-00 | Battery, type "Saft", ULM, 40 Ah, 24 V instead of standard battery | 16.8 | 37.0 | X | X | X |
| 05-62010-C | L2420-003-00 | Dual AC System (2 x 350VA) | 6.6 | 14.6 | X | X | X |
| 05-63003-B | L2432-001-00 | Starter/generators (2 x 200 A, 28 VDC), instead of standard generator | 3.6 | 7.9 | X | X | X |
| 06-45023-B | L3343-003-00 | Landing & search light, 450 W | 3.4 | 7.5 | X | X | X |
| 06-67044-B | L2563-801-06 | ELT C406-N HM (Artex) incl. NAV. opt. | 3.3 | 7.3 | | | X |
| 08-00002-A | L2300-003-04 | Avionics Solution 3 | 78.1 | 172.2 | X | X | X |
| 08-21014-C | L3441-090-04 | Radar altimeter KRA 405B (Honeywell) | 3.5 | 7.7 | | X | X |
| 08-21014-C | L3441-092-03 | Radar altimeter indicator KNI 416 (Honeywell) | 1.2 | 2.6 | | X | X |
| 08-52010-A | L3421-091-02 | 2nd directional Gyro (3" unslaved indicator) 205 1BL on copilot side (Goodrich) | 1.4 | 3.1 | X | X | X |
| 08-54001-C | L3411-001-00 | Copilot pitot static system | 1.5 | 3.3 | X | X | X |
| 08-71002-C | L2217-001-50 | IFR SAS (IFR pitch/roll Stability Augmentation System) | 12.2 | 26.9 | X | X | X |

¹⁰ First aid kit complies with german regulation rules 1. DV LuftBO paragraph 5(2). Type of operation, procedures or regulations may require a different/specific first aid kit.

¹¹ For helicopters dedicated for EMS select "Bleed air heating system: EMS version L2104-003-00" (05-41004-C) (7.0 kg / 15.4 lb.)

4.2.4 Possible add-ons

| Possible add-ons for Avionics Solution 3 | | | | | | PINA0 | | |
|--|----------------------|--|-------------------------------|------|--|-------|-------|-------|
| Document reference | Commercial reference | Title | Weight (margin $\pm 3\%$) | | | 21100 | 21110 | 21111 |
| | | | kg | lb | | | | |
| 06-67044-B | L2563-801-06 | ELT C406-N HM (Artex) incl. NAV. opt. | 3.3 | 7.3 | | X | X | |
| 08-21014-C | L3441-090-04 | Radar altimeter KRA 405B (Honeywell) | 3.5 | 7.7 | | X | | |
| | L3441-092-03 | Radar altimeter indicator KNI 416 (Honeywell) | 1.2 | 2.6 | | | | |
| 08-24015-B | L3452-092-17 | ADF system DFS-43A (Chelton / Wulfsberg) | 9.6 | 21.2 | | X | X | X |
| | L3452-092-08 | ADF control unit CD-432B (Chelton / Wulfsberg) | 1.2 | 2.6 | | | | |
|  08-35007-B | L2327-001-01 | Traffic Advisory System TAS 9900BX (Ryan) | 9.5 | 20.9 | | X | X | X |
|  08-35007-B | L3442-881-00 | Traffic Advisory System TAS 9900BX interface with GNS430 | 0.2 | 0.4 | | X | X | X |

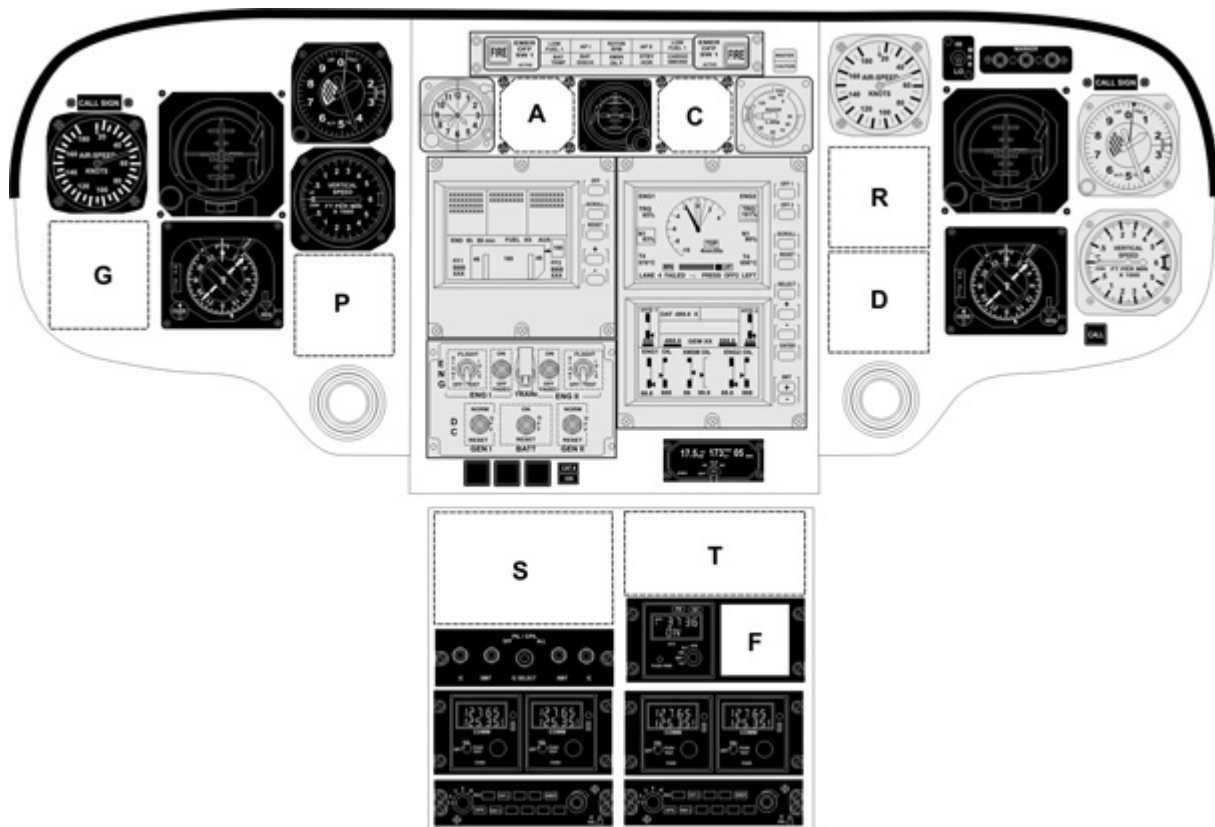
4.2.5 On request items

- Multifunction display KMD 850 (Honeywell) for weather radar or digital map
- Color weather radar RDR2000 (Honeywell) on KMD850 display
- Moving Map EURONAV IV - RN6 (Euro Avionics)

4.2.6 Further avionics add-ons see chapter 6 page 50

4.3 Dual pilot IFR packages, conventional instrumentation, enhanced (based on Avionics Solution 4)

4.3.1 Instrument panel overview




Additional space:

- A - for 2" back-up airspeed indicator (used in MEGHAS/FCDS "Glass cockpit" solutions)
- C - for 2" back-up altimeter (used in MEGHAS/FCDS "Glass cockpit" solutions)
- D - e.g. for 3" RMI or 3" CDI
- F - e.g. for ADF control unit (CD-432B)
- G - e.g. for 2nd gyro 205 1BL (Goodrich)
- P - e.g. for 3" CDI or 3" RMI
- R - e.g. for 3" radar altimeter indicator (KNI 416)
- S / T - e.g. for GPS receiver or other optional equipment

4.3.2 Contents of Avionics Solution 4

| <i>Document reference</i> | <i>Commercial reference</i> | <i>Title</i> |
|----------------------------------|-----------------------------|---|
| 08-00003-A | L2300-004-04 | Avionics Solution 4, consisting of: |
| <i>VHF AM</i> | | |
| 08-11023-B | L2313-091-03 | VHF AM/COM. system, copilot KTR 908/KFS 598 A (Honeywell) |
| 08-11023-B | L2313-092-03 | VHF AM/COM system, pilot KTR 908 / KFS 598A (Honeywell) |
| <i>Intercom System</i> | | |
| 08-16053-B | L2341-191-01 | Audio/Comm. control system AS 3100-12 (Becker), copilot |
| 08-16053-B | L2341-192-01 | Audio/Comm. control system (pilot) AS 3100-12 (Becker) incl. Intercom Select Panel (ICS mode selector) |
| 08-16053-B | L2341-293-01 | Intercom amplifier IC 3100-4 (Becker) |
| <i>Transponder</i> | | |
| 08-22027-B | L2325-092-06 | Transponder (Mode S) MST 67A (Honeywell) |
| 08-22027-B | L2325-092-15 | Transponder control unit PS 578A (Honeywell) |
| <i>DME</i> | | |
| 08-25014-A | L3169-092-02 | Distance Measuring Equipment DME indicator KDI 572 (Honeywell) |
| 08-25025-A | L3455-092-01 | Distance measuring equipment KDM 706 A (Honeywell) |
| <i>VOR / ILS / MKR receivers</i> | | |
| 08-26025-B | L3432-091-03 | VOR/ILS/MKR Navigation system, copilot KNR 634 A / KFS 564 A (Honeywell) |
| 08-26025-B | L3432-092-03 | VOR/ILS/MKR Navigation system, pilot KNR 634 A / KFS 564 A (Honeywell) |
| 08-26028-A | L3431-092-01 | Marker beacon lights KA 35 A (Honeywell) |
| <i>Radio Switch</i> | | |
| 08-29003-B | L2480-090-01 | Avionics/Radio master switches (Special Eurocopter) |
| <i>Conventional instruments</i> | | |
| 08-51012-A | L3425-091-02 | 4" Artificial horizon GH14-391 (Honeywell), copilot |
| 08-51012-A | L3425-092-02 | 4" artificial horizon GH14-391 (Honeywell), pilot |
| 08-51013-B | L3425-806-51 | 2" std-by horizon AI 804 DC (Goodrich) with emergency battery |
| 08-52014-A | L3421-092-01 | Gyro Magnetic Heading System C14D (Honeywell) |
| 08-60003-A | L3412-002-00 | Copilot 3" instruments (airspeed indicator, altimeter, vertical speed indicator (United Instruments) |
| 08-61012-A | L3165-091-01 | Horizontal Situation Indicator (HSI) KPI 552, copilot (Honeywell) |
| 08-61012-A | L3165-092-01 | Horizontal Situation Indicator (HIS) KPI 552, pilot (Honeywell) |
| <i>Miscellaneous</i> | | |
| 08-99000-A | L0000-150-04 | Avionics Solution 4 interconnection / wiring |




4.3.3 Minimum required equipment

| Minimum required equipment for Avionics Solution 4 | | | | | | PINA0 | | |
|--|----------------------|---|-------------------------------|-------|---|-------|-------|-------|
| Document reference | Commercial reference | Title | Weight (margin $\pm 3\%$) | | | 21100 | 21110 | 21111 |
| | | | kg | lb | | | | |
| 05-03007-C | L2562-001-00 | First aid kit ¹² | 1.3 | 2.9 | | | | X |
| 05-22008-C | L2621-001-00 | Engine fire extinguishing system | 3.6 | 7.9 | | | X | X |
| 05-33001-B | L3113-001-00 | Slant panel | 0.8 | 1.8 | X | X | X | X |
| 05-33002-B | L3113-004-00 | Center console | 2.2 | 4.9 | X | X | X | X |
| 05-34002-B | L2576-001-00 | Avionics compartment | 4.2 | 9.3 | X | X | X | X |
| 05-37016-C | L6701-001-00 | Copilot flight controls | 6.0 | 13.2 | X | X | X | X |
| 05-38010-B | L3111-001-00 | 10" copilot instrument panel with glare shield | 2.9 | 6.4 | X | X | X | X |
| 05-39006-B | L2514-003-01 | Map case in copilot door | 0.5 | 1.1 | X | X | X | X |
| 05-39007-B | L3111-001-10 | Map case on instrument panel glare shield | 0.6 | 1.3 | X | X | X | X |
| 05-39008-B | L3113-004-10 | Illuminated chart holder for pilot side | 0.9 | 2.0 | | | | X |
| 05-41004-C | L2104-100-00 | Bleed air heating system ¹³ | 6.6 | 14.6 | X | X | X | X |
| 05-61010-B | L2433-006-00 | Battery, type "Saft", ULM, 40 Ah, 24 V instead of standard battery | 16.8 | 37.0 | X | X | X | X |
| 05-62010-C | L2420-003-00 | Dual AC System (2 x 350VA) | 6.6 | 14.6 | X | X | X | X |
| 05-63003-B | L2432-001-00 | Starter/generators (2 x 200 A, 28 VDC), instead of standard generator | 3.6 | 7.9 | X | X | X | X |
| 06-45023-B | L3343-003-00 | Landing & search light, 450 W | 3.4 | 7.5 | X | X | X | X |
|  06-67044-B | L2563-801-06 | ELT C406-N HM (Artex) incl. NAV. opt. | 3.3 | 7.3 | | | | X |
| 08-00003-A | L2300-004-04 | Avionics Solution 4 | 78.3 | 172.6 | X | X | X | X |
| 08-21014-C | L3441-090-04 | Radar altimeter KRA 405B (Honeywell) | 3.5 | 7.7 | | | X | X |
| 08-21014-C | L3441-092-03 | Radar altimeter indicator KNI 416 (Honeywell) | 1.2 | 2.6 | | | X | X |
| 08-52010-A | L3421-091-02 | 2nd directional Gyro (3" unslaved indicator) 205 1BL on copilot side (Goodrich) | 1.4 | 3.1 | X | X | X | X |
| 08-54001-C | L3411-001-00 | Copilot pitot static system | 1.5 | 3.3 | X | X | X | X |
| 08-71002-C | L2217-001-50 | IFR SAS (IFR pitch/roll Stability Augmentation System) | 12.2 | 26.9 | X | X | X | X |

¹² First aid kit complies with German regulation rules 1. DV LuftBO paragraph 5(2). Type of operation, procedures or regulations may require a different/specific first aid kit.

¹³ For helicopters dedicated for EMS select "Bleed air heating system: EMS version L2104-003-00" (05-41004-C) (7.0 kg / 15.4 lb.)

4.3.4 Possible add-ons

| Possible add-ons for Avionics Solution 4 | | | | | | PINA0 | | |
|--|----------------------|---|-------------------------------|------|--|-------|-------|-------|
| Document reference | Commercial reference | Title | Weight (margin $\pm 3\%$) | | | 21100 | 21110 | 21111 |
| | | | kg | lb | | | | |
|  06-67044-B | L2563-801-06 | ELT C406-N HM (Artex) incl. NAV. opt. | 3.3 | 7.3 | | X | X | |
| 08-21014-C | L3441-090-04 | Radar altimeter KRA 405B (Honeywell) | 3.5 | 7.7 | | X | | |
| | L3441-092-03 | Radar altimeter indicator KNI 416 (Honeywell) | 1.2 | 2.6 | | | | |
| 08-24015-B | L3452-092-17 | ADF system DFS-43A (Chelton / Wulfsberg) | 9.6 | 21.2 | | X | X | X |
| | L3452-092-08 | ADF control unit CD-432B (Chelton / Wulfsberg) | 1.2 | 2.6 | | | | |
|  08-35007-B | L2327-001-01 | Traffic Advisory System TAS 9900BX (Ryan) | 9.5 | 20.9 | | X | X | X |
|  08-35007-B | L2327-001-11 | Traffic Advisory System TAS 9900 BX indicator 3" MHD (Ryan) | 1.5 | 3.3 | | X | X | X |
| 08-43017-B | L3442-092-12 | GPS Nav. system 2101 I/O Approach Plus (Free Flight) | 3.2 | 7.1 | | X | X | X |
| 08-63009-A | L3442-092-80 | GPS indication on HSI (KPI 552) | 2.9 | 6.4 | | X | X | X |

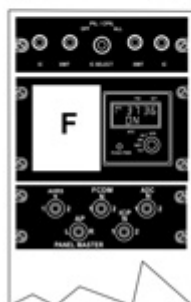
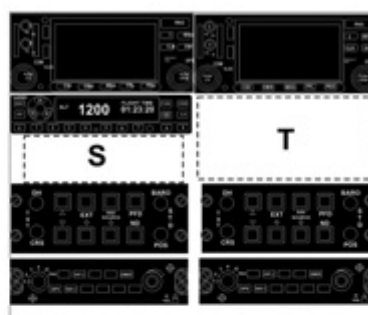
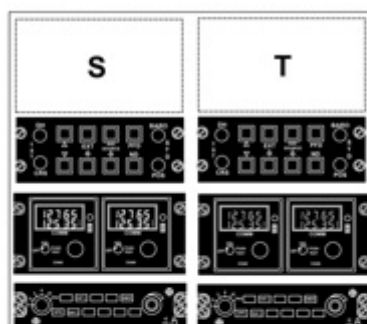
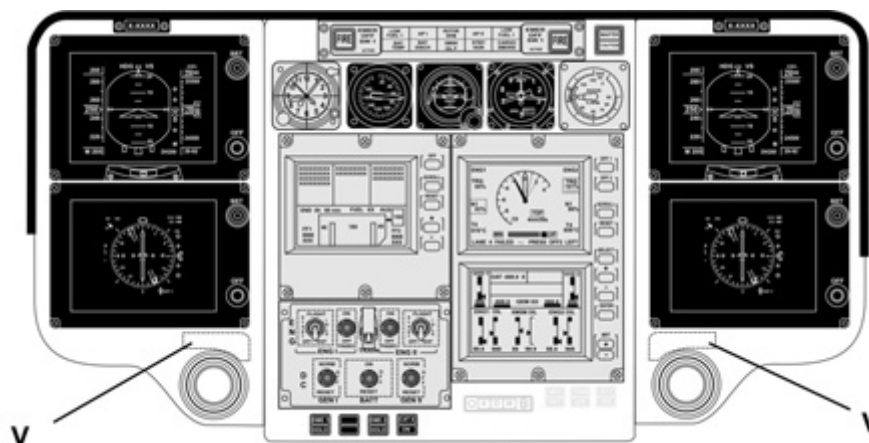
4.3.5 On request items

- Multifunction display KMD 850 (Honeywell) for weather radar or digital map
- Color weather radar RDR2000 (Honeywell) on KMD850 display
- Moving Map EURONAV IV - RN6 (Euro Avionics)

4.3.6 Further avionics add-ons see chapter 6 page 50

4.4 Dual Pilot or Single/Dual Pilot IFR Glass Cockpit, based on Avionics Solution 8 or 11

4.4.1 Instrument panel overview



Avionics Solution 8

Avionics Solution 11

Additional space:

F - e.g. for ADF control unit (CD-432B)

S - e.g. or other optional equipment

T - e.g. for autopilot (DAFCS) control unit or other optional equipment

V - for Video Radar Unit

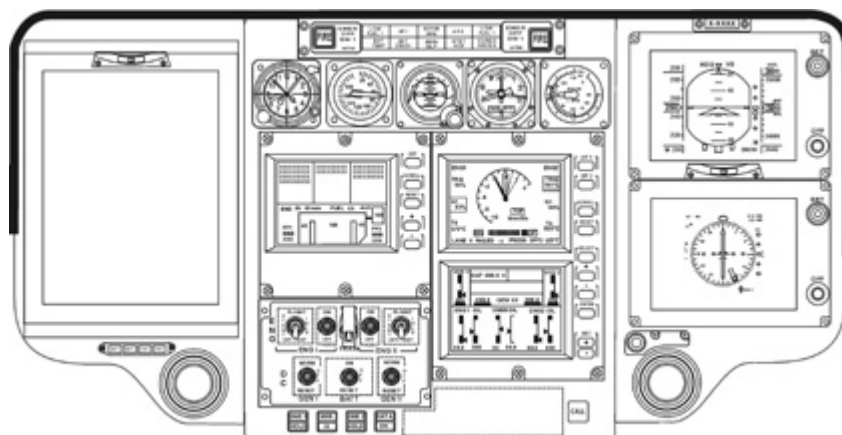
4.4.2 Contents of Avionics Solutions 8 and 11

| <i>Document reference</i> | <i>Commercial reference</i> | <i>Title</i> |
|---------------------------------|-----------------------------|---|
| 08-00005-A | L2300-008-04 | Avionics Solution 8, consisting of |
| <i>VHF AM</i> | | |
| 08-11022-B | L2313-091-08 | VHF AM/COM system, copilot VCS-40 A (Chelton / Wulfsberg) |
| 08-11022-B | L2313-091-13 | Control unit CD 402 B, copilot (Chelton / Wulfsberg) |
| 08-11022-B | L2313-092-07 | VHF AM/COM system, pilot VCS-40 A (Chelton / Wulfsberg) |
| 08-11022-B | L2313-092-13 | Control unit CD 402 B, pilot (Chelton / Wulfsberg) |
| <i>Intercom System</i> | | |
| 08-16053-B | L2341-191-01 | Audio/Comm. control system AS 3100-12 (Becker), copilot |
| 08-16053-B | L2341-192-01 | Audio/Comm. control system (pilot) AS 3100-12 (Becker) incl. Intercom Select Panel (ICS mode selector) |
| 08-16053-B | L2341-293-01 | Intercom amplifier IC 3100-4 (Becker) |
| <i>Transponder</i> | | |
| 08-22027-B | L2325-092-06 | Transponder (Mode S) MST 67A (Honeywell) |
| 08-22027-B | L2325-092-15 | Transponder control unit PS 578A (Honeywell) |
| <i>DME</i> | | |
| 08-25022-B | L3455-090-02 | Distance measuring equipment DMS-44A (Chelton / Wulfsberg) |
| <i>VOR/ILS/MKR receivers</i> | | |
| 08-26024-B | L3432-091-06 | VOR/ILS/MKR Navigation system, copilot VNS-41 A (Chelton / Wulfsberg) |
| 08-26024-B | L3432-091-09 | Control unit CD 412 B, copilot (Chelton / Wulfsberg) |
| 08-26024-B | L3432-092-07 | VOR/ILS/MKR Navigation system, pilot VNS-41 A (Chelton / Wulfsberg) |
| 08-26024-B | L3432-092-12 | Control unit CD 412 B, pilot (Chelton / Wulfsberg) |
| <i>Radio switch</i> | | |
| 08-29003-B | L2480-090-01 | Avionics/Radio master switches (Special Eurocopter) |
| <i>Conventional instruments</i> | | |
| 08-51013-B | L3425-806-51 | 2" std-by horizon AI 804 DC (Goodrich) with emergency battery |
| <i>Display system</i> | | |
| 08-65003-B | L3161-090-09 | MEGHAS - Flight Control Display System (FCDS) - Dual (4xSMD45) |
| <i>Miscellaneous</i> | | |
| 08-99000-A | L0000-150-08 | Avionics Solution 8 interconnection / wiring |



| Document reference | Commercial reference | Title |
|---------------------------------|----------------------|---|
| 08-00022-A | L2300-011-02 | Avionics Solution 11, consisting of: |
| <i>Intercom System</i> | | |
| 08-16053-B | L2341-191-01 | Audio/Comm. control system AS 3100-12 (Becker), copilot |
| 08-16053-B | L2341-192-01 | Audio/Comm. control system (pilot) AS 3100-12 (Becker) incl. Intercom Select Panel (ICS mode selector) |
| 08-16053-B | L2341-293-01 | Intercom amplifier IC 3100-4 (Becker) |
| <i>Transponder</i> | | |
| 08-22026-B | L2325-092-12 | Transponder (Mode S) GTX330 (Garmin) |
| <i>DME</i> | | |
| 08-25022-B | L3455-090-02 | Distance measuring equipment DMS-44A (Chelton / Wulfsberg) |
| <i>VOR/ILS/MKR receivers</i> | | |
| 08-26010-B | L3431-092-02 | Marker beacon receiver/lights KR 21(Honeywell) |
| <i>Radio switch</i> | | |
| 08-29003-B | L2480-090-01 | Avionics/Radio master switches (Special Eurocopter) |
| <i>GPS/NAV/COM</i> | | |
| 08-43018-B | L3442-091-07 | GPS / NAV / COM GNS 430 (Garmin), copilot interfaced with FCDS (GPS stand-alone) |
| 08-43018-B | L3442-092-07 | GPS / NAV / COM GNS 430 (Garmin), pilot, interfaced with FCDS |
| <i>Conventional instruments</i> | | |
| 08-51013-B | L3425-806-51 | 2" std-by horizon AI 804 DC (Goodrich) with emergency battery |
| <i>Display system</i> | | |
| 08-65003-B | L3161-090-09 | MEGHAS - Flight Control Display System (FCDS) - Dual (4xSMD45) |
| <i>Miscellaneous</i> | | |
| 08-99000-A | L0000-150-11 | Avionics Solution 11 interconnection / wiring |

ON REQUEST:

- NVG friendly version of Avionics Solutions 8 and 11
- SMD68 on copilot side in place of dual SMD45



4.4.3 Minimum required equipment

| Minimum required equipment for Avionics Solution 8 and 11 | | | | | | PINAO | | | | | |
|--|----------------------|---|-------------------------------|-------|---|-------|-------|-------|-------|-------|-------|
| Document reference | Commercial reference | Title | Weight (margin $\pm 3\%$) | | | 21100 | 21110 | 21111 | 31100 | 31110 | 31111 |
| | | | kg | lb | | | | | | | |
| 05-03007-C | L2562-001-00 | First aid kit ¹⁴ | 1.3 | 2.9 | | | | X | | X | |
| 05-22008-C | L2621-001-00 | Engine fire extinguishing system | 3.6 | 7.9 | | X | X | | X | X | |
| 05-33001-B | L3113-001-00 | Slant panel | 0.8 | 1.8 | X | X | X | X | X | X | |
| 05-33002-B | L3113-004-00 | Center console | 2.2 | 4.9 | X | X | X | X | X | X | |
| 05-34002-B | L2576-001-00 | Avionics compartment | 4.2 | 9.3 | X | X | X | X | X | X | |
| 05-37016-C | L6701-001-00 | Copilot flight controls | 6.0 | 13.2 | X | X | X | X | X | X | |
| 05-38010-B | L3111-001-04 | 7" copilot instrument panel with glare shield | 2.7 | 6.0 | X | X | X | X | X | X | |
| 05-39006-B | L2514-003-01 | Map case in copilot door | 0.5 | 1.1 | X | X | X | X | X | X | |
| 05-39007-B | L3111-001-10 | Map case on instrument panel glare shield | 0.6 | 1.3 | X | X | X | X | X | X | |
| 05-39008-B | L3113-004-10 | Illuminated chart holder for pilot side | 0.9 | 2.0 | | | | X | | X | |
| 05-41004-C | L2104-100-00 | Bleed air heating system ¹⁵ | 6.6 | 14.6 | X | X | X | X | X | X | |
| 05-61010-B | L2433-006-00 | Battery, type (Saft) ULM, 40 Ah, 24 V instead of standard battery | 16.8 | 37.0 | | X | X | X | X | X | |
| 05-62010-C | L2420-005-00 | AC System (50VA) ¹⁶ | 1.9 | 4.2 | X | X | X | X | X | X | |
| 05-63003-B | L2432-001-00 | Starter/generators (2 x 200 A, 28 VDC), instead of standard generator | 3.6 | 7.9 | | X | X | X | X | X | |
|  06-12008-C | L3217-001-00 | Reinforced rear landing gear cross tube | 1.0 | 2.2 | X | X | X | X | X | X | |
| 06-45023-B | L3343-003-00 | Landing & search light, 450 W | 3.4 | 7.5 | X | X | X | X | X | X | |
|  06-67044-B | L2563-801-06 | ELT C406-N HM (Artex) incl. NAV. opt. | 3.3 | 7.3 | | | | X | | X | |
| 08-00005-A | L2300-008-04 | Avionics Solution 8 | 84.2 | 185.6 | | | | | | | |
| | | or | | | | | | X | X | X | X |
| 08-00022-A | L2300-011-02 | Avionics Solution 11 | 77.8 | 171.5 | | | | | | | |
| 08-21014-C | L3441-090-04 | Radar Altimeter KRA 405B (Honeywell) | 3.5 | 7.7 | | X | X | X | X | X | |
| 08-53002-B | L2212-400-00 | MEGHAS Sensor kit | 17.0 | 37.5 | X | X | X | X | X | X | |
| 08-54001-C | L3411-001-00 | Copilot pitot static system | 1.5 | 3.3 | X | X | X | X | X | X | |
| 08-71002-C | L2217-001-50 | IFR SAS (IFR pitch/roll Stability Augmentation System) | 12.2 | 26.9 | | X | X | X | | | |
| 08-72001-B | L2212-001-00 | Digital Automatic Flight Control System - DAFCS | 27.0 | 59.6 | | | | | X | X | X |

¹⁴ First aid kit complies with german regulation rules 1. DV LuftBO paragraph 5(2). Type of operation, procedures or regulations may require a different/specific first aid kit.

¹⁵ For helicopters dedicated for EMS select "Bleed air heating system: EMS version L2104-003-00" (05-41004-C) (7.0 kg / 15.4 lb.)

¹⁶ Alternatively the AC system L2420-002-00 (05-62010-C) (350VA; 3.2kg) can be selected

4.4.4 Possible add-ons

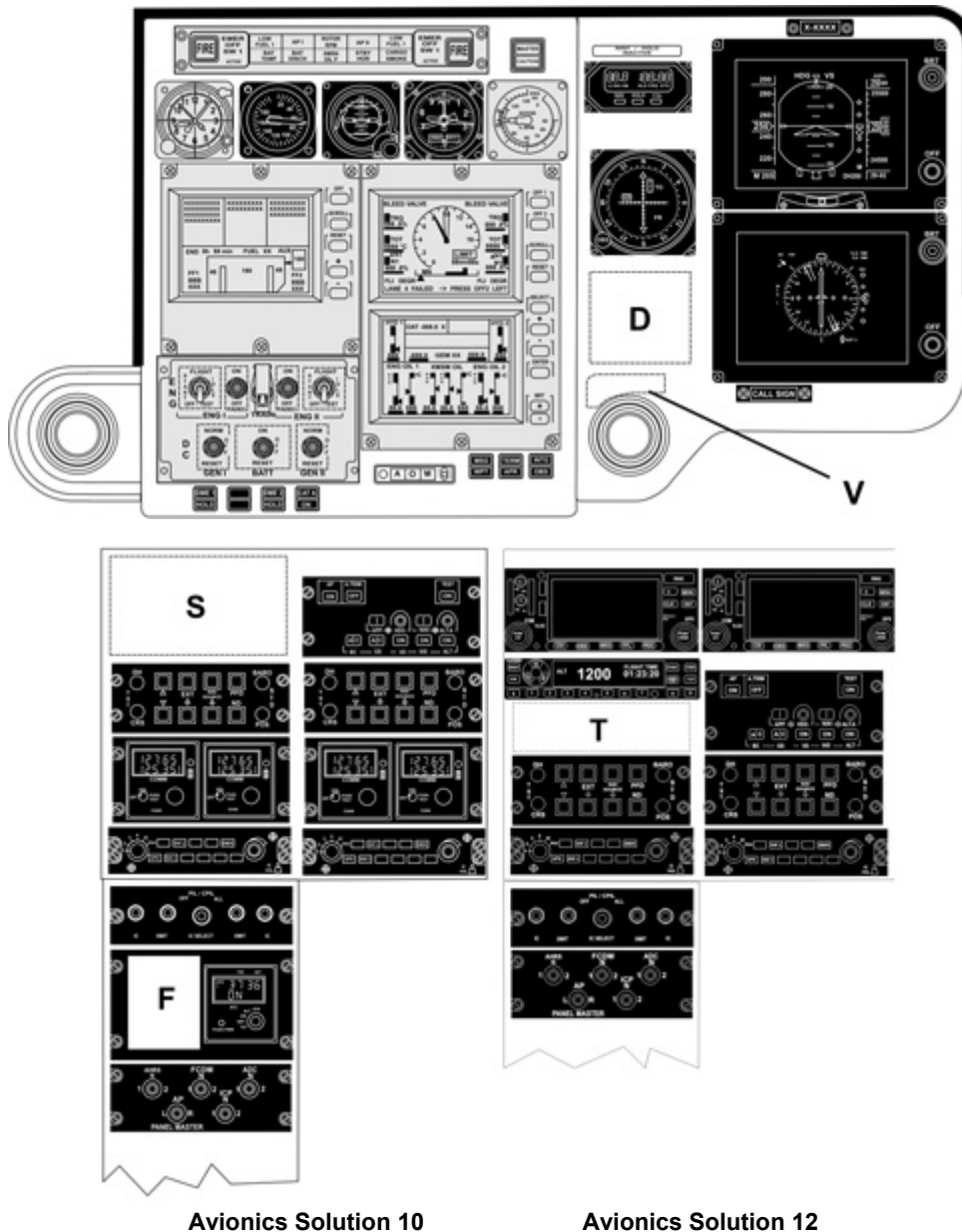
| Possible add-ons for Avionics Solution 8 and 11 | | | | | | PINA0 | | | | | |
|---|-------------------------|--|--------------------------|------|-------------------------------|-------|-------|-------|-------|-------|--|
| Document reference | Commercial reference | Title | Weight (margin ± 3 %) | | 21100 | 21110 | 21111 | 31100 | 31110 | 31111 | |
| | | | kg | lb | | | | | | | |
| 06-67044-B | L2563-801-06 | ELT C406-N HM (Artex) incl. NAV. opt. | 3.3 | 7.3 | X | X | | X | X | | |
| 08-21014-C | L3441-090-04 | Radar altimeter KRA 405B (Honeywell) | 3.5 | 7.7 | X | | | | | | |
| 08-24015-B | L3452-092-17 | ADF system DFS-43A (Chelton / Wulfsberg) | 9.6 | 21.2 | X | X | X | X | X | X | |
| | L3452-092-08 | ADF control unit CD-432B (Chelton / Wulfsberg) | 1.2 | 2.6 | | | | | | | |
| 08-31019-B | L3443-090-02 | Color weather radar RDR 2000 (Honeywell) | 6.6 | 14.6 | X | X | X | X | X | X | |
| | L2571-001-00 | Radar radome (for RDR2000) | 2.0 | 4.4 | | | | | | | |
| 08-31034-B | L3443-004-00 | Search and rescue weather radar RDR 1600 (Telephonics) | 16.7 | 36.8 | | X | X | X | X | X | |
| | L2571-002-00 | Radar radome (for RDR1600) | 4.8 | 10.6 | | | | | | | |
| 08-35007-B | L2327-001-01 | Traffic Advisory System TAS 9900BX (Ryan) | 9.5 | 20.9 | X | X | X | X | X | X | |
| 08-35007-B | L2327-001-11 | Traffic Advisory System TAS 9900 BX indicator 3" MHD (Ryan) ¹⁷ | 1.5 | 3.3 | X | X | X | X | X | X | |
| 08-35007-B | L3442-881-00 | Traffic Advisory System TAS 9900BX interface with GNS430 ¹⁸ | 0.2 | 0.4 | X | X | X | X | X | X | |
| 08-43017-B | L3442-092-12 | GPS Nav. system 2101 I/O Approach Plus (Free Flight) ⁸ | 3.2 | 7.1 | X | X | X | X | X | X | |
| 08-46020-B | L3168-092-04 | Digital moving Map EURONAV IV - RN6 (Euro Avionics), interfaced with FCDS, (basic version without maps) ¹⁹ | 5.4 | 11.9 | | X | X | X | X | X | |
| 08-65004-B | L3443-010-00 | Video Radar Unit (VRU) | 5.0 | 11.0 | X | X | X | X | X | X | |
| 08-72001-B | L2212-001-00 | Digital Automatic Flight Control System - DAFCS | 27.0 | 59.5 | instead of IFR pitch/roll SAS | | | | | | |
| 08-81018-C | L2321-007-00 | M'ARMS® Cockpit Voice and Flight Data Recorder (CVFDR), ground station not included (in combination with UMS: 18.3 kg / 40.3 lb) | 15.7 | 34.6 | DAFCS required | | | X | X | X | |
| 08-83007-C | L3171-001-00 | M'ARMS® Usage Monitoring System (UMS), ground station not included | 7.2 | 15.9 | DAFCS required | | | X | X | X | |

4.4.5 Further avionics add-ons see chapter 6 page 50

¹⁷ Only possible for Avionics Solutions 8
¹⁸ Only possible for Avionics Solutions 11
¹⁹ Approval for VFR only

4.5 Single Pilot IFR Glass Cockpit, based on Avionics Solution 10 or 12

4.5.1 Instrument panel overview



Additional space:

- D - e.g. for radar altimeter indicator, stormscope
- F - e.g. for ADF control unit (CD-432B)
- S / T - e.g. for GPS or other optional equipment
- V - for Video Radar Unit

4.5.2 Contents of Avionics Solution 10 and 12




| <i>Document reference</i> | <i>Commercial reference</i> | <i>Title</i> |
|---------------------------------|-----------------------------|---|
| 08-00007-A | L2300-010-04 | Avionics Solution 10, consisting of: |
| <i>Radio Com</i> | | |
| 08-11022-B | L2313-091-08 | VHF AM/COM system, copilot VCS-40 A (Chelton / Wulfsberg) |
| 08-11022-B | L2313-091-13 | Control unit CD 402 B, copilot (Chelton / Wulfsberg) |
| 08-11022-B | L2313-092-07 | VHF AM/COM system, pilot VCS-40 A (Chelton / Wulfsberg) |
| 08-11022-B | L2313-092-13 | Control unit CD 402 B, pilot (Chelton / Wulfsberg) |
| <i>Intercom System</i> | | |
| 08-16053-B | L2341-191-01 | Audio/Comm. control system AS 3100-12 (Becker), copilot |
| 08-16053-B | L2341-192-01 | Audio/Comm. control system (pilot) AS 3100-12 (Becker) incl. Intercom Select Panel (ICS mode selector) |
| 08-16053-B | L2341-293-01 | Intercom amplifier IC 3100-4 (Becker) |
| <i>Transponder</i> | | |
| 08-22027-B | L2325-092-06 | Transponder (Mode S) MST 67A (Honeywell) |
| 08-22027-B | L2325-092-15 | Transponder control unit PS 578A (Honeywell) |
| <i>DME</i> | | |
| 08-25022-B | L3455-090-02 | Distance measuring equipment DMS-44A (Chelton / Wulfsberg) |
| <i>VOR/ILS/MKR receivers</i> | | |
| 08-26024-B | L3432-091-06 | VOR/ILS/MKR Navigation system, copilot VNS-41 A (Chelton / Wulfsberg) |
| 08-26024-B | L3432-091-09 | Control unit CD 412 B, copilot (Chelton / Wulfsberg) |
| 08-26024-B | L3432-092-07 | VOR/ILS/MKR Navigation system, pilot VNS-41 A (Chelton / Wulfsberg) |
| 08-26024-B | L3432-092-12 | Control unit CD 412 B, pilot (Chelton / Wulfsberg) |
| <i>Radio switch</i> | | |
| 08-29003-B | L2480-090-01 | Avionics/Radio master switches (Special Eurocopter) |
| <i>Conventional instruments</i> | | |
| 08-51013-A | L3425-806-51 | 2" std-by horizon AI 804 DC (Goodrich) with emergency battery |
| <i>Display system</i> | | |
| 08-61011-A | L0000-200-12 | Back-up indicator CDI KI 204 (Honeywell), Back-up indicator SD 442 B (Chelton / Wulfsberg) |
| 08-65003-B | L3161-092-03 | MEGHAS - Flight Control Display System (FCDS) - Single (2xSMD45) |
| <i>Miscellaneous</i> | | |
| 08-99000-A | L0000-150-10 | Avionics Solution 10 interconnection / wiring |

| <i>Document reference</i> | <i>Commercial reference</i> | <i>Title</i> |
|---------------------------------|-----------------------------|---|
| 08-00023-A | L2300-012-02 | Avionics Solution 12, consisting of: |
| <i>Intercom System</i> | | |
| 08-16053-B | L2341-191-01 | Audio/Comm. control system AS 3100-12 (Becker), copilot |
| 08-16053-B | L2341-192-01 | Audio/Comm. control system (pilot) AS 3100-12 (Becker) incl. Intercom Select Panel (ICS mode selector) |
| 08-16053-B | L2341-293-01 | Intercom amplifier IC 3100-4 (Becker) |
| <i>Transponder</i> | | |
| 08-22026-B | L2325-092-12 | Transponder (Mode S) GTX330 (Garmin) |
| <i>DME</i> | | |
| 08-25022-B | L3455-090-02 | Distance measuring equipment DMS-44A (Chelton / Wulfsberg) |
| <i>VOR/ILS/MKR receivers</i> | | |
| 08-26010-B | L3431-092-02 | Marker beacon receiver/lights KR 21(Honeywell) |
| <i>Radio switch</i> | | |
| 08-29003-B | L2480-090-01 | Avionics/Radio master switches (Special Eurocopter) |
| <i>GPS/NAV/COM</i> | | |
| 08-43018-B | L3442-091-07 | GPS / NAV / COM GNS 430 (Garmin), copilot interfaced with FCDS (GPS stand-alone) |
| 08-43018-B | L3442-092-07 | GPS / NAV / COM GNS 430 (Garmin), pilot interfaced with FCDS |
| <i>Conventional instruments</i> | | |
| 08-51013-B | L3425-806-51 | 2" std-by horizon AI 804 DC (Goodrich) with emergency battery |
| <i>Display system</i> | | |
| 08-61011-A | L0000-200-12 | Back-up indicator CDI KI 204 (Honeywell), Back-up indicator SD 442 B (Chelton / Wulfsberg) |
| 08-65003-B | L3161-092-03 | MEGHAS - Flight Control Display System (FCDS) - Single (2xSMD45) |
| <i>Miscellaneous</i> | | |
| 08-99000-A | L0000-150-12 | Avionics Solution 12 interconnection / wiring |

ON REQUEST:

- NVG friendly version of Avionics Solutions 10 and 12

4.5.3 Minimum required equipment

| Minimum required equipment for Avionics Solution 10 and 12 | | | | | PINA0 | | |
|--|----------------------|--|--|-------|-------|-------|-------|
| Document reference | Commercial reference | Title | Weight (margin $\pm 3\%$) kg lb | | 11100 | 11110 | 11111 |
| 05-03007-C | L2562-001-00 | First aid kit ²⁰ | 1.3 | 2.9 | | | X |
| 05-22008-C | L2621-001-00 | Engine fire extinguishing system | 3.6 | 7.9 | X | X | |
| 05-33001-B | L3113-001-00 | Slant panel | 0.8 | 1.8 | X | X | X |
| 05-33002-B | L3113-004-00 | Center console | 2.2 | 4.9 | X | X | X |
| 05-34002-B | L2576-001-00 | Avionics compartment | 4.2 | 9.3 | X | X | X |
| 05-39007-B | L3111-001-10 | Map case on instrument panel glare shield | 0.6 | 1.3 | X | X | X |
| 05-39008-B | L3113-004-10 | Illuminated chart holder for pilot side | 0.9 | 2.0 | | | X |
| 05-41004-C | L2104-100-00 | Bleed air heating system ²¹ | 6.6 | 14.6 | X | X | X |
| 05-44002-B | L2122-001-00 | Ventilation extruder w/o copilot I-panel extension | 0.3 | 0.7 | X | X | X |
| 05-61010-B | L2433-006-00 | Battery, type "Soft", ULM, 40 Ah, 24 V instead of standard battery | 16.8 | 37.0 | X | X | X |
| 05-62010-C | L2420-005-00 | AC System (50VA) ²² | 1.9 | 4.2 | X | X | X |
| 05-63003-B | L2432-001-00 | Starter/generators (2 x 200 A, 28 VDC), instead of standard generators | 3.6 | 7.9 | X | X | X |
|  06-12008-C | L3217-001-00 | Reinforced rear landing gear cross tube | 1.0 | 2.2 | X | X | X |
| 06-45023-B | L3343-003-00 | Landing & search light, 450 W | 3.4 | 7.5 | X | X | X |
|  06-67044-B | L2563-801-06 | ELT C406-N HM (Artex) incl. NAV. Opt | 3.3 | 7.3 | | | X |
|  08-00022-A | L2300-010-04 | Avionics Solution 10 | 81.6 | 180.0 | | | |
| | | or | | | | | |
| 08-00023-A | L2300-012-02 | Avionics Solution 12 | 75.2 | 165.8 | X | X | X |
| 08-21014-C | L3441-090-04 | Radar altimeter KRA 405B (Honeywell) | 3.5 | 7.7 | X | X | X |
| 08-53002-B | L2212-400-00 | MEGHASSensor kit | 17.0 | 37.5 | X | X | X |
| 08-54001-C | L3411-001-00 | Copilot pitot static system | 1.5 | 3.3 | X | X | X |
| 08-72001-B | L2212-001-00 | Digital Automatic Flight Control System - DAFCS | 27.0 | 59.6 | X | X | X |

²⁰ First aid kit complies with german regulation rules 1. DV LuftBO paragraph 5(2). Type of operation, procedures or regulations may require a different/specific first aid kit.

²¹ For helicopters dedicated for EMS select "Bleed air heating system: EMS version L2104-003-00" (05-41004-C) (7.0 kg / 15.4 lb.)

²² Alternatively the AC system L2420-002-00 (05-62010-C) (350VA; 3.2kg) can be selected

4.5.4 Possible add-ons

| Possible add-ons for Avionics Solutions 10 and 12 | | | | | | PINA0 | | |
|---|----------------------|--|-------------------------------|------|---|-------|------|------|
| Document reference | Commercial reference | Title | Weight (margin $\pm 3\%$) | | | 1110 | 1110 | 1111 |
| | | | kg | lb | | | | |
| 06-67044-B | L2563-801-06 | ELT C406-N HM (Artex) incl. NAV. opt. | 3.3 | 7.3 | X | X | | |
| 08-24015-B | L3452-092-17 | ADF system DFS-43A (Chelton / Wulfsberg) | 9.6 | 21.2 | X | X | X | |
| | L3452-092-08 | ADF control unit CD-432B (Chelton / Wulfsberg) | 1.2 | 2.6 | | | | |
| ⚠ 08-31019-B | L3443-090-02 | Color weather radar RDR 2000 (Honeywell) | 6.6 | 14.6 | X | X | X | |
| | L2571-001-00 | Radar radome for RDR2000 | 2.0 | 4.4 | | | | |
| ⚠ 08-31034-B | L3443-004-00 | Search and rescue weather radar RDR 1600 (Telephonics) | 16.7 | 36.8 | | X | X | X |
| | L2571-002-00 | Radar radome for RDR 1600 | 4.8 | 10.6 | | | | |
| ⚠ 08-35007-B | L2327-001-01 | Traffic Advisory System TAS 9900BX (Ryan) | 9.5 | 20.9 | X | X | X | |
| ⚠ 08-35007-B | L2327-001-11 | Traffic Advisory System TAS 9900 BX indicator 3" MHD (Ryan) ²³ | 1.5 | 3.3 | X | X | X | |
| ⚠ 08-35007-B | L3442-881-00 | Traffic Advisory System TAS 9900BX interface with GNS430 ²⁴ | 0.2 | 0.4 | X | X | X | |
| 08-43017-B | L3442-092-12 | GPS Nav. system 2101 I/O Approach Plus (Free Flight) ²³ | 3.2 | 7.1 | X | X | X | |
| ⚠ 08-46020-B | L3168-092-04 | Digital moving Map EURONAV IV - RN6 (Euro Avionics), interfaced with FCDS, (basic version without maps) ²⁵ | 5.4 | 11.9 | X | X | X | |
| 08-65004-B | L3443-010-00 | Video Radar Unit (VRU) | 5.0 | 11.0 | X | X | X | |
| 08-81018-C | L2321-007-00 | M'ARMS [®] Cockpit Voice and Flight Data Recorder (CVFDR), ground station not included (in combination with UMS: 18.3 kg / 40.3 lb) | 15.7 | 34.6 | X | X | X | |
| 08-83007-C | L3171-001-00 | M'ARMS M'ARMS [®] Usage Monitoring System (UMS), ground station not included | 7.2 | 15.9 | X | X | X | |

4.5.5 Further avionics add-ons see chapter 6 page 50

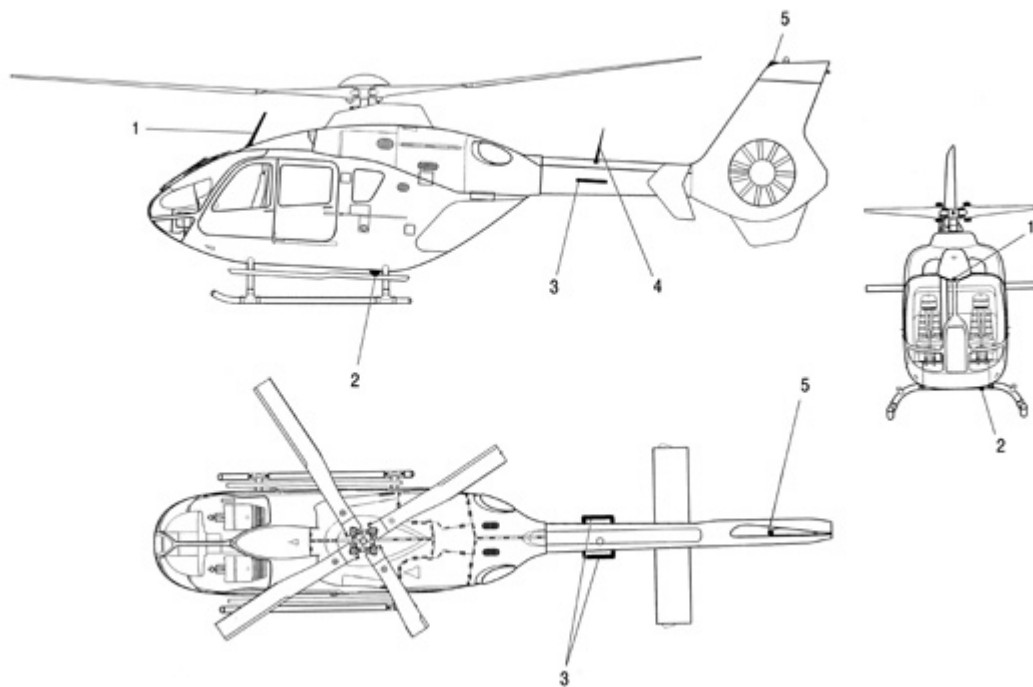
²³ Only possible with Solution 10

²⁴ Only possible with Solution 12

²⁵ Approval for VFR only

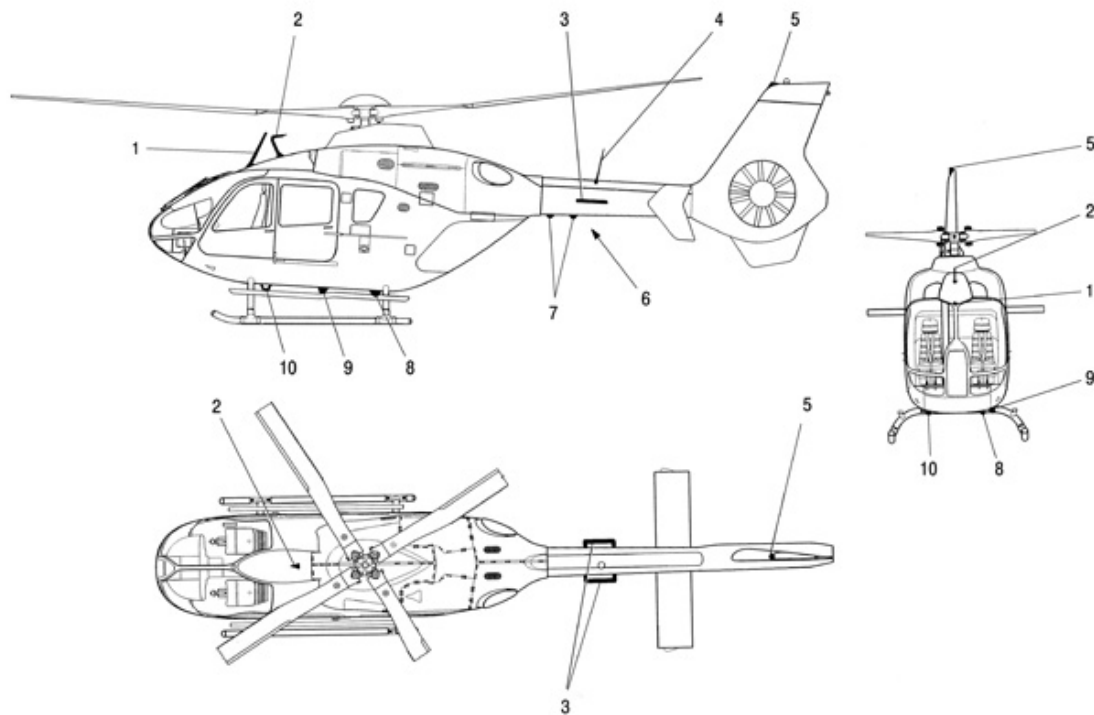
4.6 Antenna layouts

4.6.1 Typical VFR antenna layout



- 1 – ELT antenna
- 2 – ATC antenna
- 3 – VOR antennas
- 4 – VHF 1 antenna
- 5 – GPS antenna

4.6.2 Typical IFR antenna layout



- 1 – ELT antenna
- 2 – VHF2 antenna
- 3 – VOR antennas
- 4 – VHF1 antenna
- 5 – GPS antenna

- (6 – ADF antenna (if required))
- 7 – Radar altimeter antenna
- 8 – ATC antenna
- 9 – Marker antenna
- 10 – DME antenna

5 Cabin arrangement

5.1 Passenger transport

5.1.1 Five (5) Passenger configuration



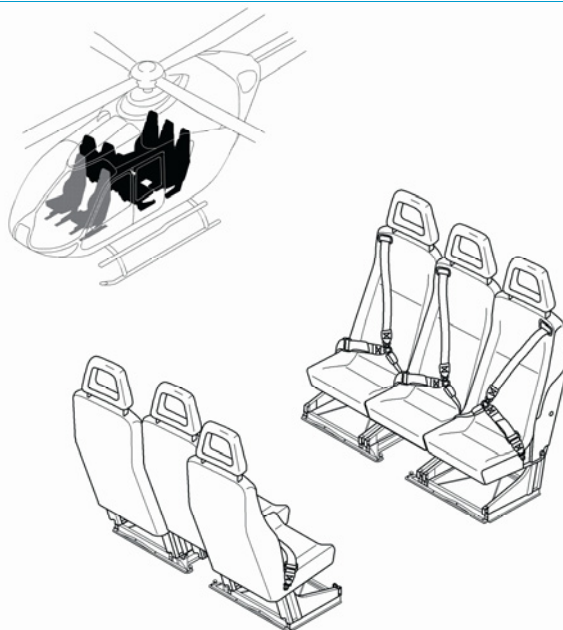
| Document reference | Commercial reference | Title | Weight (margin $\pm 3\%$) | |
|-----------------------|-------------------------|--|-------------------------------|-------|
| | | | kg | lb |
| 07-80024-A | L2521-001-00 | 5 seat passenger/ transport configuration consisting of: | 65.4 | 144.2 |
| | | 3 forward passenger seats, facing backwards | | |
| | | 2 rear passenger seats, facing forwards | | |
| | | 1 Covers for sliding door fairing LH/RH | | |
| | | 1 Map case in sliding doors LH/RH | | |
| | | 1 Variable tie-down net | | |
| | | 2 Retractable coat hooks in rear cabin | | |

5.1.2 Six (6) Passenger configuration (high density seats) ⚠

This installation is characterized by:

Weight
(margin $\pm 3\%$)

| Document reference | Commercial reference | Title | kg | lb |
|-----------------------------------|----------------------|---|------|------|
| 07-27001-B | L2522-001-00 | 3 forward passenger seats, facing backwards | 37.4 | 82.5 |
| In combination with either | | | | |
| 07-27004-B | L2522-004-10 | Utility seats for 3 rear passengers, model Eurocopter, fixed provisions | 1.2 | 2.6 |
| 07-27004-B | L2522-004-20 | Utility seats for 3 rear passengers, model Eurocopter, detachable parts | 33.2 | 73.2 |
| or | | | | |
| 07-27005-B | AEL-30130-007-1 | Utility seats for 3 rear passengers, model Aerolite, detachable parts | 37.0 | 81.6 |
| 07-27005-B | AEL-30130-007-2 | Utility seats for 3 rear passengers, model Aerolite, fixed provisions | 0.0 | 0.0 |



Model Eurocopter




Model Aerolite

5.1.3 Corporate / VIP passenger transport

5.1.3.1 Five (5) corporate passenger transport



| Document reference | Commercial reference | Title | Weight (margin $\pm 3\%$) | |
|--|-------------------------|---|-------------------------------|-------|
| | | | kg | lb |
|  | 07-80022-A L2521-002-00 | 5 seat Corporate/VIP configuration consisting of: | 115.7 | 255.0 |
| | | 1 VIP-pilot seat (instead of std. pilot seat) | | |
| | | 1 VIP-copilot seat (instead of std. copilot seat) | | |
| | | 4 VIP passenger seats (2 front and 2 rear) | | |
| | | 1 VIP passenger seat (front, middle) | | |
| | | 1 Rear cabinet with armrest | | |
| | | 1 VIP carpet for cockpit, cabin and cargo compartment ²⁶ | | |
| | | 1 Armrest in rear window niche LH / RH | | |
| | | 1 Special painted interior | | |
| | | 1 Covers for sliding door fairing LH/RH | | |
| | | 1 Map case in sliding doors LH/RH | | |
| | | 1 Variable tie-down net | | |
| | | 2 Retractable coat hooks in rear cabin | | |
| | | 1 Cabin / cargo compartment separation wall | | |
| | | 1 Enhanced sound proofing kit | | |

²⁶ If additional internal long range fuel tank is selected, a VIP carpet will only be installed in the cockpit and passenger cabin

5.1.3.2 Four (4) VIP passenger transport



Weight
(margin $\pm 3\%$)

| <i>Document reference</i> | <i>Commercial reference</i> | <i>Title</i> | <i>kg</i> | <i>lb</i> |
|-------------------------------|---------------------------------|---|-----------|-----------|
| 07-80023-A | L2521-003-00 | 4 seat Corporate/VIP configuration consisting of: | 115.9 | 255.5 |
| | | 1 VIP-pilot seat (instead of std. pilot seat) | | |
| | | 1 VIP-copilot seat (instead of std. copilot seat) | | |
| | | 4 VIP passenger seats (2 front and 2 rear) | | |
| | | 1 Front cabinet, flat | | |
| | | 1 Rear cabinet with armrest | | |
| | | 1 VIP carpet for cockpit, cabin and cargo compartment ²⁷ | | |
| | | 1 Armrest in rear window niche LH / RH | | |
| | | 1 Special painted interior | | |
| | | 1 Covers for sliding door fairing LH/RH | | |
| | | 1 Map case in sliding doors LH/RH | | |
| | | 1 Variable tie-down net | | |
| | | 2 Retractable coat hooks in rear cabin | | |
| | | 1 Cabin / cargo compartment separation wall | | |
| | | 1 Enhanced sound proofing kit | | |



²⁷ If additional internal long range fuel tank is selected, a VIP carpet will only be installed in the cockpit and passenger cabin

5.1.3.3 Customer specific Corporate / VIP solutions

Customer specific solutions can be defined by using the following table.

■ Note: Only one item per line can be selected.

| | | | |
|-----------------|---|--|--|
| COCKPIT | VIP pilot seat 07-81015-A L2525-101-11 (1.1 kg / 2.4 lb.) | VIP pilot seat height adjustable 07-81015-A L2525-101-71 (4.8 kg / 10.6 lb.) | |
| | VIP copilot seat 07-81015-A L2525-101-21 (1.1 kg / 2.4 lb.) | VIP copilot seat height adjustable 07-81015-A L2525-101-76 (4.8 kg / 10.6 lb.) | |
| PASSENGER CABIN | 4 VIP passenger seats (2 front and 2 rear) 07-81013-A L2525-102-00 (56.1 kg / 123.7 lb.) | | |
| | 1 VIP passenger seat (front, middle) 07-81013-A L2525-202-20 (14.2 kg / 31.3 lb.) | Front cabinet, flat 07-85002-A L2526-212-601 (14.5 kg / 32.0 lb.) | Front cabinet, middle 07-85002-A L2526-112-63 (22.5 kg / 49.6 lb.) |
| | Table for front cabinet 07-85002-A L2526-112-61 (3.0 kg / 6.6 lb.) | | |
| | Cooling box for front cabinet 07-85002-A L2526-112-62 (2.5 kg / 5.5 lb.) | | |
| | Rear cabinet with Armrest (Wood) 07-85003-A L2526-112-710 (10.0 kg / 22.2 lb.) | Rear cabinet, flat 07-85004-A L2526-312-701 (8.6 kg / 19.0 lb.) | Rear cabinet, high 07-85004-A L2526-112-75 (14.0 kg / 30.9 lb.) |
| | Armrests in rear window niche, LH/RH 07-82010-A L2525-102-62 (0.8 kg / 1.8 lb.) | | |











| | |
|--------------------------------|---|
| GENERAL | Special painted interior 07-86001-B L2525-100-35 (0.1 kg / 0.2 lb.) |
| | VIP carpet for cockpit, cabin and cargo compartment 07-83003-A L2525-104-00 (16.6 kg / 36.6 lb.) |
| | Retractable coat hooks (2ea) in rear cabin 07-90006-B L2514-011-00 (0.1 kg / 0.2 lb.) |
| | Map case in sliding doors LH/RH 07-60014-B L2514-013-00 (1.4 kg / 3.0 lb.) |
| | Variable tie-down net 07-60015-B L2514-014-00 (3.3 kg / 7.3 lb.) |
| | Covers for sliding door fairing LH/RH 07-50036-B L2514-012-00 (0.1 kg / 0.2 lb.) |
| | Control covers painted in harmony with carpet 07-90008-A L2525-101-65 (-2.3 kg / -5.1 lb.) |
| GROUND SUPPORT EQUIPMENT | Fabric protection cover for 1 VIP pilot or copilot seat 07-90007-A L2525-111-50 (GSE) |
| | Fabric protection cover for 1 VIP pax seat 07-90007-A L2525-112-91 (GSE) |
| | Plastic protection cover for cockpit carpet 07-90007-A L2525-111-60 (GSE) |
| | Plastic protection cover for cabin carpet 07-90007-A L2525-112-92 (GSE) |
| HIGHLY RECOMMENDED ITEMS | Air conditioning system for tropical environment 05-42020-B L2105-001-10 (57.9 kg / 127.6 lb.) + |
| | Special ducting system and additional air outlets for the front VIP - passenger-seats 05-42020-B L2525-112-46 (0.5 kg / 1.1 lb.) |

6 Optional equipment

6.1 Further available equipment









General Equipment

Weight
(margin $\pm 3\%$)





| Document reference | Commercial reference | Title | kg | lb |
|--|----------------------|---|-------------|-------------|
|  05-02016-C | L1111-002-00 | Two-color exterior painting instead of single color painting | 1.5 | 3.3 |
|  05-02016-C | L1111-004-00 | Multicolor exterior painting instead of single color painting | 2.0 | 4.4 |
|  05-03007-C | L2562-001-00 | First aid kit ²⁸ | 1.5 | 3.3 |
|  05-03008-B | L2562-001-10 | First aid kit for DGAC certification | 2.8 | 6.2 |
| 05-12001-B | L5232-001-00 | Multifunction handle on the main gear box cowling (LH and RH) | 0.6 | 1.3 |
| 05-12002-B | L2551-003-00 | Additional 4 tie-down fittings for airline attachment rails | 0.6 | 1.3 |
| 05-21015-B | L8541-001-10 | Wire strike protection system, fixed provisions | 3.3 | 7.3 |
|  05-21015-B | L8541-001-20 | Wire strike protection system, detachable parts | 8.2 | 18.1 |
| 05-22007-B | L7924-001-00 | Fuzz burners for engines | 1.2 | 2.6 |
| 05-22008-C | L2621-001-00 | Engine fire extinguishing system | 3.6 | 7.9 |
| 05-22014-A | L5371-001-00 | Engine outlet heat protection | 1.2 | 2.7 |
| 05-23006-B | L7165-002-00 | Engine compressor wash kit | 3.2 | 7.1 |
| 05-24017-B | L6211-014-00 | Sand erosion protection kit for rotor blades | 0.9 | 2.0 |
|  05-25016-D | L7161-001-10 | Sand filter system, fixed provisions | 11.3 | 24.9 |
| 05-25016-D | L7161-001-20 | Sand filter system, detachable parts | 26.2 | 57.8 |
| 05-26012-B | L1241-001-00 | Anti-corrosion protection | 1.1 | 2.4 |
|  05-31025-B | L5211-002-00 | Sliding window in sliding doors | 1.9 | 4.2 |
| 05-31026-C | L2514-002-00 | Tinted sun shades for cockpit windshield roof section | 1.9 | 4.2 |
| 05-31026-C | L5621-001-00 | Tinted window for cockpit doors | 0.0 | 0.0 |
|  05-31026-C | L5632-001-00 | Tinted windows for passenger cabin, incl. sliding windows for sliding doors | 0.6 | 1.3 |
| 05-31027-B | L5633-001-10 | Window in clam-shell door, LH | 0.8 | 1.8 |
|  05-31027-B | L5633-001-20 | Window in clam-shell door, RH | 0.8 | 1.8 |
| 05-31028-C | L2524-030-10 | IFR – training screen, fixed provisions | 0.1 | 0.2 |
| 05-31028-C | L2524-030-20 | IFR – training screen, detachable parts | 1.6 | 3.5 |
| 05-31045-A | L5211-001-11 | Lockable sliding window in copilots' door | 0.1 | 0.2 |
| 05-31045-A | L5211-001-12 | Lockable sliding window in pilots' door | 0.1 | 0.2 |
| 05-32007-B | L3042-001-00 | Windshield wiper system | 5.0 | 11.0 |
|  05-34003-B | L2576-002-00 | Dampers for avionics compartment | 1.8 | 4.0 |

²⁸ First aid kit complies with german regulation rules 1. DV LuftBO paragraph 5(2). Type of operation, procedures or regulations may require a different/specific first aid kit.

General Equipment (contd.)
Weight
(margin $\pm 3\%$)
















| Document reference | Commercial reference | Title | kg | lb |
|--|----------------------|--|------|-------|
| 05-37016-C | L6701-001-00 | Copilot flight controls | 6.2 | 13.7 |
|  05-37017-B | L6721-001-00 | Covers for copilot flight controls ²⁹ | -2.5 | -5.5 |
| 05-39006-B | L2514-003-01 | Map case in copilot door | 0.5 | 1.1 |
| 05-39007-B | L3111-001-10 | Map cases on instrument panel glare shield | 0.6 | 1.3 |
| 05-39008-B | L3113-004-10 | Illuminated chart holder for pilot side | 1.0 | 2.2 |
|  05-39008-B | L3113-004-20 | Illuminated chart holder for copilot side | 0.9 | 2.0 |
|  05-42019-B | L2105-001-00 | Air conditioning system | 54.3 | 119.7 |
|  05-42020-B | L2105-001-10 | Air conditioning system for tropical environment | 57.9 | 127.6 |
|  05-61010-B | L2433-006-00 | Battery, type "Saft", ULM, 40 Ah, 24 V instead of standard battery | 16.8 | 37.0 |
| 05-71001-C | L6351-002-00 | Rotor brake system | 5.3 | 11.7 |
|  05-81032-C | L2818-100-10 | Internal long range fuel tank system, fixed provisions | 2.6 | 5.7 |
|  05-81032-C | L2818-100-20 | Internal long range fuel tank system, detachable parts | 36.1 | 79.6 |
| 05-81033-B | L2812-001-00 | Self sealing fuel supply tanks | 4.5 | 9.9 |
| 05-85008-C | L2843-001-00 | Fuel management system (Fuel flow meters) | 1.2 | 2.6 |
| 05-92009-B | L6611-001-10 | Main rotor blade folding: basic kit | 1.3 | 2.9 |
|  05-92009-B | L6611-001-20 | Main rotor blade folding: fixed provisions for ground handling kit | 0.8 | 1.8 |
| 05-92009-B | L6611-001-30 | Main rotor blade folding: ground handling kit | GSE | GSE |
| 05-93007-B | L8544-002-00 | Lashing points (wind speeds up to 100 kts) (weight GSE: 24.9 kg) | 0.7 | 1.5 |
| 05-93008-B | L8544-001-00 | Lashing points (wind speeds up to 40 kts) | 2.4 | 5.3 |
| 05-95001-C | L1321-001-00 | Cover kit for helicopter | GSE | GSE |
| 05-97001-C | L6201-001-30 | Accelerometers for Track and Balance system | 0.0 | 0.0 |
| 05-97002-C | L6201-002-10 | Optical tracker (Chadwick Helmuth), fixed provisions | 0.1 | 0.2 |
| 05-97002-C | L6201-002-20 | Optical tracker (Chadwick Helmuth), detachable parts | 0.7 | 1.5 |

Specific Mission Equipment
Weight
(margin $\pm 3\%$)

| Document reference | Commercial reference | Title | kg | lb |
|--|----------------------|--|------|------|
| 06-11021-B | L3274-001-10 | Settling protectors, fixed provisions | 1.9 | 4.2 |
|  06-11021-B | L3274-001-20 | Settling protectors, detachable parts | 7.9 | 17.4 |
| 06-11022-C | L3272-001-10 | Snow skids, fixed provisions | 0.9 | 2.0 |
|  06-11022-C | L3272-001-20 | Snow skids, detachable parts | 21.9 | 48.3 |
|  06-12007-B | L3273-001-00 | Lengthened skids | 8.3 | 18.3 |
|  06-12009-B | L3216-001-10 | High landing gear (instead of standard landing gear) | 26.0 | 57.3 |















²⁹ Stick, Pitch and Pedals have to be removed - thus negative delta weight

Specific Mission Equipment (contd.)
Weight
(margin $\pm 3\%$)

| Document reference | Commercial reference | Title | kg | lb |
|--|--------------------------------|--|-------------|--------------|
|  06-21017-C | L8512-001-10 | External hoist LH, fixed provisions ³⁰ | 8.6 | 19.0 |
|  06-21017-C | L8512-001-12 | External hoist RH, fixed provisions ³⁰ | 8.6 | 19.0 |
|  06-21017-C | L8512-001-20 | External hoist 50m, detachable parts ³⁰ , (incl. 1 week winch operator training) | 59.2 | 130.5 |
|  06-21017-C | L8512-001-21 | External hoist 90m, detachable parts ³⁰ , (incl. 1 week winch operator training) | 62.7 | 138.2 |
|  06-26011-C | L8511-002-10 | Cargo hook mirrors RH, fixed provisions | 0.8 | 1.8 |
|  06-26011-C | L8511-002-20 | Cargo hook mirrors RH, detachable parts | 3.9 | 8.6 |
|  06-26011-C | L8511-006-10 | Cargo hook mirrors LH & RH, fixed provisions | 1.0 | 2.2 |
|  06-26011-C | L8511-006-20 | Cargo hook mirrors LH & RH, detachable parts | 7.7 | 17.0 |
|  06-27019-D | L8511-001-10 | Cargo hook system, fixed provisions | 3.7 | 8.2 |
| | 06-27019-D L8511-001-30 | Cargo hook system, detachable parts | 16.5 | 36.4 |
|  06-27022-B | L8511-005-10 | Dual cargo hook system, fixed provisions | 4.3 | 9.5 |
|  06-27022-B | L8511-005-20 | Dual cargo hook system, detachable parts | 21.5 | 47.4 |
| | 06-45023-B L3343-003-00 | Landing & search light, 450 W | 3.1 | 6.8 |
| | 06-46001-B L3344-001-00 | Strobe lights, white | 1.4 | 3.1 |
|  06-61015-B | L3215-001-10 | Emergency floats, fixed provisions | 7.8 | 17.2 |
|  06-61015-B | L3215-001-21 | Emergency floats, detachable parts | 64.6 | 142.4 |
| | 06-65002-B L2566-001-00 | Emergency hammer | 0.2 | 0.4 |
| | 06-65004-B L2625-003-00 | 2nd portable fire extinguisher | 2.8 | 6.2 |
| | 06-66008-C L3353-005-00 | Emergency lights | 2.9 | 6.4 |
| | 06-66009-B L3322-001-00 | Boarding step illumination | 0.2 | 0.4 |
|  06-66010-B | L3353-006-20 | Illuminated signs "NO SMOKING/FASTEN SEAT BELT" | 0.8 | 1.8 |
| | 06-69005-B L2341-006-61 | Voice alert generator 611-014 (NAT) | 0.4 | 0.9 |
|  06-81009-B | L8503-001-10 | Fire extinguishing bucket attachment (Bambi Bucket), fixed provisions | 0.8 | 1.8 |


³⁰ Communication via copilot audio / comm. control unit

Interior Layout
Weight
(margin $\pm 3\%$)

| Document reference | Commercial reference | Title | kg | lb |
|--|----------------------|---|------|------|
| 07-15016-B | L2512-003-10 | Height adjustable pilot seat instead of standard pilot seat | 1.8 | 4.0 |
| 07-15016-B | L2512-003-20 | Height adjustable copilot seat instead of standard copilot seat | 1.8 | 4.0 |
| 07-30012-B | L2581-001-00 | Enhanced sound proofing kit | 4.8 | 10.6 |
|  07-30013-B | L2524-002-00 | Separation curtain for cockpit / cabin | 2.1 | 4.6 |
|  07-30014-B | L2524-021-00 | Separation curtain for cabin / cargo compartment | 2.5 | 5.5 |
|  07-30015-B | L2524-001-00 | Separation wall for cabin / cargo compartment | 4.4 | 9.7 |
| 07-30018-A | L5213-003-00 | Curtains for cabin windows (grey) | 1.6 | 3.5 |
|  07-40005-B | L2513-200-00 | Washable floor covering for cockpit | 3.9 | 8.6 |
|  07-40005-B | L2513-210-00 | Washable floor covering for cargo compartment | 3.0 | 6.4 |
|  07-40005-B | L2513-220-00 | Washable floor covering for cockpit, cabin and cargo compartment | 11.5 | 25.3 |
|  07-40006-A | L2513-300-00 | Carpet for cockpit and cabin | 6.6 | 14.6 |
|  07-40006-A | L2513-310-00 | Carpet for cockpit, cabin and cargo compartment | 8.9 | 19.5 |
| 07-40017-A | L8552-001-00 | Sealed cabin floor | 0.3 | 0.7 |
|  07-50025-C | L5211-004-00 | Securing device for complete opening of cockpit doors | 0.8 | 1.8 |
| 07-50026-B | L5231-001-00 | One-hand latching system for clam-shell doors | 1.0 | 2.2 |
| 07-50027-B | L5231-002-00 | Extended opening fasteners for clam-shell doors | 0.3 | 0.9 |
|  07-50028-A | L5213-001-11 | Sliding door fastener, intermediate and max. position, LH | 1.0 | 2.2 |
|  07-50028-A | L5213-001-12 | Sliding door fastener, max. position, LH | 0.4 | 0.9 |
|  07-50028-A | L5213-001-21 | Sliding door fastener, intermediate and max. position, RH | 1.0 | 2.2 |
|  07-50028-A | L5213-001-22 | Sliding door fastener, max. position, RH | 0.4 | 0.9 |
| 07-50030-A | L8534-005-10 | Abrasion protection at sliding door, LH | 0.1 | 0.2 |
|  07-50030-A | L8534-005-20 | Abrasion protection at sliding door, RH | 0.1 | 0.2 |
| 07-50034-B | L5212-001-00 | Jettisonable cockpit doors | 1.2 | 2.6 |
| 07-50035-B | L8504-813-00 | Spoiler position for cockpit doors | 1.1 | 2.4 |
| 07-50039-B | L5211-010-00 | Door catch system for cockpit doors | 0.1 | 0.2 |
| 07-50039-B | L5211-011-00 | Door catch system for sliding doors | 0.2 | 0.4 |
| 07-74043-A | BUL-30420-001-1 | Stretcher FERNO Foldable for ECD-Floor, fixed provisions – Bucher | 0.1 | 0.2 |
| 07-74043-A | BUL-30420-001-2 | Stretcher FERNO Foldable for ECD-Floor, detachable parts – Bucher | 16.5 | 36.4 |
| 07-74042-B | AEL-30120-007-1 | Carbon stretcher dual use, fixed provisions – Aerolite | 1.0 | 2.2 |
| 07-74042-B | AEL-30120-007-2 | Carbon stretcher dual use, detachable parts – Aerolite | 11.0 | 24.3 |
| 07-74045-A | AAT-30120-006-2 | Rescue Stretcher foldable and LOCKER SYSTEM LH or RH, FWD and AFT – AAT | 19.1 | 42.1 |
| 07-83006-A | L2513-004-40 | Quick detachable VIP carpet for passenger cabin | 3.0 | 6.6 |

Avionics

Weight
(margin $\pm 3\%$)

| Document reference | Commercial reference | Title | kg | lb |
|--|----------------------|--|-----|------|
| 08-15028-B | L2319-002-41 | Fixed provisions for GSM phone (antenna, 28V DC, interfacing to ICS) | 0.5 | 1.1 |
| 08-15507-B | L2315-092-00 | IRIDIUM satellite phone AEROPHONE (Aerodata) incl. cockpit dial panel | 4.7 | 10.4 |
| 08-16053-B | L2341-193-01 | Audio/Comm. control system AS 3100-12 (Becker), PAX, in cabin ceiling (LH) | 3.6 | 7.9 |
|  08-17032-A | L2331-003-00 | Cabin loudspeaker | 1.7 | 3.7 |
| 08-18018-B | L2315-001-10 | Headset H 10-76 (David Clark), Low Impedance Spiral Wire | 0.6 | 1.3 |
| 08-18018-B | L2315-001-14 | Headset H10-76 ANR/ENC (David Clark), Low Impedance Spiral Wire | 0.8 | 1.8 |
| 08-18049-A | L2341-007-66 | Adapter cable for ANR headset | 0.2 | 0.5 |
| 08-16063-A | L2341-007-54 | Fixed provisions for ANR | 0.5 | 1.1 |
| 08-18048-A | L2315-001-71 | Bose ANR Headset | 0.4 | 0.9 |
| 08-18049-A | L2341-007-66 | Adapter cable for ANR headset | 0.2 | 0.5 |
| 08-16063-A | L2341-007-54 | Fixed provisions for ANR | 0.5 | 1.1 |
| 08-53004-B | L3424-000-00 | AHRS Free Steering Mode | 0.5 | 1.1 |

NVG Equipment

Different solutions can be offered on request
Some avionics solutions can be NVG modified
NVG compatible cabin and cargo compartment lighting
NVG friendly external lighting kit, comprising position and anti-collision lights
Landing & search light 400/200 W, NVG compatible

Tactical radios

Fixed provisions can be offered on request

Broadcast, Thermal Imaging and Video Surveillance Equipment

Different Thermal Imaging and Video Surveillance Equipment systems and operator consoles on request

6.2 Police/Parapublic

Different equipment can be offered on request:



- SP / DP IFR with FMS / NMS
- SMD68 on copilots' side (6" x 8" display)
- NVG compatible cockpit
- NVG friendly external lighting
- Thermal Imaging and Video Surveillance Equipment with Operator Console and Digital Video Downlink
- SX16 with IFCO, Laserpointer and slaving unit
- Loudspeaker System
- Weather radar
- Tactical radios
- Rappelling devices for 2+2 persons
- Fastroping
- IRIDIUM satellite phone
- Direction finder



6.3 Offshore

Different equipment can be offered on request:

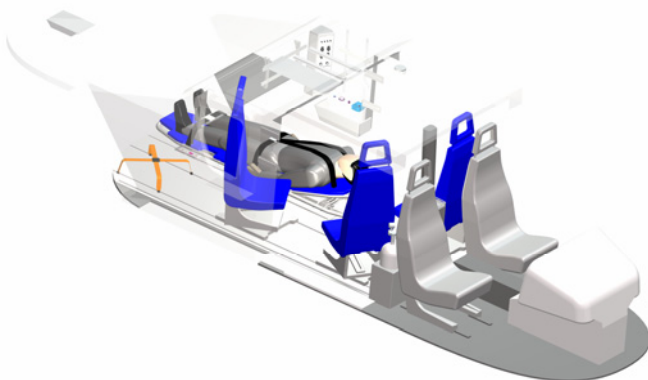


- Automatic Deployable ELT
- Emergency floats
- Radar altimeter with Voice Alert Generator
- Cockpit door jettison
- Emergency EXIT lighting
- AHRS Free Steering Mode
- HEELS
- Underwater Locator Beacon
- Search and weather radar
- Traffic Advisory System TAS 9900BX
- Cabin loudspeaker / Passenger address system
- Light twin HUMS
- Dynamic Monitoring System (DMS)
- Corrosion prevention treatment for offshore operation
- Rear window with push-out (escape window)
- Life rafts

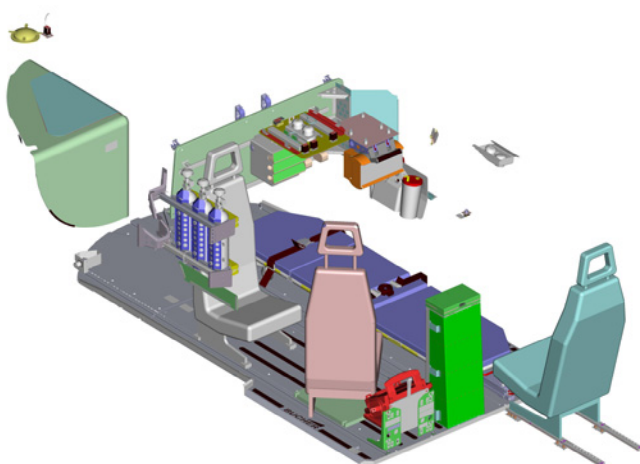


6.4 EMS

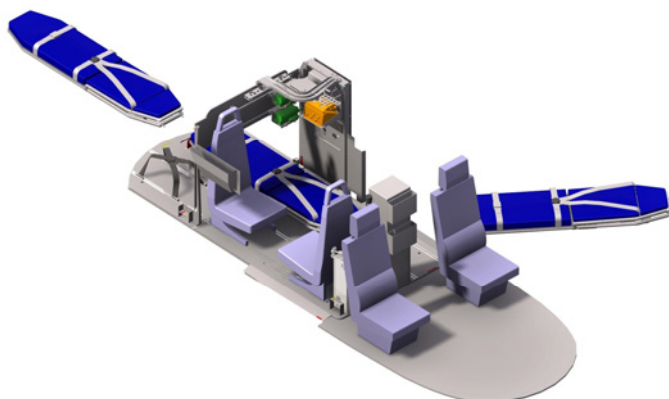
A wide range of equipment can be offered on request from three different manufacturers for medevac, primary and secondary EMS missions as well as for multirole missions.



aerolite



BUCHER



air ambulance technology



Available EMS equipment:

- Stretchers and loading devices
- Incubator transportation systems
- Cabinets
- Medical crew seats
- Special floors
- Retainers for medical apparatuses
- Attachment points for equipment
- Additional stowage options in cabin and cargo compartment
- Oxygen systems
- Suction systems
- Special electrical systems
- Special lighting systems
- EMS Ground Power Unit connector

For efficient and effective operational usage, the EMS equipment needs to be adapted to the specific mission requirements of the customer. Therefore, the definition of the EMS equipment is arranged in a two step approach.

Step 1: Definition of basic mission equipment e.g. stretchers, seats, floors, etc.

The answers to the following questions will be used to prepare an initial EMS configuration based on the customer's mission, operational area, equipment layout, preferred EMS manufacturer and flight crew.

Mission:

- ☐ Primary EMS missions e.g. traffic accidents
- ☐ Secondary EMS missions e.g. hospital transfer flights
- ☐ Multirole missions e.g. primary mission: passenger transport / secondary mission: backup as EMS helicopter
- ☐ Other missions e.g. police, civil protection, VIP

Operational area:

- ☐ City
- ☐ Countryside
- ☐ Mountain
- ☐ Sea

Equipment layout:

- ☐ Basic
- ☐ Advanced
- ☐ High Sophisticated

Preferred EMS manufacturer:

- ☐ Air Ambulance Technology
- ☐ Bucher Leichtbau AG
- ☐ Aerolite AG
- ☐ No preferred manufacturer

Flight crew:

Number of pilot's ____
 Number of medical crew / passenger's ____
 Number of patient's ____

Step 2: Definition of special mission equipment e.g. retainers for medical devices, special electrical systems, oxygen systems, etc.

In addition to the basic mission equipment, further special mission equipment is required to assure a full EMS mission. As the requirements of every customer are different, the clarification of the details should be done in a separate dialogue with the EMS specialists of Eurocopter.

7 Table of constraints

Each item or list of items shown beside the symbol  are concerned by this chapter, please read it exactly in order to find all constraints.

7.1 General Checklist for Incompatibilities

- Detachable parts require the related fixed provisions.
- All recommended configurations in Chapter 5 exclude each other. Mixed Configurations are possible but have to be individually checked.
- Quick detachable VIP carpet for passenger cabin can only be combined with 5 Passenger Transport layout and washable floor covering.
- Only one option (out of several possibilities listed in chapter 7) can be selected for each following specific category
 - external painting
 - first aid kit
 - air conditioning system
 - battery
 - external hoist
 - cargo hook system
 - sliding door fastener (for each side of the helicopter)
 - separation between cabin and cargo compartment
 - floor covering
 - weather radar
 - moving map
- External Hoist has priority over (NSU – respective systems will be deactivated):
 - air conditioning systems
 - sandfilter

| Commercial reference | Title | MSG | Commercial reference | Title |
|----------------------|--|---|----------------------|--|
| L8541-001-20 | Wire Strike Protection System, (WSPS) detachable parts | The protective capability is significantly degraded in combination with | L8511-002-20 | Cargo hook mirrors, detachable parts |
| | | | L2571-001-00 | Radar radome |
| | | | L3216-001-10 | High landing gear |
| | | | | Thermal Imaging and Video Surveillance Equipment / SX16 installation |

7.2 Legend and constraints chart

- XCL** Impossibility of simultaneous fitment of the fixed parts of 2 items of equipment
- NSF** Total or partial incompatibility of simultaneous fitment of the removable parts of two items of equipment
- REQ** Requires the fitting of

| Commercial Reference | Installation | Nature of the Constraint | Commercial Reference | Installation |
|----------------------|---|--------------------------|----------------------|---|
| - | Pure single pilot cockpits (w/o copilot extension) | XCL | L3113-004-20 | Illuminated chart holder for copilot side |
| L2212-001-00 | DAFCS | REQ | L3441-090-04 | Radar altimeter KRA 405B |
| L2513-310-00 | Carpet for cockpit, cabin and cargo compartment floor | XCL | L2818-100-20 | Internal long range fuel tank system, detachable parts |
| L2514-011-00 | Retractable coat hooks | REQ | L2576-001-00 | Avionics compartment |
| L2522-004-10 | Utility seats for 3 rear passengers, | XCL | L2524-001-00 | Separation wall for cabin / cargo compartment |
| L2522-004-10 | Utility seats for 3 rear passengers, | XCL | L2524-021-00 | Separation curtain for cabin / cargo compartment |
| L2522-004-20 | Utility seats for 3 rear passengers, | XCL | L2524-001-00 | Separation wall for cabin / cargo compartment |
| L2522-004-20 | Utility seats for 3 rear passengers, | XCL | L2524-021-00 | Separation curtain for cabin / cargo compartment |
| L2327-001-01 | Traffic Advisory System TAS 9900BX (Ryan) | REQ | L3442-881-00 | Traffic Advisory System TAS 9900BX interface with GNS430 |
| | | | OR | |
| | | | L2327-001-11 | Traffic Advisory System TAS 9900 BX indicator 3" MHD (Ryan) |
| L2524-002-00 | Separation curtain for cockpit / cabin | REQ | L2625-003-00 | 2nd portable fire extinguisher |
| L2524-002-00 | Separation curtain for cockpit / cabin | REQ | L3353-006-20 | Illuminated signs |
| L2524-001-00 | Separation wall for cabin / cargo compartment | REQ | L2576-001-00 | Avionics compartment |
| L2524-021-00 | Separation curtain for cabin / cargo compartment | REQ | L2576-001-00 | Avionics compartment |
| L2525-101-65 | Control covers painted | REQ | L6701-001-00 | Copilot flight controls |
| L2525-104-00 | VIP carpet for cockpit, cabin & cargo | XCL | L2818-100-00 | Internal long range fuel tank system |
| L2526-112-63 | Front cabinet, middle | REQ | L2625-003-00 | 2nd portable fire extinguisher |
| L2563-801-06 | ELT C406-N HM (Artex) w. NAV opt | REQ | | A GPS receiver |
| L2576-002-00 | Dampers for avionics compartment | REQ | L2576-001-00 | Avionics compartment |
| L3168-092-04 | Moving map EURONAV IV | REQ | | A GPS receiver |
| L3168-092-04 | Moving map EURONAV IV | REQ | L3443-010-00 | VRU |
| L3215-001-11 | Emergency floats, fixed provisions | XCL | L3216-001-10 | High landing gear |
| L3215-001-21 | Emergency floats, detachable parts | XCL | L3273-001-00 | Lengthened skids |
| L3215-001-21 | Emergency floats, detachable parts | NSF | L8511-005-20 | Double cargo hook system, det. parts |
| L3216-001-10 | High landing gear | XCL | L3217-001-00 | Reinforced. rear landing gear cross tube |
| L3216-001-10 | High landing gear | XCL | L3273-001-00 | Lengthened skids |
| L3272-001-20 | Snow skids, detachable parts | NSF | L3274-001-20 | Settling protectors, detachable parts |

| Commercial Reference | Installation | Nature of the Constraint | Commercial Reference | Installation |
|------------------------|---|--------------------------|----------------------|--|
| L3274-001-20 | Settling protectors, detachable parts | NSF | L3272-001-20 | Snow skids, detachable parts |
| L3353-005-00 | Emergency lights | XCL | L3322-001-00 | Boarding step illumination |
| L3443-004-00 | Weather radar RDR 1600 | REQ | L3443-010-00 | VRU |
| L3443-090-02 | Weather radar RDR 2000 | REQ | L3443-010-00 | VRU |
| L5211-002-00 | Sliding window in sliding doors | XCL | L5632-001-00 | Tinted windows for passenger cabin |
| L5211-004-00 | Securing device for complete opening of cockpit doors | REQ | L3411-001-00 | Copilot pitot static system |
| L6201-001-30 | Accelerometers | XCL | L6201-004-00 | VMS II (Track & Balance system) ³¹ |
| L6611-001-20 | Main rotor blade folding: fixed provisions for ground handling kit | REQ | L6611-001-10 | Main rotor blade folding: basic kit |
| L8503-001-10 | Fire extinguisher bucket attachment | REQ | L8511-001-10 | Cargo hook system, fixed provisions |
| L8503-001-10 | Fire extinguisher bucket attachment | REQ | L8511-005-10 | Dual cargo hook system, fixed provisions |
| L8511-005-20 | Double cargo hook system, det. parts | NSF | L3215-001-21 | Emergency floats, detachable parts |
| L8511-006-10 | Cargo hook mirrors RH & LH, fixed provisions | XCL | L8511-002-10 | Cargo hook mirrors RH, fixed provisions |
| L8511-006-20 | Cargo hook mirrors RH & LH, detachable parts | XCL | L8511-002-20 | Cargo hook mirrors RH, detachable parts |
| AEL-30130-007-1 | Utility seats for 3 rear passengers, model Aerolite, fixed provision | NSF | L2524-001-00 | Separation wall for cabin / cargo compartment |
| AEL-30130-007-2 | Utility seats for 3 rear passengers, model Aerolite, detachable parts | NSF | L2524-021-00 | Separation curtain for cabin / cargo compartment |

³¹ Accelerometers are already included in VMS II

8 Main performance

The following performance values and figures refer to an EC135, equipped with average production engines.

Unless otherwise specified, the values and figures refer to a clean helicopter at Sea Level (SL), in International Standard Atmosphere (ISA) and zero wind condition.

Performance on 2 engines (AEO) Pratt & Whitney PW206B2

| Gross Weight | kg lb | 2,720 6,000 | 2,835 6,250 | 2,910 6,415 |
|--|------------------------|------------------------|---|---|
| ■ Maximum speed (V_{NE}) | km/h kts | 278 150 | 259 140 | 259 140 |
| ■ Maximum cruising speed (V_H) | km/h kts | 257 139 | 256 138 | 254 137 |
| ■ Fuel consumption at fast cruise speed | kg/h lb/h | 234.5 517 | 234.5 517 | 234.5 517 |
| ■ Economical cruising speed | km/h kts | 226 122 | 228 123 | 230 124 |
| ■ Fuel consumption at economical cruising speed | kg/h lb/h | 200.5 442 | 204.5 451 | 208.5 460 |
| ■ Fuel consumption at 65 KIAS | kg/h lb/h | 158.5 349.5 | 162 357 | 164.5 363 |
| ■ Rate of climb, TOP, SL, ISA | m/s ft/min | 8.9 1,750 | 8.1 1,600 | 7.6 1,500 |
| ■ Hover ceiling IGE (4 ft AGL), TOP, no wind or headwind, ISA | m ft | 4,140 13,600 | 3,655 ³² 12,000 ³² | 3,045 ³³ 10,000 ³³ |
| ■ Hover ceiling IGE (4 ft AGL), TOP, no wind or headwind, ISA + 20°C | m ft | 3,095 10,150 | 2,695 8,850 | 2,435 8,000 |
| ■ Hover ceiling OGE, TOP, ISA | m ft | 3,430 11,050 | 2,685 8,800 | 2,010 6,600 |
| ■ Hover ceiling OGE, TOP, ISA + 20°C | m ft | 2,210 7,250 | 1,785 5,850 | 1,480 4,850 |
| ■ Service ceiling, MCP, (climb reserve 200 ft/min), ISA | m ft | 5,155 16,900 | 3,655 ³² 12,000 ³² | 3,045 ³³ 10,000 ³³ |
| ■ Maximum range (without fuel reserve at economical cruise speed) | | | | |
| ■ standard fuel tank configuration (560 kg) | km nm | 645 348 | 640 345 | 635 342 |
| ■ long range fuel tank configuration (730 kg) | km nm | 850 459 | 840 454 | 835 451 |
| ■ Maximum endurance (without fuel reserve at 65 KIAS) | | | | |
| ■ standard fuel tank configuration (560 kg) | | | | |
| ■ long range fuel tank configuration (730 kg) | h:min h:min | 3:43 4:55 | 3:38 4:49 | 3:35 4:45 |

³² 12,000 ft pressure altitude certification limit

³³ 10,000 ft pressure altitude certification limit

Performance on 1 engine (OEI) Pratt & Whitney PW206B2

| | | | | |
|---|---------------|--------------|--------------|--------------|
| Gross Weight | kg | 2,720 | 2,835 | 2,910 |
| | lb | 6,000 | 6,250 | 6,415 |
| ■ Service ceiling with 100 ft/min climb reserve, MCP OEI-power, ISA | m | 3,275 | 2,925 | 2,715 |
| | ft | 10,750 | 9,600 | 8,900 |
| ■ Service ceiling with 100 ft/min climb reserve, MCP OEI-power, ISA + 20°C | m | 2,375 | 1,965 | 1,710 |
| | ft | 7,800 | 6,450 | 5,600 |
| ■ Rate of climb, MCP OEI-power, SL | m/s | 1.9 | 1.4 | 1.1 |
| | ft/min | 375 | 275 | 215 |
| ■ Max. temperature for CAT A, take-off from clear heliport at SL | °C | + 50 | + 46 | + 43 |
| ■ Max. gross weight hover IGE (4ft AGL), SL, ISA, no wind, 2 min OEI power | kg | | 2,870 | |
| | lb | | 6,325 | |
| ■ Max. gross weight hover IGE (4ft AGL), SL, ISA + 20°C, no wind, 2 min OEI power | kg | | 2,675 | |
| | lb | | 5,885 | |
| ■ Max. gross weight hover OGE, SL, ISA, no wind, 30 sec OEI power | kg | | 2,665 | |
| | lb | | 5,875 | |
| ■ Max. gross weight hover OGE, SL, ISA + 20°C, no wind, 30 sec OEI power | kg | | 2,585 | |
| | lb | | 5,687 | |
| ■ Max. gross weight CAT A, VTOL, SL, ISA | kg | | 2,910 | |
| | lb | | 6,415 | |
| ■ Max. gross weight CAT A, VTOL, SL, ISA + 20°C | kg | | 2,845 | |
| | lb | | 6,270 | |

Performance on 2 engines (AEO) Turbomeca Arrius 2B2

| Gross Weight | kg lb | 2,720 6,000 | 2,835 6,250 | 2,910 6,415 |
|---|------------------------|------------------------|---|---|
| ■ Maximum speed (V_{NE}) | km/h kts | 278 150 | 259 140 | 259 140 |
| ■ Maximum cruising speed (V_H) | km/h kts | 257 139 | 256 138 | 254 137 |
| ■ Fuel consumption at fast cruise speed | kg/h lb/h | 234.5 517 | 234.5 517 | 234.5 517 |
| ■ Economical cruising speed | km/h kts | 237 128 | 239 129 | 240 130 |
| ■ Fuel consumption at economical cruising speed | kg/h lb/h | 215 474 | 219 483 | 221 487 |
| ■ Fuel consumption at 65 KIAS | kg/h lb/h | 167.5 369 | 170.5 376 | 173 381 |
| ■ Rate of climb, TOP, SL, ISA | m/s ft/min | 8.9 1,750 | 8.1 1,600 | 7.6 1,500 |
| ■ Hover ceiling IGE (4 ft AGL), TOP, no wind or headwind, ISA | m ft | 4,325 14,200 | 3,655 ³² 12,000 ³² | 3,045 ³³ 10,000 ³³ |
| ■ Hover ceiling IGE (4 ft AGL), TOP, no wind or headwind, ISA + 20°C | m ft | 3,080 10,100 | 2,670 8,750 | 2,395 7,850 |
| ■ Hover ceiling OGE, TOP, ISA | m ft | 3,430 11,050 | 2,685 8,800 | 2,010 6,600 |
| ■ Hover ceiling OGE, TOP, ISA + 20°C | m ft | 2,175 7,150 | 1,740 5,700 | 1,450 4,750 |
| ■ Service ceiling, MCP, (climb reserve 200 ft/min), ISA | m ft | 5,155 16,900 | 3,655 ³² 12,000 ³² | 3,045 ³³ 10,000 ³³ |
| ■ Maximum range (without fuel reserve at economical cruise speed) | | | | |
| ■ standard fuel tank configuration (560 kg) | km nm | 630 340 | 625 337 | 620 334 |
| ■ long range fuel tank configuration (730 kg) | km nm | 825 446 | 820 443 | 815 440 |
| ■ Maximum endurance (without fuel reserve at 65 KIAS) | | | | |
| ■ standard fuel tank configuration (560 kg) | | | | |
| ■ long range fuel tank configuration (730 kg) | h:min h:min | 3:29 4:36 | 3:26 4:32 | 3:23 4:28 |

Performance on 1 engine (OEI) Turbomeca Arrius 2B2

| Gross Weight | | kg | 2,720 | 2,835 | 2,910 |
|---|---------------|-----------|--------------|--------------|--------------|
| | | lb | 6,000 | 6,250 | 6,415 |
| ■ Service ceiling with 100 ft/min climb reserve, MCP OEI-power, ISA | m | | 3,520 | 3,185 | 2,955 |
| | ft | | 11,550 | 10,450 | 9,700 |
| ■ Service ceiling with 100 ft/min climb reserve, MCP OEI-power, ISA + 20°C | m | | 2,500 | 2,070 | 1,795 |
| | ft | | 8,200 | 6,800 | 5,900 |
| ■ Rate of climb, MCP OEI-power, SL | m/s | | 1.9 | 1.4 | 1.1 |
| | ft/min | | 375 | 275 | 215 |
| ■ Max. temperature for CAT A, take-off from clear heliport at SL | °C | | + 50 | + 47 | + 43.5 |
| ■ Max. gross weight hover IGE (4ft AGL), SL, ISA, no wind, 2 min OEI power | kg | | | 2,870 | |
| | lb | | | 6,325 | |
| ■ Max. gross weight hover IGE (4ft AGL), SL, ISA + 20°C, no wind, 2 min OEI power | kg | | | 2,690 | |
| | lb | | | 5,930 | |
| ■ Max. gross weight hover OGE, SL, ISA, no wind, 30 sec OEI power | kg | | | 2,665 | |
| | lb | | | 5,875 | |
| ■ Max. gross weight hover OGE, SL, ISA + 20°C, no wind, 30 sec OEI power | kg | | | 2,615 | |
| | lb | | | 5,765 | |
| ■ Max. gross weight CAT A, VTOL, SL, ISA | kg | | | 2,910 | |
| | lb | | | 6,415 | |
| ■ Max. gross weight CAT A, VTOL, SL, ISA + 20°C | kg | | | 2,860 | |
| | lb | | | 6,305 | |

OPERATING LIMITATIONS (valid for both versions, EC135 P2i and EC135 T2i)

The helicopter can be operated within the following altitude and temperature limitations (according to the Flight Manual):

| Gross Weight | 2,720 kg | 2,835 kg | 2,910 kg |
|--|---------------------------------------|----------------------------|----------------------------|
| | 6,000 lb. | 6,250 lb. | 6,415 lb. |
| ■ Maximum operating altitude | 6,095 m PA 20,000 ft PA | 3,655 m PA 12,000 ft PA | 3,045 m PA 10,000 ft PA |
| ■ Maximum operating altitude for hover in ground effect, takeoff and landing | 4,570 m DA 15,000 ft DA | 3,655 m PA 12,000 ft PA | 3,045 m PA 10,000 ft PA |
| ■ Minimum temperature | -35°C (-31°F) | | |
| ■ Maximum temperature | ISA + 39°C (max. + 50°C / + 122°F) | | |

Abbreviations

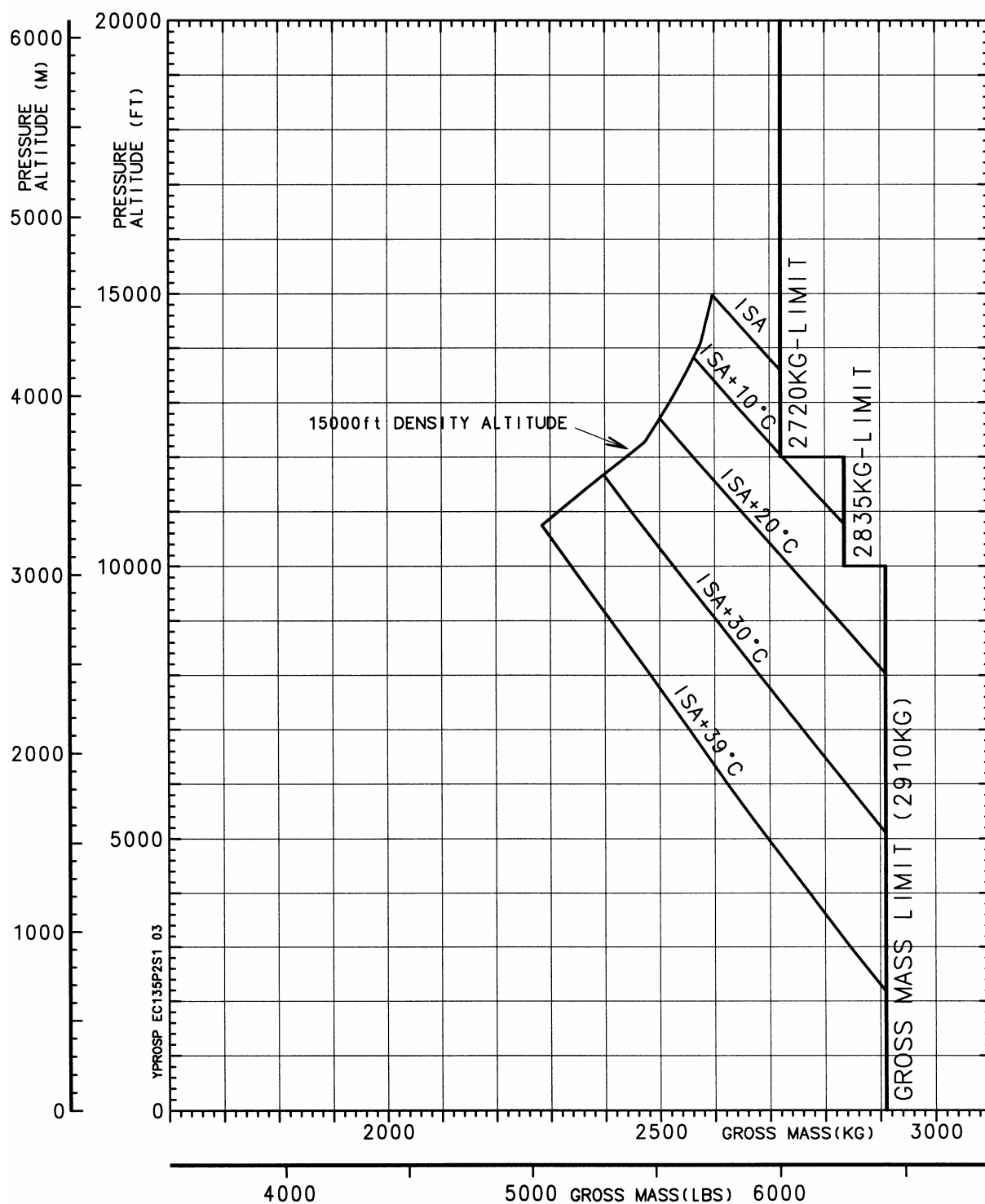
| | | | |
|-----|-----------------------------------|------|-------------------------------|
| AGL | Above Ground Level | OGE | Out Of Ground Effect |
| DA | Density Altitude | PA | Pressure Altitude |
| IGE | In Ground Effect | SL | Sea Level |
| ISA | International Standard Atmosphere | TOP | Take-Off Power |
| MCP | Maximum Continuous Power | VNE | Never-Exceed Speed |
| OEI | One Engine Inoperative | VTOL | Vertical Take-Off and Landing |

Hover In Ground Effect (HIGE, TOP, no wind)

with two PW206B2 engines

| |
|-----------------|
| ISA |
| ISA+10/20/30 °C |
| ISA+39 °C |

| |
|-------------------|
| TAKEOFF-POWER |
| MGT=869 °C TQ=78% |
| BLEED AIR OFF |
| SKID HEIGHT 4 FT |

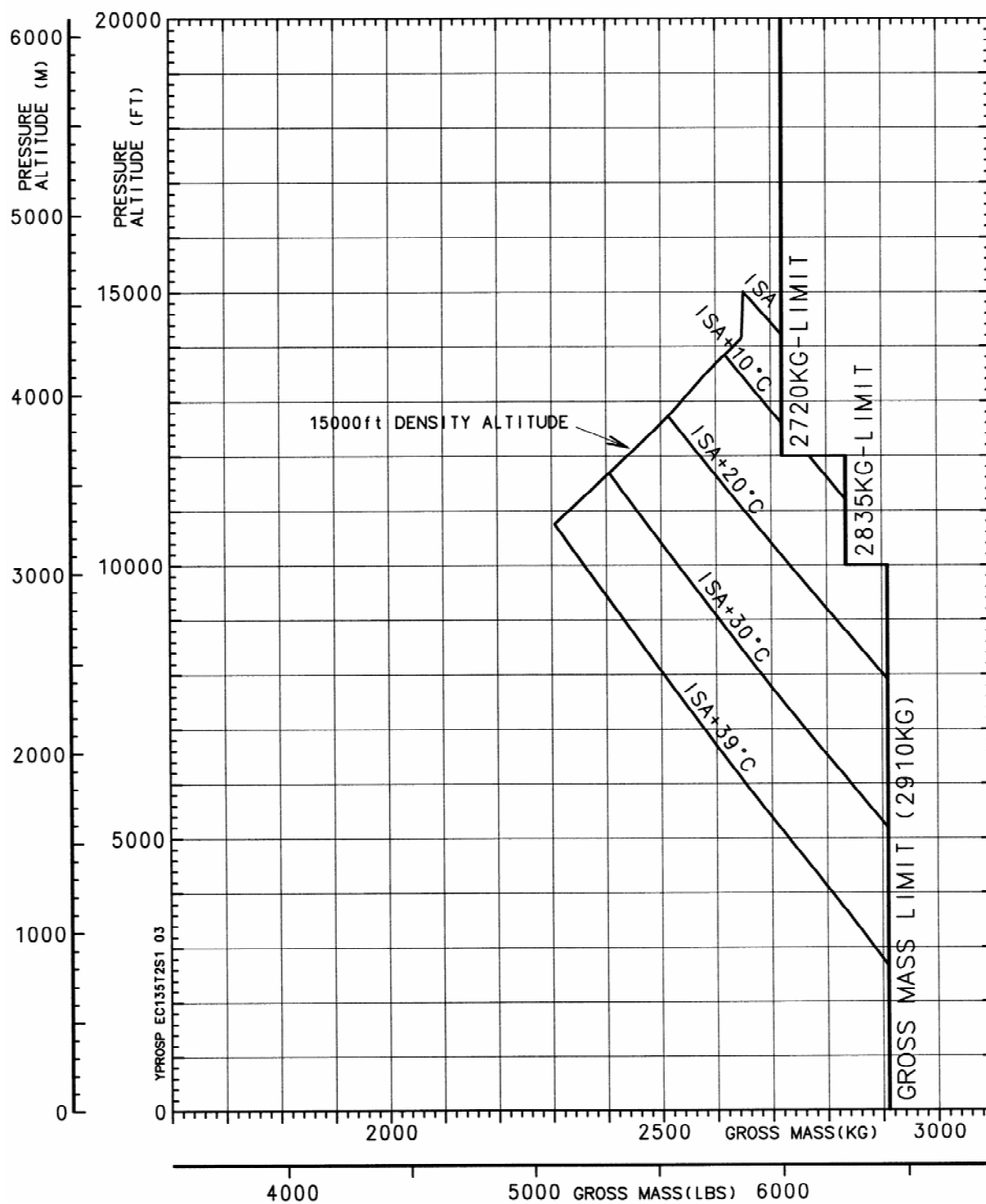


Hover In Ground Effect (HIGE, TOP, no wind)

with two ARRIUS 2B2 engines

| |
|----------------|
| ISA |
| ISA+10/20/30°C |
| ISA+39°C |

| |
|----------------------------|
| TAKEOFF-POWER |
| $\Delta N1 = 0.0\%$ TQ=78% |
| BLEED AIR OFF |
| SKID HEIGHT 4 FT |

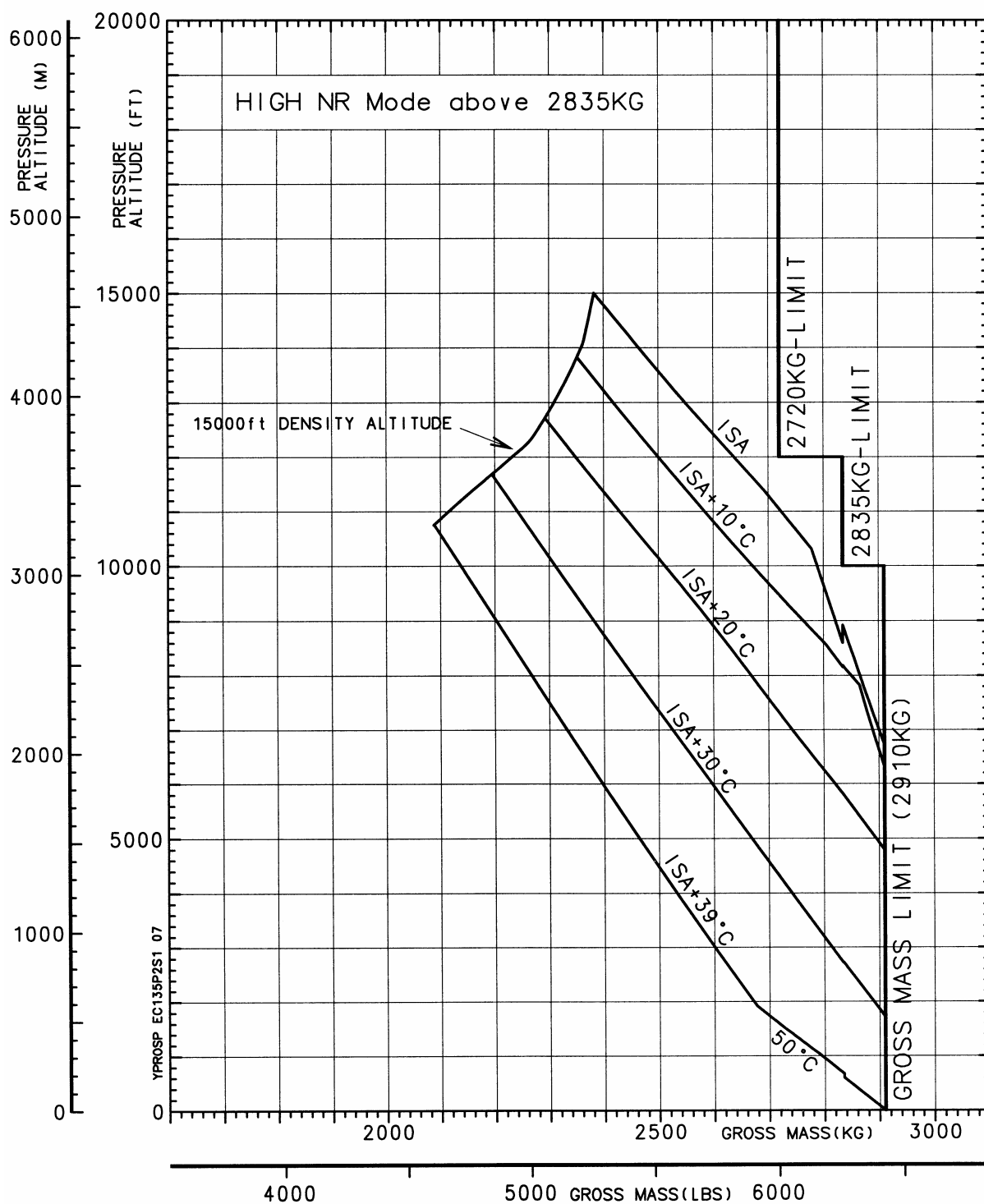


Hover Out Of Ground Effect (HOGE, TOP)

with two PW206B2 engines

| |
|-----------------|
| ISA |
| ISA+10/20/30 °C |
| ISA+39 °C |

| |
|-------------------|
| TAKEOFF - POWER |
| MGT=869 °C TQ=78% |
| BLEED AIR OFF |

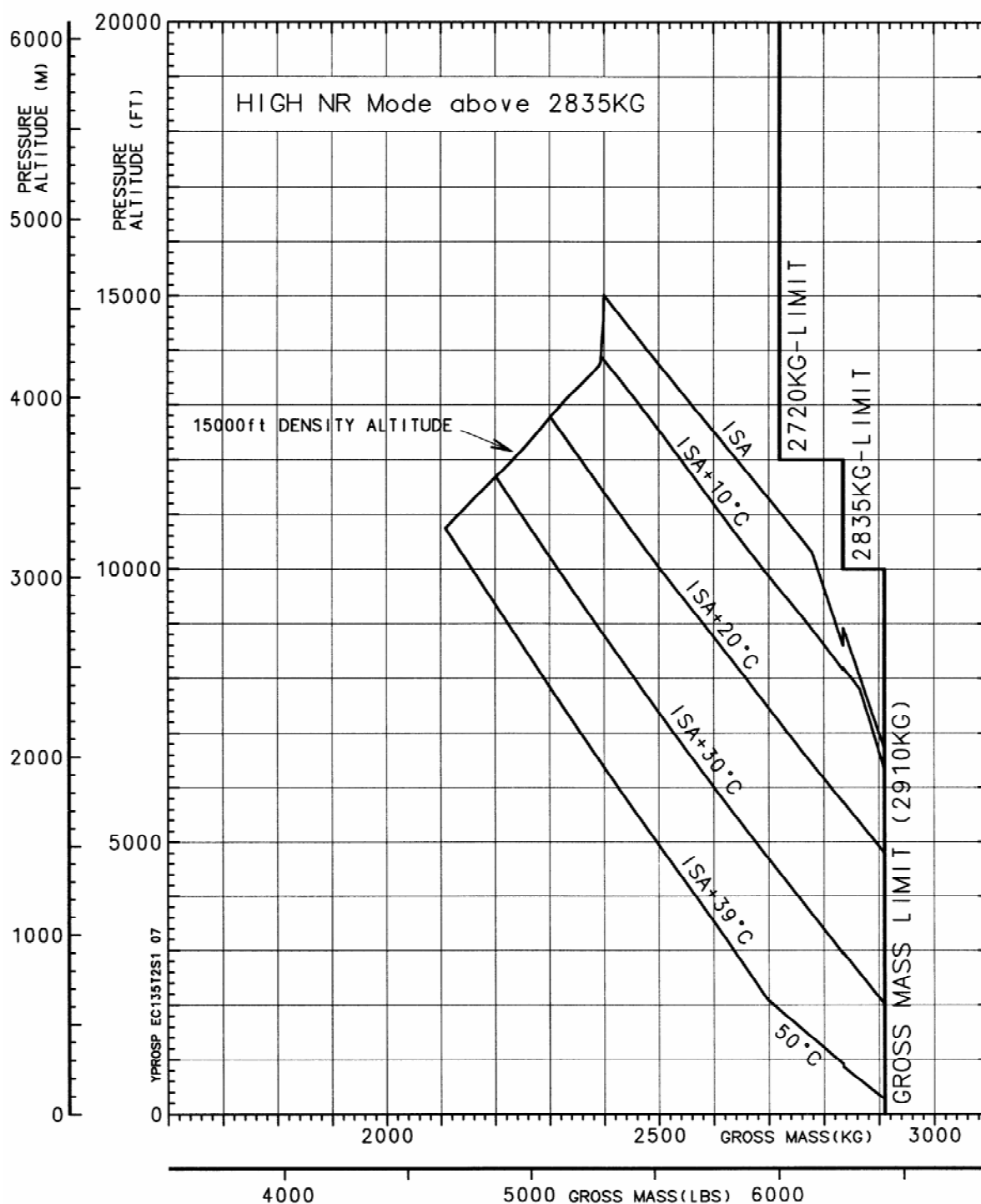


Hover Out Of Ground Effect (HOGE, TOP)

with two ARRIUS 2B2 engines

| |
|----------------|
| ISA |
| ISA+10/20/30°C |
| ISA+39°C |

| |
|----------------------------|
| TAKEOFF-POWER |
| $\Delta N1 = 0.0\%$ TQ=78% |
| BLEED AIR OFF |

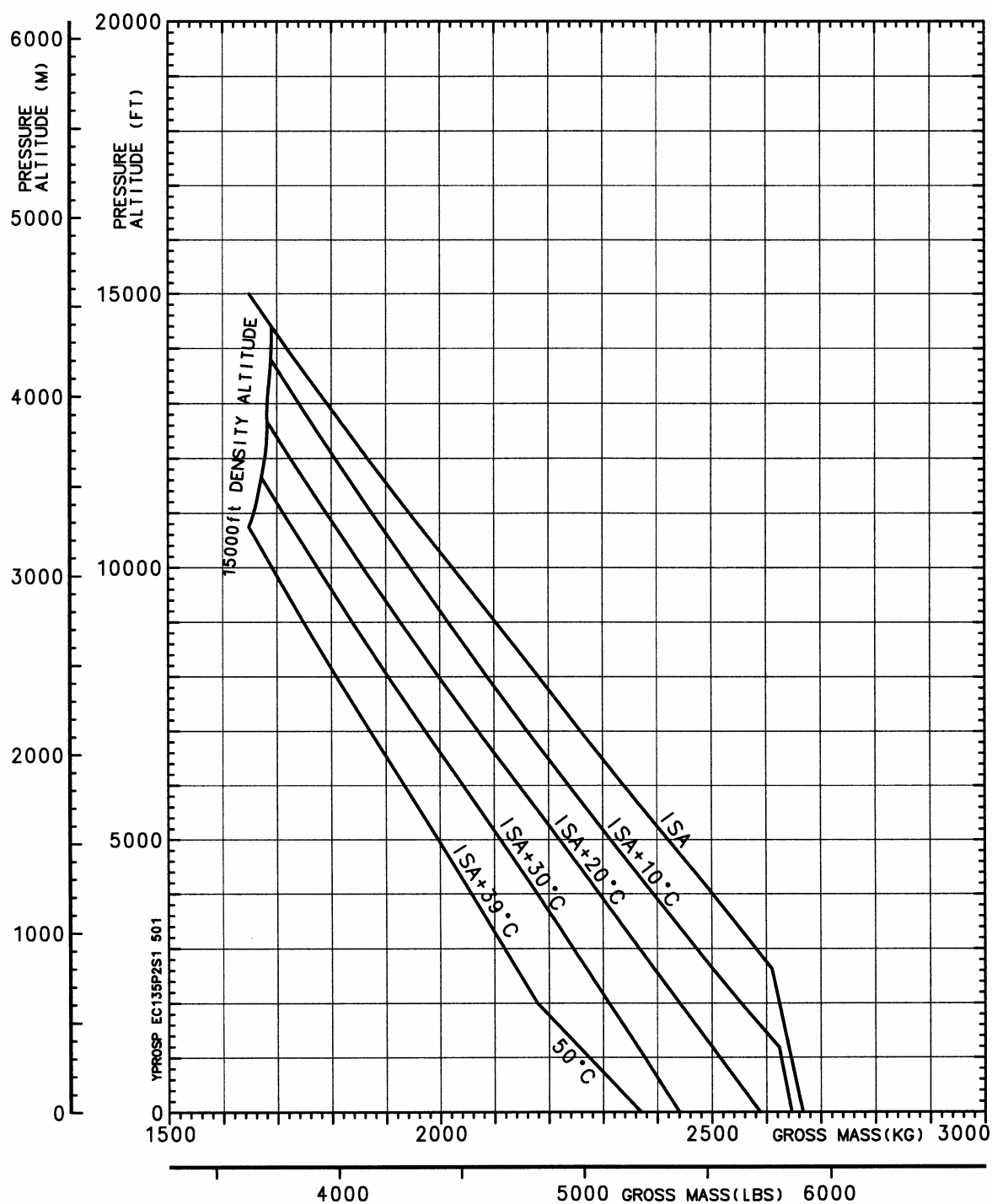


Hover Out Of Ground Effect (HOGE, 30 sec OEI-power)

with one PW206B2 engine

| |
|----------------|
| ISA |
| ISA+10/20/30°C |
| ISA+39°C |

| |
|-------------------|
| OEI 30 SEC-POWER |
| MGT=990°C TQ=128% |
| BLEED AIR OFF |

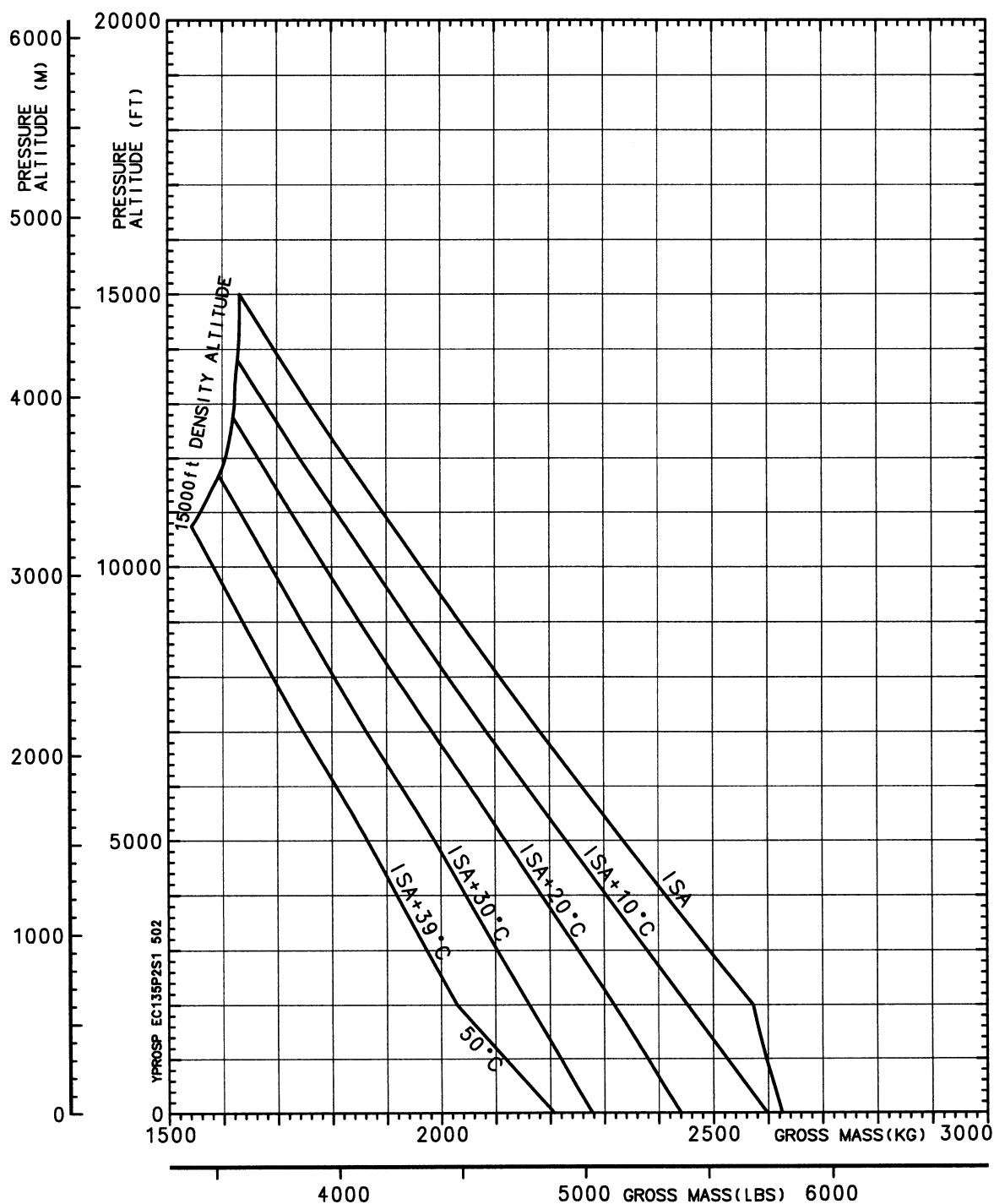


Hover Out Of Ground Effect (HOGE, 2.0 min OEI-power)

with one PW206B2 engine

| |
|----------------|
| ISA |
| ISA+10/20/30°C |
| ISA+39°C |

| |
|-------------------|
| OEI 2.0 MIN-POWER |
| MGT=950°C TQ=125% |
| BLEED AIR OFF |

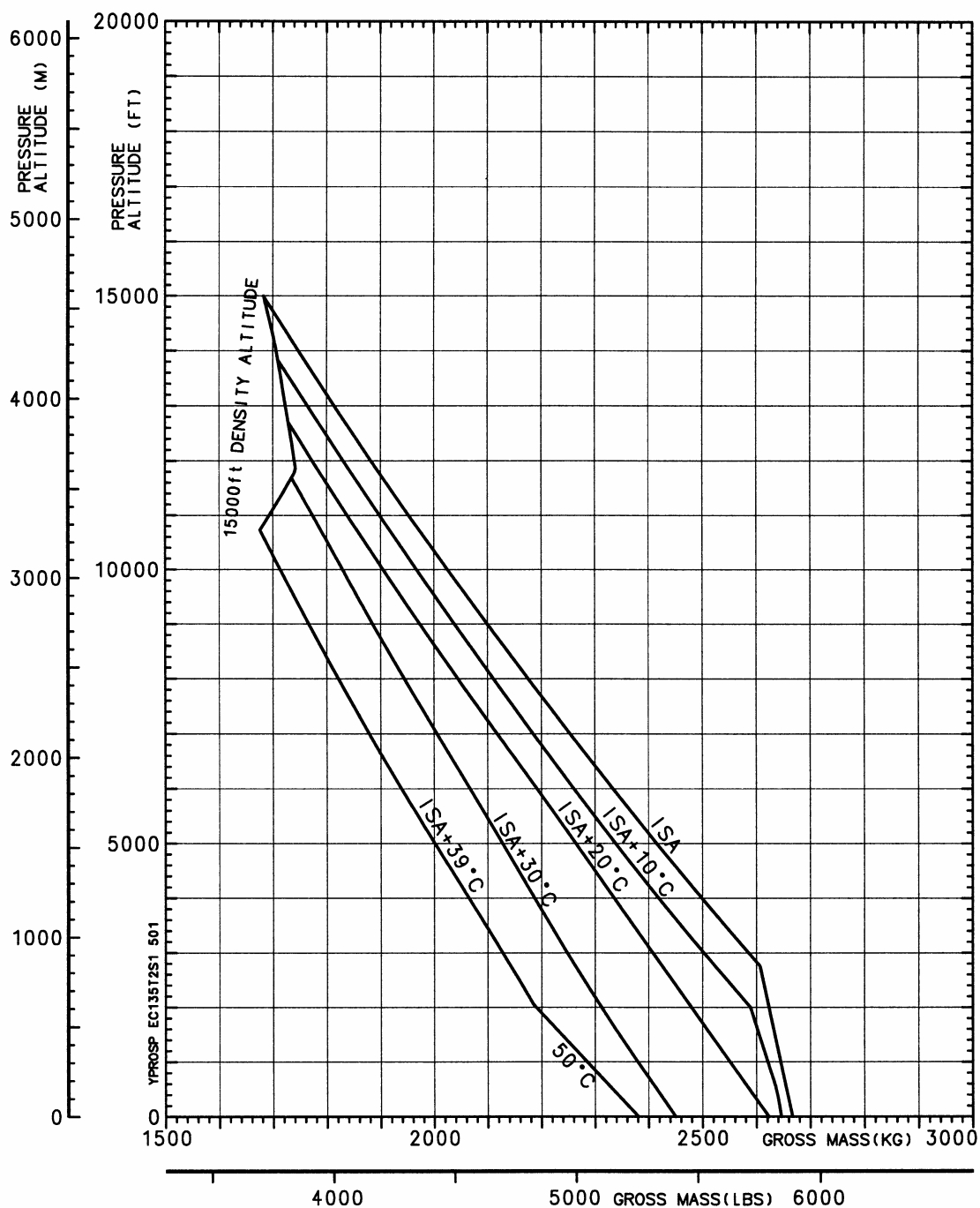


Hover Out Of Ground Effect (HOGE, 30 sec OEI-power)

with one ARRIUS 2B2 engine

| |
|-----------------|
| ISA |
| ISA+10/20/30 °C |
| ISA+39 °C |

| |
|------------------------------|
| OEI 30 SEC-POWER |
| $\Delta N1 = +4.8\%$ TQ=128% |
| BLEED AIR OFF |

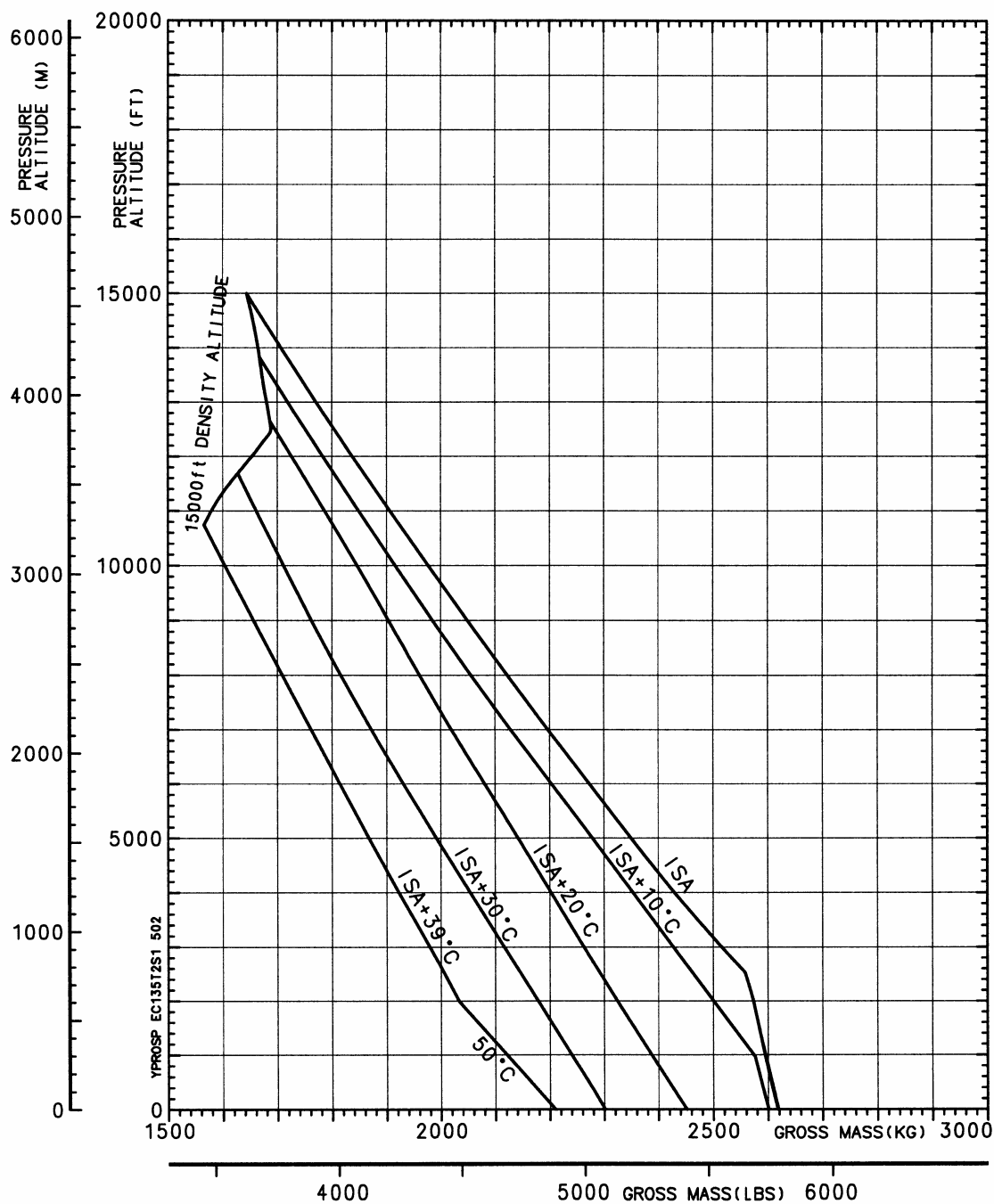


Hover Out Of Ground Effect (HOGE, 2.0 min OEI-power)

with one ARRIUS 2B2 engine

| |
|----------------|
| ISA |
| ISA+10/20/30°C |
| ISA+39°C |

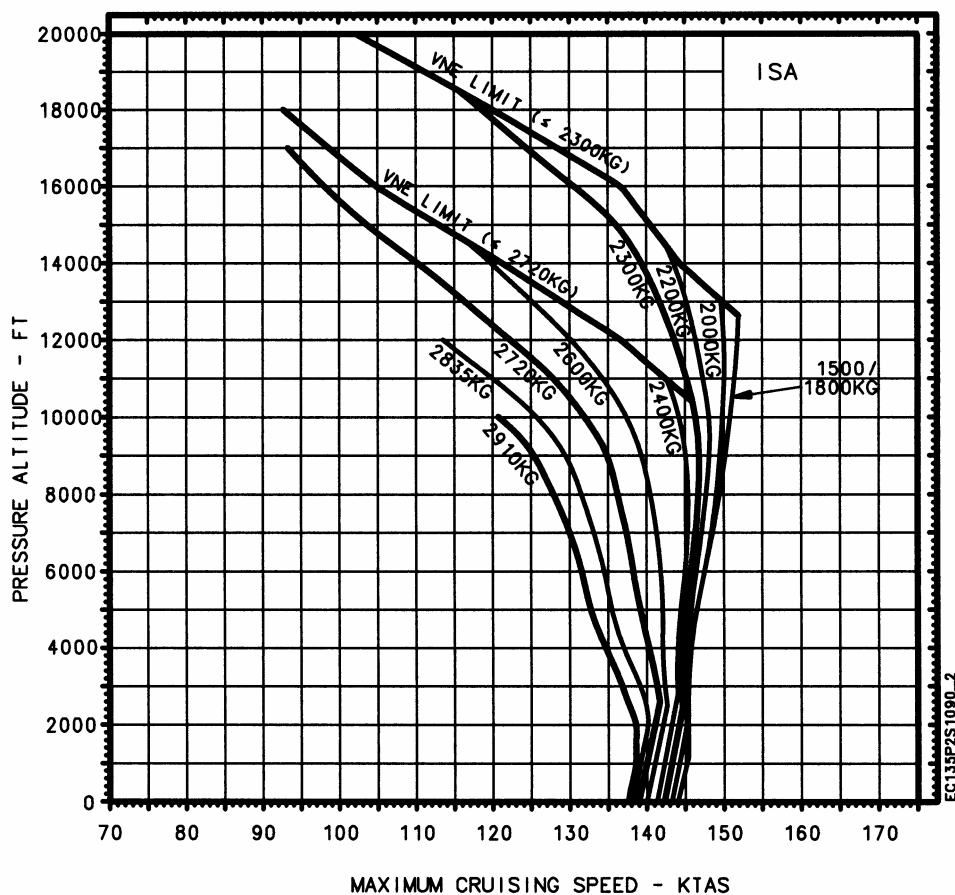
| |
|------------------------------|
| OEI 2.0 MIN-POWER |
| $\Delta N1 = +3.5\%$ TQ=125% |
| BLEED AIR OFF |



Maximum Cruising Speed

with two PW206B2 engines

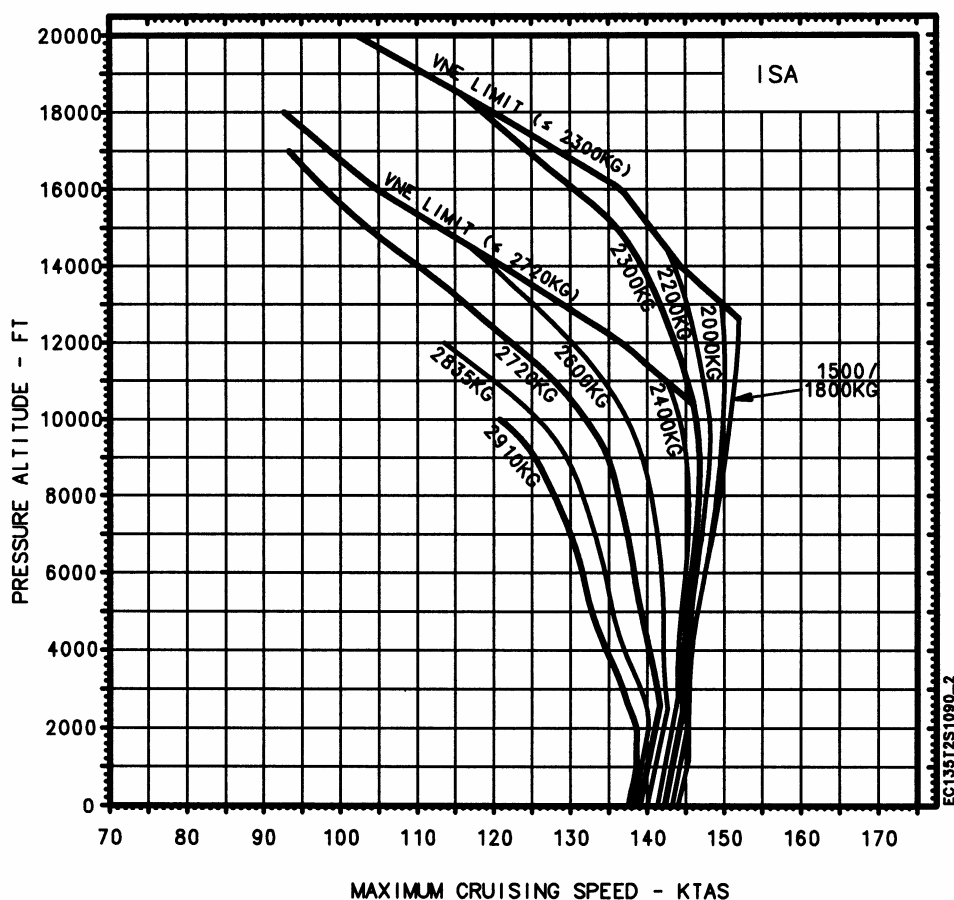
EC135 P2+
MAX. CRUISING SPEED
2 X PRATT & WHITNEY PW206B2
MCP POWER MGT = 835°C
TRANSMISSION LIMIT 69 % TORQUE
BLEED AIR OFF



Maximum Cruising Speed

with two ARRIUS 2B2 engines

EC135 T2+
MAX. CRUISING SPEED
2 X TURBOMECA ARRIUS 2B2
MCP POWER $\Delta N1 = -1.0\%$
TRANSMISSION LIMIT 69 % TORQUE
BLEED AIR OFF

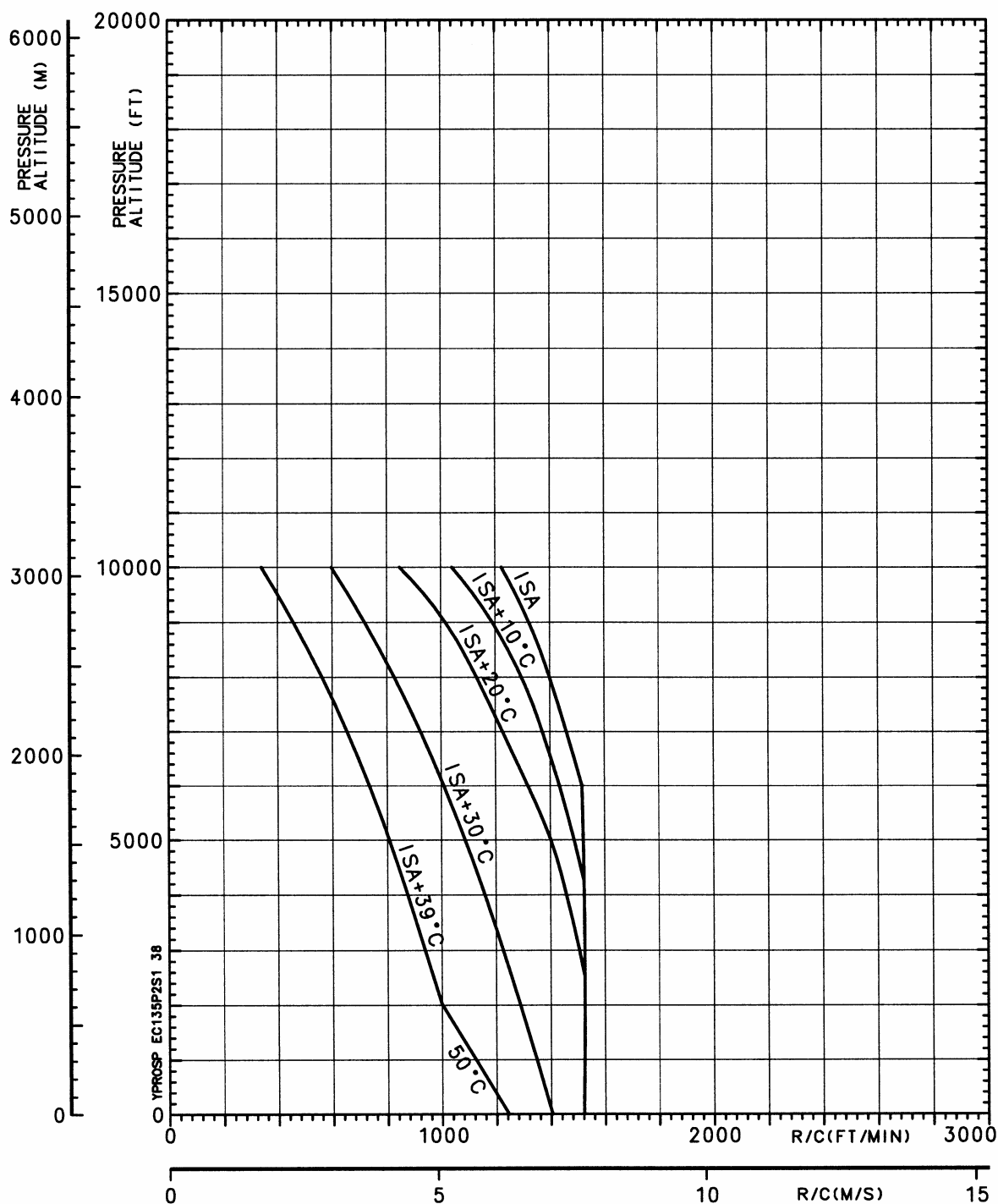


Maximum Rate Of Climb, TOP

with two PW206B2 engines,

| |
|----------------|
| ISA |
| ISA+10/20/30°C |
| ISA+39°C |

| |
|-----------------------|
| AEO TOP-POWER VY=65KT |
| MGT=869°C TQ=78% |
| BLEED AIR OFF |
| GROSS MASS 2910KG |

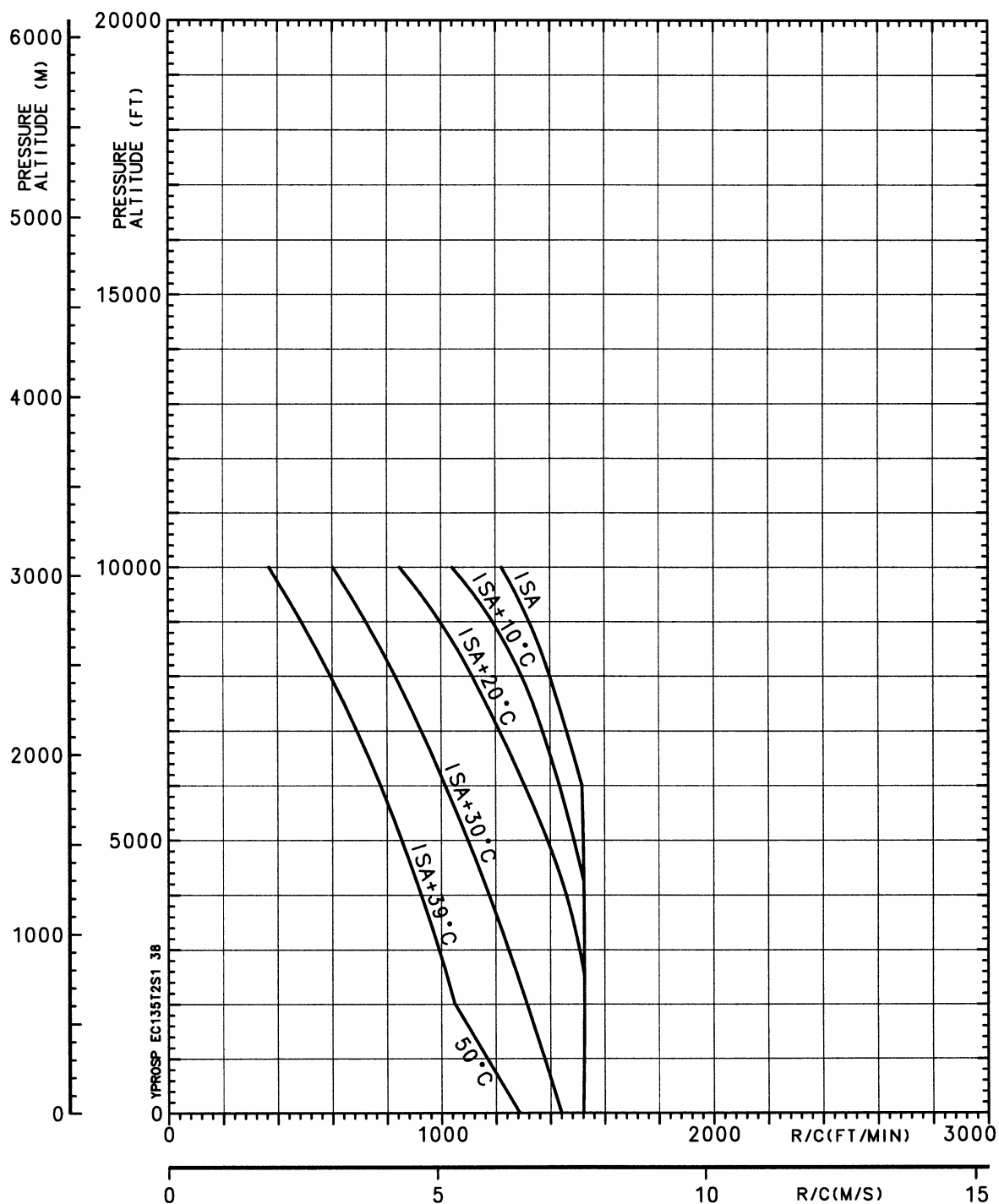


Maximum Rate Of Climb, TOP

with two ARRIUS 2B2 engines,

| |
|----------------|
| ISA |
| ISA+10/20/30°C |
| ISA+39°C |

| |
|-----------------------|
| AEO TOP-POWER VY=65KT |
| ΔN1= 0.0% TQ=78% |
| BLEED AIR OFF |
| GROSS MASS 2910KG |

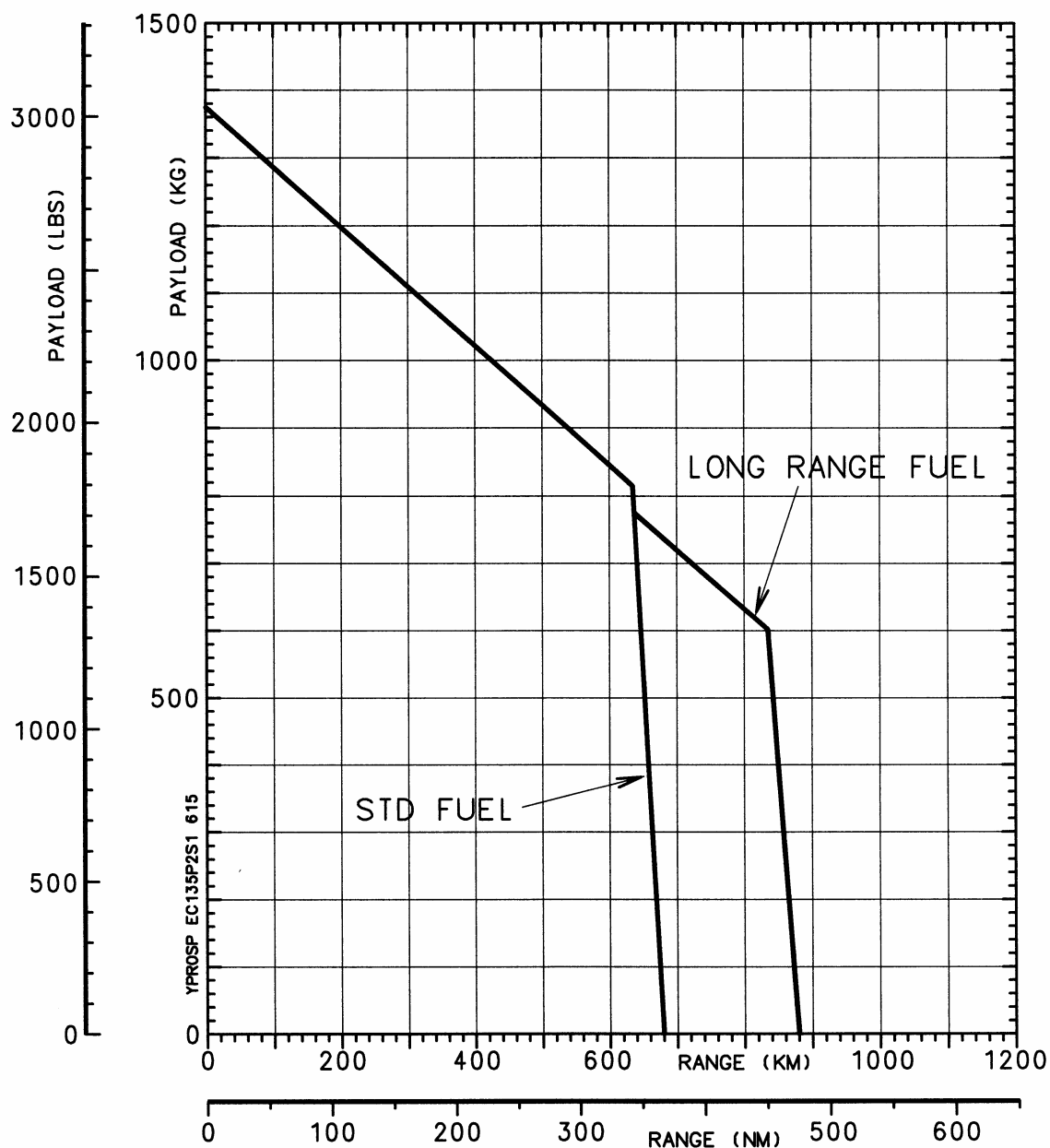


Payload / Range

with two PW206B2 engines

| |
|------------|
| TOW 2910KG |
| NO RESERVE |
| SL / ISA |

| |
|----------------------------|
| EMPTY WEIGHT 1455KG/1498KG |
| USABLE STD FUEL 560KG |
| LONG RANGE FUEL TANK 170KG |
| PILOT 80KG |



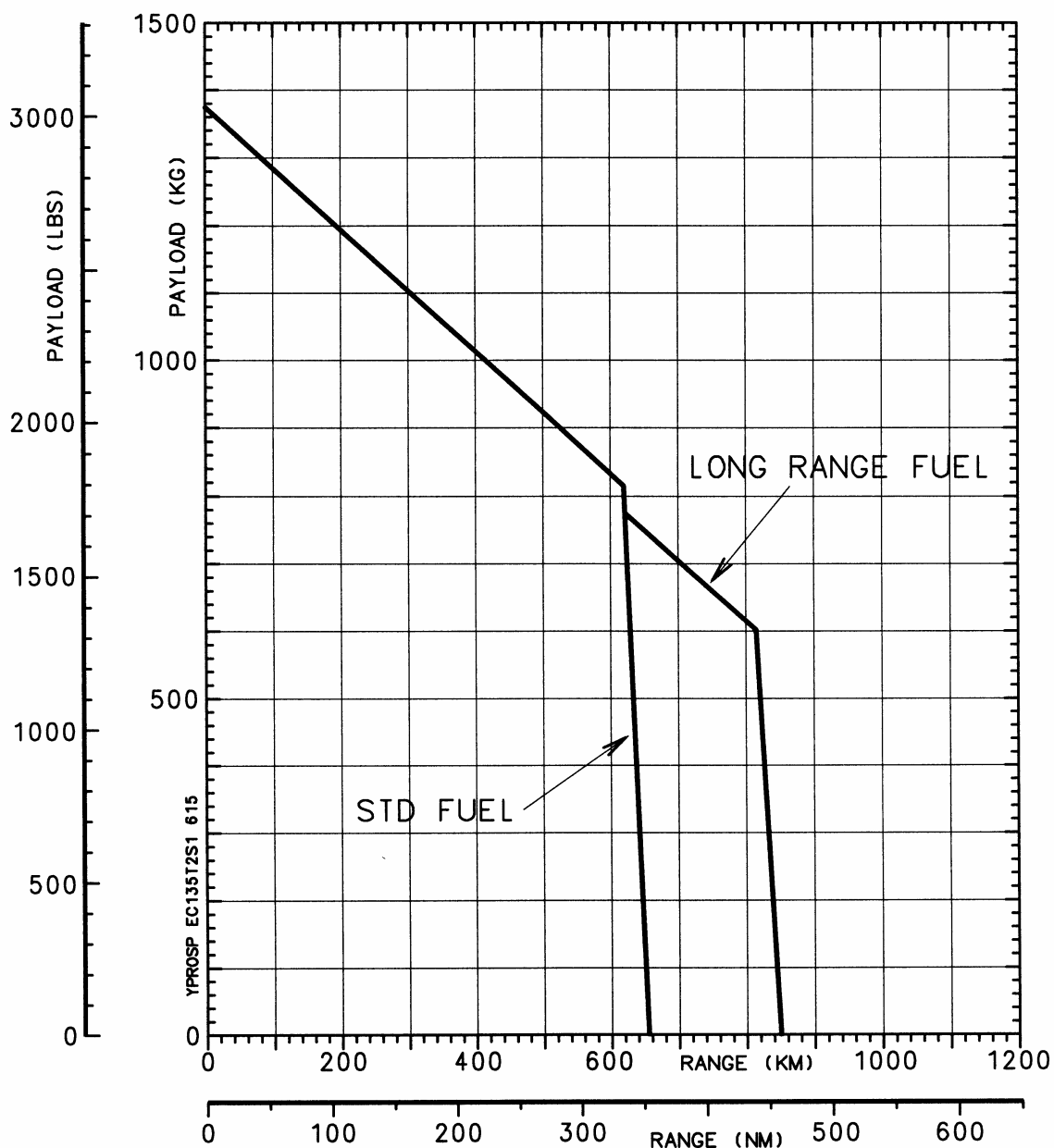
PAYLOAD RANGE DIAGRAM (STD FUEL 560KG)

Payload / Range

with two ARRIUS 2B2 engines

| |
|------------|
| TOW 2910KG |
| NO RESERVE |
| SL / ISA |

| |
|----------------------------|
| EMPTY WEIGHT 1455KG/1498KG |
| USABLE STD FUEL 560KG |
| LONG RANGE FUEL TANK 170KG |
| PILOT 80KG |



PAYLOAD RANGE DIAGRAM (STD FUEL 560KG)

9 Support information

9.1 Assets

Proven reliability and availability based on experience

Eurocopter's helicopter production programs have developed a strong reputation world-wide for being fully committed to providing customers with operational, capable aircraft that achieve high availability combined with cost-effective support systems. To achieve this record of performance, Eurocopter has stressed the importance of working together with its customers to ensure constant feedback on their demonstrated in-service Reliability, Availability and Maintainability/Testability (RAM) data. The main objective is to reach the most optimized operational cost ensuring the highest flight safety.

Eurocopter has built and delivered EC135 helicopters for almost 10 years. More than 600 helicopters are in service world wide. The total flight hours accumulated at the end of the year 2007 are more than 1,040,000 Fh.

9.2 Inspection Program

Airframe Inspection Program

The Maintenance Program specifies the intervals between maintenance operations that are recommended by Eurocopter, irrespective of whether they are mandatory or not.

The program can:

- either be used as is,
- or be adapted by each operator to suit his own specific organization, provided he complies with the maximum intervals.

For the performance of the scheduled and unscheduled maintenance, Eurocopter has classified the maintenance tasks by the following three maintenance levels :

Organizational Level (O)

This level comprises of tasks such as the daily servicing, pre-flight checks, visual inspections for condition, replacement of modular components and simple repairs. This work can be performed by trained helicopter mechanics with standard tools only (no special tools necessary).

Intermediate Level (I)

This level comprises of tasks such as smaller repairs ON/OFF helicopter, periodical inspections and replacement of modular components as well as modifications. This work can be performed by trained and experienced mechanics and may require the usage of special tools and/or test equipment.

Depot Level (D)

This level comprises of tasks such as repairs OFF helicopter, overhaul of major components and heavy airframe repair. This work can be performed by OEM, authorized Repair Centers or Eurocopter and requires the usage of special overhaul and repair tools, as well as test benches (if required) according to repair manuals or instructions. The maintenance tasks defined as O-and I-Level are described in the Aircraft Maintenance Manual (AMM) and can be conducted by operator's mechanics after successful completion of the relevant training courses.

The following table provides an overview of all Scheduled Airframe Inspections. Scheduled inspections with shorter time intervals are fully incorporated into those with longer time intervals, this means, that for example the Periodical Inspection contains the 12-Month Inspection.

| Scheduled Airframe Inspection | Maintenance Level | Estimated Mean Man Hours |
|---|-------------------|--------------------------|
| Pre-flight Check | O - Level | 0.5 hours |
| 12-month inspection (max. exceedance by 3 months) | O - Level | 25 hours |
| Intermediate inspection every 400 hours (max. exceedance by 40 hours) | O - Level | 45 hours |
| Periodical inspection every 800 hours or 3 years (max. exceedance by 80 hours or 3 months) | I - Level | 79 hours |
| Supplementary Inspections according to operating time (max. exceedance by 10%) | I - Level | n/a |
| Conditional Inspections after specific operational incidents (max. exceedance by 10%) | I - Level | n/a |

All those "Estimated Mean Man Hours" mentioned here above refer only to the scheduled inspection in accordance with the Master Servicing Manual (MSM) for the standard helicopter without optional equipment. The announced Man Hours are without ground run and test flight, work preparation and documentation work as well as reworking and servicing.

Scheduled Engine Maintenance for PW206B2

Periodic inspections

| | Maintenance level | Estimated Mean Man Hour |
|-----------|-------------------|-------------------------|
| 12 months | O Level | 1 hour |
| 800 Hours | O Level | 6 hours |

Scheduled Engine Maintenance for ARRIUS 2B2

Periodic inspections

| | Maintenance level | Estimated Mean Man Hour |
|-----------|-------------------|-------------------------|
| 400 Hours | Level 1 | 1,5 hours |
| 800 Hours | Level 1 | 4,0 hours |

9.3 Main components

Time Between Overhaul (TBO) / Service Life Limit (SLL)

| Main Components | TBO as per MSM Rev. 8 | SLL as per MSM Rev. 8 |
|---|------------------------------|-----------------------------|
| Fuel system: | | |
| Motor Pump Cartridge | 5,000 Fh | |
| Ventilation / heating system: | | |
| Blower drive motor (depending on carbon brushes wear) | 800 Fh | |
| Starter/Generator 200 A | 2,000 Fh | |
| Portable fire extinguisher | 10 years TSN 10 years TSO | |
| Main rotor blade damper | | 8,000 Fh or 6 years |
| Swash plate: | | |
| Rings (control ring and bearing ring) | | 8,300 Fh |
| Mixing Lever Gear unit: bolt | | 8,700 Fh |
| Mixing Lever Gear unit: forked lever | | 9,000 Fh |
| Main Transmission | 4,000 Fh | |
| Main Transmission: pinions | | 8,000 Fh |
| Tail Rotor Gearbox | 3,600 Fh | |
| Tail Rotor Gearbox: bevel drive pinion | | 8,000 Fh |
| Tail rotor blade | | 8,000 Fh |
| Engine | 3,500 Fh | |

Note: list includes only components under 10,000 Fh. The figures can be changed.

Time Between Overhauls (TBO) :

The component in question must be removed at each interval that corresponds to the value indicated, in order to undergo the operations in a specialized workshop that will enable it to be put back into service for the next interval. A TBO is granted with a 10 % operational margin, limited at +300 hours. All subcomponents may have a Service Life limit, rated above the TBO limit.

Service Life Limited (SLL) :

The service life limit is an airworthiness limit. The component in question must be removed from service when it reaches the limit indicated.

9.4 Eurocopter Maintenance Support Programs

EUROCOPTER offers its clients a comprehensive array of repair and overhaul services to ensure availability and costs control. This array of services ranges from basic OEM repair and overhaul services up to comprehensive Parts By the Hour (PBH) maintenance programs.

The different services are each tailored for one different user profiles and demands, such as customers:

- with a high number of flight hours,
- with a low number of flight hours,
- looking for immediate component availability,
- that wish budget control,
- ...

To respond to the different customers' demands *EUROCOPTER* offers the following flexible and modular services:

- Classical Support
- Standard exchange
- Repair with guaranteed Turn Around Times (TAT)
- Guaranteed Direct Maintenance Costs (DMC)
- Unscheduled Maintenance Insurance Plan
- Parts by the Hour service

On request, Eurocopter is able to propose extensive services adapted to specific mission and customer expectations, as stock and maintenance management.

9.4.1 Classical Support

The classical support consists of a comprehensive Initial Provisioning package to sustain aircraft operation. This package includes Spare Parts, Tools, Test Equipment, etc..

The required level of operational availability determines the quantity and therefore the investment required. With this support package the Customer bears the responsibility to monitor their repair; manage obsolescence and to procure the right mix and quantity of components and spare parts.

9.4.2 Maintenance, Repair and Overhaul

Eurocopter offers a complete range of industry-leading inspection, repair and overhaul capabilities for the full range of Eurocopter helicopters, with guaranteed turn-around times and cost-efficiency.

Approved by the most stringent airworthiness authorities in the world, our industrial facilities ensure the highest quality for the maintenance, repair and upgrading works. Eurocopter Maintenance Repair and Overhaul covers the four major following areas:

- Blades
- Equipment
- Dynamic component
- Maintenance & upgrades.

9.4.3 Standard Exchange

The Standard Exchange consists in replacing a defective part with a serviceable and interchangeable part within 48 hours subject to availability. This service is available for equipment, blades and dynamic components.

9.4.4 Guaranteed DMC (Repair by the Hour)

The Guaranteed DMC services offers guaranteed repair and overhaul TATs as well as guaranteed prices. This addition to the classical repair and overhaul enables the customer to best size its inventory. Price for this service is calculated per flight hour, thus enabling the customer to spread and predict both his scheduled as unscheduled maintenance expenses. The guaranteed DMC service is available for dynamic components, blades and basic equipment.

9.4.5 Unscheduled Maintenance Insurance Plan (UMIP)

With the UMIP, *EUROCOPTER* gives the customer the option to secure unscheduled maintenance costs while remaining responsible for the scheduled events (overhaul, life limited part replacement). Price for this service is calculated per flight hour.

The UMIP service includes component unscheduled repairs and guaranteed parts replacement within 24H through Standard Exchange based on a dedicated inventory. This service is available for dynamic components, blades and basic equipment

9.4.6 Parts By the Hour (PBH)

The Parts by the Hour (PBH) service is a comprehensive program that offers and balances at the same time guaranteed maintenance costs, reduced inventory and minimized helicopter downtime. This service is intended for Customers looking for total cost control and high level of aircraft readiness. Price for this service is calculated per flight hour.

The PBH service includes component unscheduled repairs component overhauls as well as Life Limited part replacement. Parts replacement is guaranteed within 24H through Standard Exchange based on a dedicated inventory. This service is available for dynamic components, blades and equipment.

9.5 Engine Maintenance program

9.5.1 Pratt&Whitney

With over 4,500 P&WC engines on a “pay-by-the-hour” concept, the industry has demonstrated its confidence on the advantages of the following services:

Eagle Service Plan ®
Fleet Management Plan ®
e-Business Customer Portal

9.5.1.1 PW206B2 Mean Time Between Overhaul (MTBO)

The PW206B2 engine can be overhauled as two (2) individual modules:

- (a) Perform overhaul of the turbomachinery module every 3500 hours.
- (b) Perform overhaul of the reduction gearbox module every 3500 hours.

No Hot-section inspection (HSI) is scheduled and, if necessary, component replacement will be on an on-condition basis.

The maintenance plan for the PW200 engine incorporates the following features:

The minimisation of scheduled maintenance in accordance with MSG-III style analysis.

Helicopter Engine Condition and Trend Monitoring (HECTM®) used to determine the economical threshold to perform refurbishment.

Repair development will continue during service of the PW200 engine with the objective of expanding on repair schemes currently contained in both the Maintenance Manual and Overhaul Manual. This will keep engine maintenance costs, aircraft downtime, spare engine inventories and total operating costs to a minimum.

Accessories Time Between Overhaul

Engine accessories are monitored, as defined by the Illustrated Parts Catalog, with a removal threshold that is based on the operator's experience, with the following exception: TBO recommendation for the Fuel Management Module (FMM) is engine TBO plus 500 hours.

Periodic Inspection – Tolerances

Unless otherwise stated, the tolerance for periodic or scheduled inspections are as follows:

- Hourly Intervals tolerance: Ten percent (10%), or up to a maximum of 100 hours operating time, whichever is less.
- Calendar Intervals tolerance: Ten percent (10%), or up to a maximum of 30 days calendar time, whichever is less.

NOTE: The tolerance is established for maintenance scheduling convenience only and must be approved by the governing civil aviation authority. Subsequent intervals will be adjusted to re-establish the original schedule. When an inspection is done more than 10% early, subsequent inspections will be advanced as required so as to not exceed the maximum tolerance. Concurrence and final approval of the inspection interval tolerance by the governing civil aviation authority is the responsibility of the owner/operator.

9.5.1.2 Service Life Limited (SLL) Components

Three components on the PW206B2 are cycle limited:

| | |
|--------------------------------|---------------|
| Impeller: | 15,000 Cycles |
| Compressor Turbine Disk: | 10,000 Cycles |
| Power Turbine Disk: | 15,000 Cycles |

The compressor Turbine Disk is commercially supported to 15,000 cycles.

9.5.2 Turbomeca

Always looking to maximize your efficiency and reduce your costs, Turbomeca, the engine manufacturer has developed an improved service offering.

Whatever the mission, wherever it may be, for business or pleasure, Turbomeca offers a range of services tailored to your needs.

Standard Exchange: reduces lead times and operational losses by allowing you to exchange the engine or module up for repairs or maintenance at a preferred rate.

AOG: when you need the speed, efficiency, dedicated resources and clear plan of action, an AOG team and dedicated communication channels are implemented to resolve the situation in the most efficient and timely fashion.

Help Line: a multi-skilled team is at your disposal 24/7 to answer questions and quickly implement effective solutions for all your emergency situations.

Warranty: 3 years / 2500 h whichever comes first.

Tailored Support By the Hour® "SBH" contracts: packages are customized and charged monthly according to the number of hours you fly, with no minimums. From basic engine support to fully comprehensive care, SBH® offers flexibility and ease of use for the coverage you need when you need it (OH for engines/modules/accessories, Life Limited Parts, standard exchange or rental system, modular or complete engine maintenance, FOD, ASH, on-site technical assistance, etc.)

Turbomeca Internet Web Site – TOOLS: fully interactive web site developed and programmed specifically for your support needs.

Environment: clean burning with no visible smoke or exhaust soot; the Arrius smoke number is one of the lowest in the industry.

9.5.2.1 Time Between Major Overhauls

The initial TBO value of the ARRIUS 2B2 applicable to the engine / modules / accessories are :

Complete engine 3,500 hours
Module 01 (accessory & reduction gearbox) 3,500 hours
Module 02 (Gas generator) 3,500 hours
Fuel Control Unit 3,500 hours or 10 years (first occurs)

All Gas Generator parts can be replaced in the field through 20 Deep Maintenance procedures.

9.5.2.2 Life-Limited Parts (LLP)

Certain components of the engine, failure of which is classified as a Hazardous effect according to Engine Regulation, have service Life Limits expressed in cycles (reference flight cycle). They are known as critical Life-Limited Parts (LLP).

The limited service life to which this term refers is the number of cycles that a critical life-limited part can run before having to be withdrawn from service.
Cycles are calculated automatically by the Engine Electronic Control Unit. Average experienced rate is about 0,5 cycles / flight hour.

The Life-Limited Parts on the ARRIUS 2B2 are :

Centrifugal impeller 20,000 cycles
HP turbine disc 13,000 cycles
Power turbine disc 14,000 cycles

9.6 Training

With more than 50 years of experience, the Eurocopter training centers provide the most comprehensive, coherent and highest standard helicopter training in the world for pilots and technicians, whether civilian or military.

Qualification training, allowing operators to comply with regulatory requirements, and services training, more mission oriented and tailored to the customers' operational needs, are addressed.

All training courses are established according to the relevant civil aviation authorities' requirements. The centers are approved by the relevant airworthiness authorities (EASA, FAA, DGAC, LBA, CAA...). We are certified ISO 9001: V2000 and regularly audited by independent organisation such as Véritas, AFAQ...

Eurocopter training centers provide a wide range of courses and services, from basic training up to preparation for the most sophisticated civil and military missions.

As part of the full range of services on offer, Eurocopter also plays an active role in helicopter pilot development through its Ab Initio programs.

Centers are equipped with multimedia classrooms. This includes computers overhead projectors and state-of-the-art means such as Computer Aided Instruction (CAI), Computer Based Training (CBT). Some centers also have self-learning laboratories.

Eurocopter has set up a network of 14 training centers. For detailed information refer to Eurocopter specific publication.

Starting mid 2008 training can be performed on EUROCOPTER owned simulators FTD (Flight Training Devices) of highest level according to EASA and JAA standards – based in Europe and North America.



9.7 Technical publications

Eurocopter provides all the technical publications necessary for safely operating and maintaining its aircraft cost effectively.

The printed version of the Aircraft Maintenance Manual (AMM), the System Description Section (SDS) and the Wiring Diagram Manual (WDM) are tailored to the configuration of your helicopter(s).

Eurocopter technical publications are available on an interactive electronic medium or in hard copies.

The INDOC CD-ROM includes the Aircraft Maintenance Manual (AMM), System Description Section (SDS), Master Servicing Manual (MSM), Illustrated Parts Catalogue (IPC) and the Wiring Diagram Manual (WDM), the Corrosion and Erosion Control Guide (CECG) and the Structural Repair Manual (SRM).

The component maintenance manual (CMM) is available on CD-ROM or hard copy, depending on the Vendor.

Along with the INDOC CD-ROM, Eurocopter provides a hard copy of the Airworthiness and Technical Publication (Flight Manual, Pilots Check List, Master Servicing Manual ...) as well as the Service Bulletin Catalogue.

The CD ROM is available in English, French or German; it includes the latest information and is updated regularly.

9.8 T.I.P.I. (Technical Information Publication on Internet)

9.8.1 Description

T.I.P.I. website is entirely dedicated to provide a real-time issuing service for the following publications:

- Service Bulletin, Service Letter, Service Information, Technical Information Letter
- List of Applicable Publications (LOAP)
- List of Master Minimum Equipment List (MMEL)
- Service Life Limited Parts

9.8.2 Main features

- Each time a publication is issued, the customer is automatically informed by an e-mail.
- A small summary, already included in the e-mail, helps the customer to understand quickly the subject.
- Small icons allow the customer to identify immediately the type of information received.
- The download of the publication in pdf format is possible after logging on the T.I.P.I. website.
- A keywords search tool is provided (aircraft family, type of publication, date of edition...).

Address: [www. Eurocopter.com/techpub](http://www.Eurocopter.com/techpub)

The publications are available in English, French or German depending on the case.

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